International Standard



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Hardmetals – Compression test

Métaux-durs - Essai de compression

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Descriptors : hardmetals, mechanical tests, compression tests, proof stress, designation.

Foreword

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Draft International Standards adopted by the technical committees are circulated to the member bodies for approval before their acceptance as International Standards by the ISO Council.

l'eh IEW International Standard ISO 4506 was developed by Technical Committee ISO/T Powder metallurgical materials and products, and was circulated to the member bodies in July 1978.

It has been approved by the member bodies of the following countries

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Austria Bulgaria Czechoslovakia Egypt, Arab Rep. of France Ireland

Italy Japan Mexico Poland Portugal Romania

9e51d/iso-4506-1979 Sweden baf0723 United Kingdom USA USSR Yugoslavia

The member body of the following country expressed disapproval of the document on technical grounds :

Germany, F. R.

International Organization for Standardization, 1979 \bigcirc

Hardmetals – Compression test

1 Scope and field of application

This International Standard specifies a method of determining the ultimate strength and proof stress of hardmetals under uniaxial compressive loads.

2 Principle

Axial loading of a test piece, placed between two hardmetal bearing blocks, until the intended deformation occurs or until the test piece fractures.

3 Symbols and designations

4 Apparatus

The test machine shall be designed and constructed so that loads can be applied at a uniform rate and so that, within the measuring range in question, the maximum loading error will be \pm 1 %.

The test piece shall be affixed between two well-centred and rigidly secured hardmetal anvils with a hardness not less than 1 600 HV. These contact surfaces shall be perpendicular to the loading axis and parallel to each other within 0,5 μ m/mm. An example of a suitable anvil is given in figure 1.

iTeh STANDARD PREVIEV ^{ations} 5 Test piece (standards.iteh.ai)

Symbol Designation 19HH S_0 Minimum original cross-sectional area mm² F_{c} (with index) Load at proof stress, for example : Fc 0.2 Load at 0,2 % proof stress Ν Ultimate load, i.e. load at instant of F_{CU} N fracture R Stress N/mm² Strain εc % Ε Youna's modulus N/mm² R_c (with index) Proof stress, for example : R_{c 0,2} 0,2 % proof stress N/mm^2 Ultimate compressive strength R_{cm} N/mm² **5.1** The dimensions of the test piece shall conform to <u>19</u> figure 2. The end faces and the cylindrical surfaces of the <u>system</u> arged ends shall be ground. Other surfaces should not be <u>4</u> ground (Grinding or polishing may affect the result of the test.)¹¹

5.2 The minimum diameter of the test piece shall be measured with an accuracy of \pm 0,02 mm.

6 Procedure

6.1 Rate of stress increase

The rate at which the load is applied shall be as uniform as possible, and any changes in this rate shall be made gradually and without shock. The rate shall not exceed 8 000 N/s, corresponding approximately to 100 N/($mm^2.s$).

¹⁾ Cylindrical test pieces such as those specified in ASTM E9-1977 or USSR TU-48-19-280-78 may be used to obtain results with potentially less ac curacy.

6.2 Determination of proof stress

6.2.1 The proof stress, for example the 0,2 % proof stress, is determined according to figure 3. This method is based on the fact, valid for almost all metals, that if a load is removed after the elastic limit, D, has been exceeded, the load-compression curve will follow a linear path that is roughly parallel to the loading curve below the elastic limit.

6.2.2 Determination of proof stress using the graphic intersection method is carried out as follows.

6.2.2.1 Apply a pre-load not greater than that required to keep the test piece positioned properly in the machine.

6.2.2.2 Determine the stress-strain curve.

 $\mathsf{NOTE}-\mathsf{Because}$ of the shortness of the test zone and the hardness of the material, practical difficulties are involved in measuring changes in length using displacement gauges of the clamp-on type (extensometers). It is therefore recommended that changes in length be measured using a resistive strain gauge. Two or four gauges should be applied symmetrically at the centre of the test zone. The active length of the gauges should not exceed 8 mm. The results obtained represent an average of the change in length of the test zone.

6.2.3 The proof stress R_{cq} , in newtons per square millimetre, is given by the formula

$$R_{\rm cq} = \frac{F_{\rm cq}}{S_{\rm o}}$$

6.3 Determination of ultimate compressive strength

6.3.1 Load the test piece to fracture.

6.3.2 The ultimate compressive strength $R_{\rm cm}$, in newtons per square millimetre, is given by the formula

$$R_{\rm cm} = \frac{F_{\rm cu}}{S_{\rm o}}$$

7 **Expression of results**

Report the arithmetical mean of at least five determinations, rounded to the nearest 10 N/mm².

CHEST PROFILE 8

The test report shall include the following informations :

ISO 4506a) 77 eference to this International Standard;

6.2.2.3 On the graph thus plotted (figure 3), make OB equal b), all details necessary for identification of the test piece; to the specified residual strain (offset) and draw a line Bafford 39e51d/iso-4506-19/9

to the specified residual strain (offset), and draw a line BA from B parallel to OC. Ordinate F_{c} of intersection point Q has the value F_{ca} and represents the load that corresponds to the proof stress.

It is sometimes difficult to ascertain the direction of line OC from a graph; in such a case, this line can be drawn on the basis of an agreed value of the Young's modulus.

c) the result obtained;

d) all operations not specified by this International Standard, or regarded as optional;

e) details of any occurence which may have affected the result.

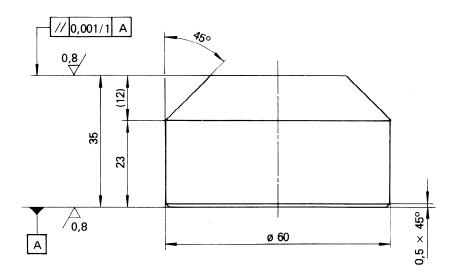
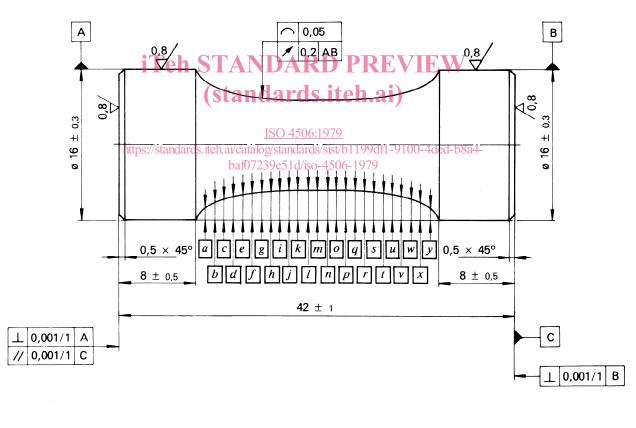


Figure 1 – Suitable hardmetal anvil



а	b	с	d	е	\int	g	h	i	j	k	l
1,21	1,90	2,29	2,54	2,69	2,79	2,86	2,91	2,94	2,96	2,98	2,99

m	n	0	р	q	r	5	t	и	ν	w	X	у
3,00	2,99	2,98	2,96	2,94	2,91	2,86	2,79	2,69	2,54	2,29	1,90	1,21

Figure 2 – Test piece (The 25 co-ordinates *a* to *y* are at 1 mm intervals)

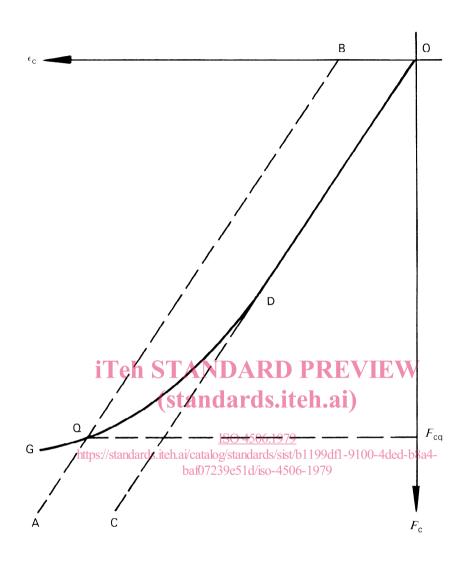


Figure 3 - Load-compression curve

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