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Standard Guide for
Design, Fabrication, and Erection of Fiberglass Reinforced (FRP) Plastic Chimney Liners with Coal-Fired Units[1]

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INTRODUCTION

Federal and state environmental regulations have imposed strict requirements to clean the gases leaving a chimney. These regulations have resulted in taller chimneys (600–1000 ft (183–305 m)) and lower gas temperatures (120–200°F (49–93°C)) due to the use of Air Quality Compliance Systems (ACQS). These regulations led to the development of fiber reinforced plastics (FRP) chimney liners in the 1970's.

Fiberglass-reinforced plastic liners have proven their capability to resist corrosion and carry loads over long periods of time. Successful service has been demonstrated in the utility and general-process industries for over 40 years. The taller FRP structures and larger diameters (10–30 ft (3–9 m)) imposed new design, fabrication, and erection challenges.

The design, fabrication, and erection of FRP liners involves disciplines which must address the specific characteristics of the material. Areas that have been shown to be of importance include the following:

(1) Flue-gas characteristics such as chemical composition, water and acid dew points, operating and excursion temperature, velocity, etc.

(2) Plant operation as it relates to variations in the flue-gas characteristics.

(3) Material selection and laminate design.

(4) Quality control throughout the design, fabrication, and erection process to ensure the integrity of the corrosion barrier and the structural laminate.

(5) Secondary bonding of attachments, appurtenances, and joints.

(6) Installation and handling.

(7) Inspections and Confirmation Testing.

Chimney components include an outer shell, one or more inner liners, breeching ductwork, and miscellaneous platforms, elevators, ladders, and miscellaneous components. The shell provides structural integrity to environmental forces such as wind, earthquake, ambient temperatures, and supports the liner or liners. The liner or liners inside the shell protects the shell from the thermal, chemical, and abrasive environment of the hot boiler gases (generally 120–560°F (49–293°C)). These liners have been made of FRP, acid-resistant brick, carbon steel, stainless steel, high-alloy steel, shotcrete-coated steel, and shotcrete-coated shells. The selection of the material type depends on the chemical composition and temperature of the flue gas, liner height, diameter, and seismic zone. Also, variations in flue-gas characteristics and durations of transient temperatures affect material selection and design. For FRP liners, the flue gas maximum operating temperature is generally limited to 200°F (90°C) for 2 hours and for maximum transient temperatures to 400°F (204°C) for 30 minutes.

1. Scope

1.1 This guide offers direction and guidance to the user concerning available techniques and methods for design, material selection, fabrication, erection, inspection, confirmatory testing, quality control and assurance.

1.2 These minimum guidelines, when properly used and implemented, can help ensure a safe and reliable structure for the industry.

1.3 This guide offers minimum requirements for the proper design of a FRP liner once the service conditions relative to thermal, chemical, and erosive environments are defined. Due

to the variability in liner height, diameter, and the environment, each liner must be designed and detailed individually.

1.4 Selection of the necessary resins and reinforcements, composition of the laminate, and proper testing methods are offered.

1.5 Once the material is selected and the liner designed, procedures for proper fabrication of the liner are developed.

1.6 Field erection, sequence of construction, proper field-joint preparation, and alignment are reviewed.

1.7 Quality control and assurance procedures are developed for the design, fabrication, and erection phases. The quality-assurance program defines the proper authority and responsibility, control of design, material, fabrication and erection, inspection procedures, tolerances, and conformity to standards. The quality-control procedures provide the steps required to implement the quality-assurance program.

1.8 Appendix X1 includes research and development subjects to further support recommendations of this guide.

1.9 *Disclaimer*—The reader is cautioned that independent professional judgment must be exercised when data or recommendations set forth in this guide are applied. The publication of the material contained herein is not intended as a representation or warranty on the part of ASTM that this information is suitable for general or particular use, or freedom from infringement of any patent or patents. Anyone making use of this information assumes all liability arising from such use. The design of structures is within the scope of expertise of a licensed architect, structural engineer, or other licensed professional for the application of principles to a particular structure.

NOTE 1—There is no known ISO equivalent to this standard.

1.10 The values stated in inch-pound units are to be regarded as standard. The values given in parentheses are mathematical conversions to SI units that are provided for information only and are not considered standard.

1.11 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety, health, and environmental practices and determine the applicability of regulatory limitations prior to use.*

1.12 *This international standard was developed in accordance with internationally recognized principles on standardization established in the Decision on Principles for the Development of International Standards, Guides and Recommendations issued by the World Trade Organization Technical Barriers to Trade (TBT) Committee.*

2. Referenced Documents

2.1 *ASTM Standards:*[2]

C177 Test Method for Steady-State Heat Flux Measurements and Thermal Transmission Properties by Means of

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[2] For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

3. Terminology

3.1 *Definitions:*

3.1.1 Terms used in this guide are from Terminology D883 unless otherwise indicated in 3.2.

3.2 *The following applicable definitions in this guide are provided for reference:*

3.2.1 *accelerator*—a material added to the resin to increase the rate of polymerization (curing).

3.2.2 *axial*—in the direction of the axis (lengthwise centerline) of the equipment.

3.2.3 *Barcol hardness*—measurement of the degree of cure by means of resin hardness. The Barcol impressor is the instrument used (see Test Method D2583).

3.2.4 *binder*—chemical treatment applied to the random arrangement of glass fibers to give integrity to mats. Specific binders are utilized to promote chemical compatibility with various laminating resins used.

3.2.5 *blister*—refer to Terminology D883.

3.2.6 *bonding*—joining of two or more parts by adhesive forces.

3.2.7 *bond strength*—force per unit area (psi) necessary to rupture a bond in interlaminar shear.

3.2.8 *buckling*—a mode of failure characterized by an unstable lateral deflection due to compressive action on the structural element involved.

3.2.9 *burned areas*—areas of laminate showing evidence of decomposition (for example, discoloration and cracking) due to excessive resin exotherm.

3.2.10 *burn out (burn off)*—thermal decomposition of the organic materials (resin and binders) from a laminate specimen in order to determine the weight percent and lamination sequence of the glass reinforcement.

3.2.11 *carbon veil*—a nonwoven surface veil that is made of carbon fiber or is coated with conductive carbon for purposes of providing static dissipation. This could be carbon veil, or polyester veil impregnated with carbon.

3.2.12 *catalyst*—an organic peroxide material used to activate the polymerization of the resin.

3.2.13 *chopped-strand mat*—reinforcement made from randomly oriented glass strands that are held together in a mat form by means of a binder.

3.2.14 *chopper gun*—a machine used to cut continuous fiberglass roving to predetermined lengths [usually 0.5–2 in. (13–51 mm)] and propel the cut strands to the mold surface. In the spray-up process, a catalyzed resin is deposited simultaneously on the mold. When interspersed layers are provided in filament winding, the resin spray is not used.

3.2.15 *contact molding*—process for molding reinforced plastics in which reinforcement and resin are placed on an open mold or mandrel. Cure is without application of pressure; includes both hand-lay-up and spray-up.

3.2.16 *corrosion barrier*—the integral inner barrier of the laminate which is made from resin, veil, and chopped mat.

3.2.17 *coverage*—see *winding cycle.*

3.2.18 *crazing*—the formation of tiny hairline cracks in varying degrees throughout the resin matrix, particularly in resin-rich areas.

3.2.19 *cut edge*—end of a laminate resulting from cutting that is not protected by a corrosion barrier.

3.2.20 *delamination*—physical separation or loss of bond between laminate plies.

3.2.21 *dry spot*—an area where the reinforcement fibers have not been sufficiently wetted with resin.

3.2.22 *edge sealing*—application of reinforcement and resin, or resin alone, to seal cut edges and provide a corrosion-resistant barrier. The final layer should be paraffinated.

3.2.23 *entrapped-air void*—see *void*.

3.2.24 *environment*—state of the surroundings in contact with the internal and external surfaces, including the temperature, pressure, chemical exposure, relative humidity, and presence of liquids or gases.

3.2.25 *exotherm*—evolution of heat by the resin during the polymerization reaction.

3.2.26 *exotherm ply*—that ply of chopped mat at which the lamination process is stopped to allow gelation and exotherm of the existing laminate.

3.2.27 *fabricator*—the producer of the equipment who combines resin and reinforcing fibers to produce the final product.

3.2.28 *fatigue*—the change in properties of the laminate over time under cycling of loads, including mechanical, temperature, and other environmental exposures.

3.2.29 *fiber(glass)*—a fine, continuously formed thread of glass. *E*-glass is used for strength and durability, *E-CR*-glass is a modified *E*-glass with improved corrosion resistance to most acids, and *C*-glass is resistant to corrosion by most acids.

3.2.30 *fiberglass roving*—see *roving*.

3.2.31 *fiberglass woven roving*—heavy fabric woven from strands of glass fiber.

3.2.32 *fiber wetting*—coating of the fiberglass with resin by means of rollout or immersion.

3.2.33 *filament winding*—a process for forming FRP parts by winding resin-saturated continuous-roving strands onto a rotating mandrel.

3.2.34 *fillers*—inert materials that are added to the resin to increase density, increase viscosity, improve abrasion resistance, enhance resin-application properties, decrease resin shrinkage, and reduce cost.

3.2.35 *fill picks*—the rovings in a woven roving that run in the transverse direction of the fabric, that is, across the fabric roll width.

3.2.36 *flame-retardant resin*—halogenated resins that can be used with or without additives to provide a laminate having a reduced flame-spread rating as measured in accordance with Test Method E84. The resins are not flame retardant in their liquid state.

3.2.37 *flame-spread rating*—index number for any laminate of definite composition resulting from testing in accordance with Test Method E84.

3.2.38 *gap filling*—the filling of voids between joined parts, elements, or components with resin putty or resin.

3.2.39 *gel*—the initial jelly-like solid phase that develops during the polymerization of resin.

3.2.40 *gel time*—time from the initial mixing of the resin with catalyst to gelation.

3.2.41 *glass*—see *fiber(glass)*.

3.2.42 *glass content*—weight percent of glass-fiber reinforcement in the laminate.

3.2.43 *gun roving*—fiberglass roving designed for use in a chopper gun.

3.2.44 *hand lay-up*—see *contact molding*.

3.2.45 *heat-deflection temperature (HDT)*—temperature at which a specified bar specimen deflects 0.010 in. (0.25 mm) when loaded as a simple beam at a constant 264 psi (1820 kPa). Test Method D648 usually refers to a cured-resin casting, not a laminate.

3.2.46 *helical winding*—filament winding where the angle at which the reinforcement is placed is other than 0 or 90°.

3.2.47 *hoop winding*—filament winding where the winding angle is essentially 90°. The winding strands are applied immediately adjacent to the strands applied on the previous mandrel revolution.

3.2.48 *intersperse*—chopped fiberglass used in a filament-wound laminate, usually in thin layers between winding coverages.

3.2.49 *isotropic*—having uniform properties in all directions. The measured properties of the material are independent of the axis of testing. The opposite is anisotropic, which is the case for FRP laminates.

3.2.50 *joint overlay*—an overlay that joins the adjoining surfaces of two contacting parts or elements.

3.2.51 *laminate*—the total of the part constructed by combining one or more layers of material (reinforcement and resin). As used in this guide, the laminate consists of the corrosion barrier on the inner surface, the interior structural layer, and the outer surface.

3.2.52 *laminate composition*—the sequence of reinforcement materials on a type, class, and category basis that make up a laminate.

3.2.53 *lamination analysis*—procedure by which, given the amount and properties of the resin and the properties and orientation of the reinforcement, it is possible to calculate the elastic physical and mechanical properties of the individual layers of a laminate and using weighted-averaging techniques to determine the elastic properties of the total laminate (see section 2.4).

3.2.54 *lamination theory*—see *lamination analysis*.

3.2.55 *mandrel*—mold around which a laminate is formed to fabricate a cylindrical section.

3.2.56 *macro*—denotes the properties of the laminate as a total structural element.

3.2.57 *matrix*—resin phase of a fiberglass-reinforced laminate.

3.2.58 *micro*—denotes the properties of the constituent elements of the laminate; that is, matrix and reinforcements and interface only, and their effect on the laminate properties.

3.2.59 *mold*—form over or into which resin and reinforcements are placed to form the laminate product shape.

3.2.60 *monomer*—the basic polymerizing element for the formation of the matrix; in FRP-liner fabrication, this is mostly styrene.

3.2.61 *overlay*—laminate applied over base FRP structures to secure a joint, seal a seam, or attach a nozzle.

3.2.62 *paraffinated resin*—resin containing a small amount of dissolved paraffin wax. This wax will come out of the solution during cure and bloom to the surface, preventing the normal air inhibition at the atmospheric exposed surface.

3.2.63 *parting agents*—compounds that assist in releasing the FRP part from its mold; also referred to as mold-release agents.

3.2.64 *pass*—in filament winding, one "round trip" of the carriage (which applies the winding strand to the mandrel) from one end of the mandrel to the other and return.

3.2.65 *pit*—crater-like area in the surface of the laminate.

3.2.66 *polyester resin*—resin produced by the condensation of dihydroxy glycols and dibasic organic acids or anhydrides. In FRP fabrications, the polyester plastic contains at least one unsaturated constituent and is dissolved in styrene and subsequently reacted to give a highly crosslinked thermoset matrix.

3.2.67 *profile*—the roughness (or smoothness) of a surface that has been prepared for bonding.

3.2.68 *promoter*—a material which activates the catalyst that cures the resin.

3.2.69 *PVA*—abbreviation for polyvinyl alcohol, a widely used parting agent.

3.2.70 *reinforcement*—glass fibers in the form of continuous strand, chopped-strand, or fabric. These fibers are added to the resin matrix to give strength and other properties to the laminate.

3.2.71 *release film*—film used to facilitate removal of the fabricated part from the mold. Oriented polyester film, 3 to 5 mils thick has been found suitable for this purpose.

3.2.72 *resin putty*—resin filled with clay, silica fume, milled fibers, or other inert materials, or both, to yield a material for filling gaps, cracks, and fillets.

3.2.73 *resin richness*—excessive amounts or uneven distribution of resin in the laminate. Such areas are the result of improper wetout or drainage and are prone to cracking.

3.2.74 *roll-out*—densification of the laminate by working reinforcement into and air out of the resin, using a serrated thermoplastic or metal roller.

3.2.75 *roving*—a number of strands or filaments gathered with little or no twist in a package called a roving ball. Also see *woven roving*.

3.2.76 *secondary bond strength*—adhesive force that holds a separately cured laminate to the basic substrate laminate.

3.2.77 *sizing*—surface treatment or coating applied to filaments to improve the filament-to-resin bond.

3.2.78 *spray-up*—method of contact molding where resin and chopped strands of continuous-filament glass fiber are deposited on the mold directly from a chopper gun.

3.2.79 *strain*—elongation per unit length.

3.2.80 *stress*—load per unit area.

3.2.81 *structural layer*—the portion of the laminate having the primary mechanical strength.

3.2.82 *surface preparation*—the act of roughening, priming, or otherwise treating the laminate surface to achieve surface conditions that are conducive to adhesion of a subsequently applied laminate.

3.2.83 *surfacing veil*—a very thin (10 to 20 mils) mat of *C*-glass or synthetic material such as non-woven polyester fabric, used to reinforce the corrosion-resistant resin on the inside or outside surface of the FRP laminate.

3.2.84 *unidirectional roving*—continuous parallel roving held together with periodic stitching.

3.2.85 *vinyl ester resin*—resin characterized by reactive unsaturation, located predominately in terminal positions that can be compounded with styrene and reacted to produce crosslinked copolymer matrices.

3.2.86 *void*—unfilled space caused by air or gas in the resin mix or entrapment of such gases during lay-up of individual plies of glass.

3.2.87 *warp ends*—the strands in a woven roving that run in the longitudinal direction of the fabric, that is, along the roll length of the fabric.

3.2.88 *winding angle*—the angle between the winding strand and the longitudinal axis of the cylindrical liner, sometimes called the helix angle. The winding angle can be determined by measuring the included angle along the longitudinal axis of the pipe at the intersection of strands and dividing this angle by two.

3.2.89 *winding cycle*—the complete covering of the mandrel surface by two bi-directional layers of filament winding. Hoop winding will use one pass; in helical winding many passes are required to complete one winding cycle.

3.2.90 *woven roving*—a plain-weave reinforcement fabric made of rovings. The standard configuration requires five rovings in the warp direction and four rovings in the weft direction and a nominal weight of 24 oz/yd^2 (814 g/m^2).

3.3 *Definitions of Terms Specific to This Standard:*
3.3.1 *can*—an individual fabricated cylindrical liner section.

3.3.2 *quality assurance (QA)*—a system, employed by the owner or his designate, to monitor the manufacturer's quality control and to recognize and resolve any nonconformances. This system is administered by a quality-assurance representative who is empowered to verify the QA and the resolution of all noncompliances.

3.3.3 *quality-assurance program*—a plan that documents the procedures or instructions used to ensure the quality control of the manufacturing process.

3.3.4 *quality control (QC)*—a system of measurements and checks employed to monitor the manufacture of the FRP

TABLE 1 Stress and Modulus of Elasticity Symbols, psi

Description	Stress Type		
	Membrane Tension	Membrane Compression	Bending
Calculated longitudinal	f_z^t	f_z^c	f_z^b
Calculated circumferential	f_θ^t	f_θ^c	f_θ^b
Allowable longitudinal	F_z^t	F_z^c	F_z^b
Allowable circumferential	F_θ^t	F_θ^c	F_θ^b
Ultimate longitudinal	F_z^{tu}	. . .	F_z^{bu}
Ultimate circumferential	F_θ^{tu}	. . .	F_θ^{bu}
Critical buckling, longitudinal	. . .	F_z^{cr}	. . .
Critical buckling, circumferential	. . .	F_θ^{cr}	. . .
Modulus of elasticity, longitudinal	E_z^t	E_z^c	E_z^b
Modulus of elasticity, circumferential	E_θ^t	E_θ^c	E_θ^b

† Editorially corrected column alignment in March 2010.

chimney liner and to assess compliance of manufacture to the critical quality requirements.

3.4 *Symbols:* (*see* Table 1)

a	=	winding angle (with respect to the longitudinal axis of the liner), degree
A_θ	=	hoop membrane stiffness of the liner wall, lb/in.
AT	=	abnormal temperature load
CP	=	circumferential pressure load, psi
D	=	dead load
D_s	=	theoretical draft (without losses), inches of water
D_x, D_θ	=	longitudinal and hoop bending stiffness, of the liner wall, lb-in.²/in.
$(EI)s$	=	transformed flexural stiffness of ring stiffener, lb-in.²
EQ	=	earthquake load
f	=	ovalling natural frequency, cycles per second
g	=	acceleration due to gravity, in/s²
H	=	total height of liner above breeching, ft
h_1	=	flue-gas film coefficient of thermal conductivity, BTU/ft²/in/h/°F
h_3	=	film coefficient of thermal conductivity outside of liner, BTU/ft²/in/h/°F
I	=	center-line moment of inertia of liner section, in.⁴ = $\pi r^3 t$
I_s	=	moment of inertia of one stiffener
k	=	coefficient of thermal conductivity for FRP liner (in absence of data use $k = 2$), BTU/ft²/in/h/°F
k_n	=	knockdown factor
k_R	=	ratio of thermal resistance from gas stream to the middle of the liner wall to the total radial thermal resistance of liner
L	=	distance between lateral supports, ft
L_1	=	spacing between full circumferential stiffeners, in, determined as the sum of half the distance to adjacent stiffeners on either side of the stiffener under consideration
LF	=	load factor

L_L	=	total length of the continuous liner
MRF	=	material resistance factor
P	=	external pressure, psi
p'	=	atmosphere pressure at plant grade level, psi
r	=	average radius of the liner wall, in.
R_1	=	displacement-induced seismic response (force, displacement, or stress)
R_2	=	inertia-induced seismic response (force, displacement, or stress)
R_c	=	radius of the liner to the centroid of the stiffener
R_s	=	radius of the stiffener
R_t	=	total seismic response (force, displacement, or stress)
RF	=	capacity-reduction factor = MRF × TTRF
t	=	thickness of the liner (structural) wall, in.
T	=	normal temperature load,
T_a	=	ambient air temperature, Degrees Fahrenheit
t_c	=	thickness of corrosion barrier, in.
t_e	=	equivalent liner thickness, including stiffener contribution, on basis of equal mass
T_g	=	flue gas temperature, Degrees Fahrenheit
T_m	=	mean liner temperature, $(T_1 + T_2)/2$, Degrees Fahrenheit
T_n	=	annulus air temperature, Degrees Fahrenheit
T_o	=	temperature at inside surface of corrosion barrier, Degrees Fahrenheit
T_1	=	temperature at interface between corrosion barrier and structural layer, Degrees Fahrenheit
T_2	=	temperature at outside surface of structural layer, Degrees Fahrenheit
ΔT_g	=	flue-gas temperature difference across the diameter of the liner, at height z, °F
$(\Delta T_g)_{BASE}$	=	ΔT_g at top of breeching, °F (minimum $T_{gBASE} = 25$°F)
ΔT_m	=	difference of temperature, T_m, across the diameter of the liner, °F
ΔT_w	=	temperature differential across the structural layer, °F $(T_2 - T_1)$
$TTRF$	=	time and temperature reduction factor
W	=	wind load
W_{cm}	=	compressive modulus of elasticity of the winding material (glass), psi
W_L	=	total weight of the continuous liner, including corrosion barrier, and stiffeners
W_{tm}	=	tension modulus of elasticity of the winding material (glass), psi
z	=	distance from top of breeching, in.
α	=	coefficient of thermal expansion in the direction specified by subscript, in./in./°F
μ	=	average Poisson's ratio

$$= (\mu_{z\theta} \times \mu_{\theta z})^{1/2}$$

$\mu_{\theta z}$	=	Poisson's ratio of longitudinal strain to an imposed hoop strain
$\mu_{z\theta}$	=	Poisson's ratio of hoop strain to an imposed longitudinal strain
γ	=	unit weight of liner, lb/in.³
γ_a	=	specific weight of ambient air, lb/ft³
γ_g	=	specific weight of gas, lb/ft³

δ = longitudinal deflection, in.

4. Significance and Use

4.1 This guide provides information, requirements and recommendations for design professionals, fabricators, installers and end-users of FRP chimney liners. FRP is a cost-effective and appropriate material of construction for liners operating at moderate temperatures in a corrosive chemical environment.

4.2 This guide provides uniformity and consistency to the design, fabrication, and erection of fiberglass-reinforced plastic (FRP) liners for concrete chimneys with coal-fired units. Other fossil fuels will require a thorough review of the operating and service conditions and the impact on material selection.

4.3 This guide is limited specifically to FRP liners within a supporting concrete shell and is not applicable to other FRP cylindrical structures.

5. Service and Operating Environments

5.1 *Service Conditions:*

5.1.1 To properly select the optimum design for an FRP chimney liner, it is essential to define the operating and service conditions and the effect they may have on the lining. The chemical, erosion/abrasion, and temperature environments should be determined for the full height of the FRP liner.

5.1.2 Owing to the variability in details of design and system configuration, each FRP liner design must be considered individually. The information given is for coal-fired units, but the general principles are applicable to units fired with other fuels.

5.2 *Environmental Condition*—The environment for a chimney liner is classified as to its chemical, erosion, and temperature condition. Two chemical conditions, three erosion conditions, and four temperature conditions are identified, together with the circumstances in which they usually occur. The combinations of circumstances applicable to a particular chimney liner should be determined.

5.3 *Chemical Environment:*

5.3.1 *Condition 1*—Occasional exposure of certain areas to low pH from acid condensation, occurring with reheated gas or un-scrubbed gas at localized cold areas, such as the liner hood or during start-up.

5.3.2 *Condition 2*—Constant exposure to low pH, acid condensation with concentration based on equilibrium concentration of H_2SO_4, water vapor in the gas stream at temperatures above the water dew point. This operating condition is usually for scrubber systems without reheat, with essentially saturated gas with temperatures from ambient to 140°F (60°C), or when there is insufficient reheat to raise the gas temperature above the acid dew point. Start-up conditions are covered by the operating conditions.

5.4 *Erosion/Abrasion Environment:*

5.4.1 *Condition 1*—Normal-velocity gas flow (45–100 fps (14–31 m/s)) with particulate removal equipment in service. Most particulate removal and flue-gas desulfurization (FGD) systems have velocities in this range.

5.4.2 *Condition 2*—Normal-velocity gas flow with particulate removal equipment out of service. This condition would be infrequent, such as when precipitator electric power is out or when bag houses are bypassed. The duration should be determined, as the plant may reduce load or shut down when such a condition occurs.

5.4.3 *Condition 3*—High-velocity gas flow (higher than 100 fps (31 m/s)), by design, or at sharp corners, turning vanes, and struts. Erosion will likely occur at these locations.

5.5 *Operating Temperature Environment:*

5.5.1 *Condition 1*—Saturated flue gas, ambient to 140°F (60°C). This is the usual operating condition for chimney liners on systems with wet scrubbers without reheat. Start-up conditions are covered by the operating conditions. Where bypass of scrubbers is provided, conditions are described in 5.6.

5.5.2 *Condition 2*—Normal gas temperature from 140 to 200°F (60 to 93°C), with moisture content and acid condensation determined by the individual conditions. This is the usual operating range for wet scrubber systems with reheat. Start-up, high-temperature, and by-pass conditions will be the same as described in 5.6.

5.5.3 *Condition 3*—Normal gas temperature from 140 to 200°F (60 to 93°C), with temperatures high enough for condensation not to occur during normal operation. This is the usual operating range for spray dryer-baghouse and spray dryer-precipitator combinations. Condensation at start-up is minimized by not introducing water to the spray dryers until coal firing is started. Temperatures during by-pass and for excursions are as described in 5.6.

5.5.4 *Condition 4*—Normal gas temperature from 200 to 330°F (93 to 166°C). This is the usual operating range for plants without scrubbers. This condition is also applicable to systems in which the particulate removal or flue-gas desulfurization (FGD) system, or both, can be bypassed, with temperatures determined by the gas flow that can be bypassed compared to the total gas flow of the system.

5.5.5 This guide covers FRP liners for Conditions 1, 2, and 3. Condition 4 is not covered in this guide, although applications over 200°F (93°C) operating temperature condition are in service. Condition 4 requires additional considerations in evaluating materials and composite designs.

5.6 *Abnormal Environments*—Abnormal environments, such as stoppage of an air preheater or malfunction of the scrubber sprays, or both, can result in short-term conditions more severe than those covered. The severity and duration of the abnormal conditions depend on the design and operation of the plant and should be determined for each project. In many cases, these conditions are of short duration because a major upset in the boiler draft system, or in the FGD or particulate removal system, means a reduction in load or plant shutdown to protect the equipment or stay within the emission criteria.

5.6.1 *Condition 1*—Flue-gas-temperature excursion of up to 250°F (121°C) maximum, maintained by a quench system.

5.6.2 *Condition 2*—Flue-gas-temperature excursion up to 400°F (227°C) maximum.

5.6.3 FRP liners may be used for abnormal Condition 1, but its use for Condition 2 is not considered in this guide.

5.6.4 The gas temperature shall be maintained by a quench system at or below a temperature of 250°F (121°C).

5.6.5 In case of a gas-temperature upset 25°F (–4°C) above the established operating temperature, an additional deluge system should be used to bring the gas temperature back to normal operating temperatures.

5.7 *Other Operating and Service Environments:*

5.7.1 Start-up of coal-fired units is usually accomplished with fuel other than coal, such as diesel oil, natural gas, or liquefied natural gas. These fuels, which result in flue-gas compositions different from that produced by coal-firing, should be considered in the design of the liner.

5.7.2 The temperatures given are average temperatures of flue gases entering the chimney liners. Gas temperatures vary as the gas rises up the chimney and at breaching openings, and they vary with the start-up condition of the unit.

5.8 *Static Electricity Build-Up*—FRP in a chimney-liner application is subject to the build-up of static electricity that may be a consideration in some installations. A static-charge dissipation system must be provided where considered necessary (see 6.3.6).

5.9 *Flame Spread*—FRP chimney liners are subject to conditions that propagate flame spread. Specific requirements will vary, depending upon operating and maintenance conditions. However, all FRP liners shall have a flame-resistant resin as in 6.3.5.

6. Materials

6.1 *Raw Materials:*

6.1.1 *Resin:*

6.1.1.1 The selected resin shall be either a polyester or vinylester that provides the properties to withstand the conditions of the operating environment described in Section 5. Resins shall conform to the requirements of Specification C582.

6.1.1.2 FRP chimney liners are fabricated with a flame-retardant resin and, when required, additional flame-retardant synergist added. The resin shall, at minimum, have been demonstrated to withstand 25 % sulfuric acid at 180°F (82°C) for a duration of one year with a minimum retained strength of 50 %, in accordance with Practice C581, or under the actual anticipated environmental-service condition.

6.1.1.3 The resin in the corrosion barrier shall be chosen for its corrosion resistance and flame-retardant properties. Due to physical and mechanical requirements, a different corrosion-resistant resin may be used in the corrosion barrier than in the structural layer.

6.1.2 *Other Additives*—The resin may contain diluents such as added styrene, fillers, dyes, pigments, or flame retardants only when agreed upon between the fabricator and the owner. Such uses shall conform to the descriptions of diluents, resin pastes, and ultraviolet absorbers as explained in Specification C582. Additionally, carbon filler may be added for static-charge dissipation.

6.1.3 *Reinforcements*—Reinforcements shall conform to the requirements of Specification C582 for contact molding and Specification D3299 for filament winding. These specifications require the sizing and binder systems to be compatible with the resins selected.

6.1.3.1 Glass reinforcements shall be Type *E* or *E-CR* glass fibers having a sizing compatible with the resin.

6.1.3.2 The surface veil used in the corrosion barrier should be Type *C* glass fibers, or a synthetic material as approved by the owner. If specified by the purchaser, a carbon veil may be added for static-charge dissipation as in section 6.3.6.

6.2 *Laminate Composition*—FRP chimney-liner laminates consist of a corrosion barrier, a structural layer, and an exterior surface. The FRP composition shall include a thermoset polyester or vinylester resin, reinforced with glass fiber and containing various other raw materials to provide specific properties. The corrosion barrier provides primary corrosion resistance, flame retardant, and shall follow laminate construction described in Specification C582. The structural layer shall primarily provide the mechanical properties and strength of the design. The outer layer shall contain a paraffinated resin to prevent air from inhibiting the cure process and shall provide weather or environmental protection, or both. Liner extending above the chimney cap shall be protected against ultraviolet (UV) rays and in cold weather regions, against ice forming on the liner surfaces from freezing of water droplets in the gas phase.

6.2.1 *Corrosion Barrier*—The corrosion barrier shall be as described in Specification C582. Additional plies of surfacing mat and chopped-strand mat may be used in particularly severe chemical environments, but consideration shall be given to the effects of thermal and mechanical shock.

6.2.2 *Structural Layer*—The structural layer shall meet the physical properties required by the design in Section 7. The fabrication process is typically filament winding, as described in Specification D3299 and Section 8, but may include contact molding, as described in Specification C582, or a combination of both.

6.2.3 *Outer Layer*

6.3 *Laminate Properties:*

6.3.1 *Physical and Mechanical*—The following physical-property test methods are designed for use on entire laminates or individually on the corrosion barrier, the structural layer, or repeating structural units, and external overlays. The following test methods shall be used for determination of initial design data and QA/QC procedures:

6.3.1.1 *Tensile Modulus (Axial Direction)*—Test Method D638 shall be used; or the test results used in conjunction with laminate theory as in section 6.3.1.6.

6.3.1.2 *Flexural Modulus (Axial and Hoop Directions)*—Test Method D790 shall be used; or the test results used in conjunction with laminate theory as in section 6.3.1.6.

6.3.1.3 *Compressive Modulus*—The compressive modulus shall be obtained in accordance with Test Method D695, with the following modifications. The specimens shall be 2 in. (51 mm) in the test direction by 0.5 in. (13 mm) thick with the corrosion barrier removed by machining. Strain shall be measured by the use of an extensometer or other strain gages centered on the specimen in the 2-in. direction. The extensometer arms shall be spaced to 1.5 in. (38 mm) apart at their attachment points to the specimen. The test results may be used in conjunction with laminate theory as in section 6.3.1.6.

6.3.1.4 *Coefficient of Thermal Expansion*—Coefficient of thermal expansion shall be measured in accordance with Test Method E228, over an appropriate temperature range using specimens constructed with the same composition, resin, construction sequence, glass content, type and weight of reinforcement, and cure conditions used in the actual liner. The glass content of the test laminate should be within 5 % of the glass content of the actual chimney-liner laminate. The direction of measurement in relation to the orientation of glass shall be considered in interpretation of the results. Unidirectional roving may be used to approximate filament winding.

6.3.1.5 *Coefficient of Thermal Conductivity*—The coefficient of thermal conductivity shall be determined by Test Method C177 or C518 on representative laminate for either the entire liner laminate to be used, or for each of the following laminate components; that is, corrosion barrier, structure layer, and exterior coating, if any. The representative laminate shall be a flat laminate constructed with the same resin, construction sequence, glass content, type and weight of reinforcements, and cure conditions used in the actual laminate. The direction of measurement in relation to the orientation of glass shall be considered in interpretation of the results. Unidirectional roving may be used to approximate filament winding.

6.3.1.6 Laminate theory may be used instead of physical testing to determine axial tensile, flexural, and compressive moduli only. In such a case, the axial tensile and hoop flexural moduli computed from laminate theory shall be verified by comparison with the results obtained by physical testing. The test results shall be at least 90 % of the computed values.

6.3.2 *Chemical*—The corrosion resistance of the resins used shall have been characterized by either Practice C581 or Test Method D4398. Resins may also be evaluated for vapor exposure in accordance with Practice C581, except that specimens are exposed totally in the vapor space above the liquid returning from the condenser. The resin shall have been deemed acceptable for long-term use in the environments described in Section 5 (Conditions 1 or 2, or both), by either of the above test methods or verifiable actual field environments.

6.3.3 *Erosion/Abrasion*—In areas where Condition 3 erosion/abrasion is expected, resin additives such as silicon carbide may be considered.

6.3.4 *Temperature:*

6.3.4.1 The material properties of the laminate shall be suitable for operating-temperature environments as defined by normal Conditions 1, 2, and 3, and abnormal Condition 1. These conditions define only typical temperature environments. It is essential for the owner to fully provide specific temperature conditions in order to properly select the optimum resin.

6.3.4.2 The maximum operating temperature at the interface between the corrosion barrier and the structural layer shall not exceed the heat-deflection temperature (HDT) of the structural-layer resin. The HDT shall be determined in accordance with Test Method D648.

6.3.4.3 The temperature at the corrosion barrier/structural layer interface may be determined by a thermocouple embedded in the laminate or by correlating gas-stream temperature and thermal gradient through the laminate with the temperature at the corrosion liner/structural layer interface.

6.3.5 *Flame Retardancy*—Selection of the flame spread rating is governed by local building codes. The purchaser shall specify the rating for each location.

6.3.5.1 Flame spread is determined in accordance with Test Method E84 (see Note 2) by using a standard laminate construction as determined in accordance with 4.1.2 of Specification C582. The standard laminate is 0.125 in. (3 mm) thick, flat, reinforced with all mat and has a glass content of 25 to 30 % by weight. Flame-retardant synergists of the type and level used in the actual laminate construction shall be used in this test.

Note 2—This flame-spread rating is based on a laboratory test, which is not necessarily predictive of product performance in a real fire situation, and is therefore not intended to reflect hazards presented by this or any other material under actual fire conditions.

6.3.6 *Static-Charge Dissipation*—Operation of FRP chimney liners can build up significant static charges. This may be a safety hazard to personnel and appropriate grounding shall be considered.

6.3.7 *Grounding System*—Each can in the liner shall have at least two proof tested grounding patches. The patches shall be externally connected to a grounding system conforming to the requirements of NFPA 77 and 77-16. Patches should be approximately 2 feet square containing conductive carbon filler and carbon veil covering a ½ in. bolt that connects to the grounding system. Every bolt patch shall be proof tested (ohmmeter) to assure it will provide an appropriate path to ground.

7. Design

7.1 *Design:*

7.1.1 Standard guidelines and minimum requirements are provided for the structural design of fiber-reinforced plastic (FRP) chimney liners, based on load and resistance-factor design procedures.

7.1.2 The objective is to provide a uniform procedure for computing forces and displacements of the liner based on the present state of the art and science of design of fiber-reinforced plastic liners.

7.1.3 The design is limited to FRP chimney liners supported laterally and vertically by the concrete shell.

7.1.4 The design of a fiber-reinforced plastic liner is an iterative process and similar to most engineering designs, but with the following significant differences:

7.1.4.1 Fiber-reinforced plastic is a composite material and its behavior is different from isotropic materials.

7.1.4.2 The practical variation in the physical properties of the liner material in the thermal and chemical design environments are more than those encountered in most other structural materials.

7.1.5 The scope is limited to chimney liners designed to operate continuously in operating temperature environments defined by Conditions 1, 2, and 3 in Section 5 under normal operating conditions, and for Condition 1 for abnormal environments defined in Section 5. The design of liners operating in

normal-temperature environment, Condition 4, or abnormal Condition 2, is not covered.

7.2 *Assumptions:*

7.2.1 The design procedure is based on the overall stress resultants acting on the wall and on the properties of the laminate.

7.2.2 For the purpose of analysis, the cross section is considered to consist of the average diameter and structural layer thickness. The thickness of the corrosion barrier is excluded. For structure design, properties of the material are based on centerline temperature of the section (that is, average through-wall thickness), except where variation through the thickness is being explicitly considered.

7.2.3 For analysis of the liner behavior due to dead load, wind, earthquake, and thermal loads, the liner may be treated as a beam column and beam theory used for calculating the resultant stresses and displacements.

7.2.4 For the beam-column analysis, the liner may be considered uncoupled (separate) or coupled (jointly) with respect to the concrete column.

7.2.4.1 For the uncoupled system, the liner is considered to be rigidly supported by the concrete column and shall be designed as a continuous beam column. The displacements at points of support or restraint shall equal the independently computed displacements of the concrete shell.

7.2.4.2 For the coupled system, the coupling of the liner and the concrete column system is achieved by incorporating the flexibility of the concrete column in the liner design. A larger combined structure must be analyzed to obtain forces in the liner member(s) as a beam column.

7.2.4.3 The flexibility of the supports and restraints between the liner and the concrete column and the local flexibility of the liner at lateral support points may be incorporated in the analysis of the coupled system.

7.2.5 If the frequency of vibration of the liner, based on beam theory, is within ±20 % of the concrete column frequency, a dynamic analysis of the coupled system shall be made. A dynamic analysis of the coupled FRP liner/concrete column system shall be made, unless it is demonstrated by evaluation of the mass frequency ratios that coupling effects are negligible.

7.2.6 At points of application of loads (that is, at supports and restraints) on the liner beam-column, the liner may be adequately stiffened locally so that the liner roundness is maintained and the liner indeed functions as a beam-column.

7.2.7 Resultant forces are computed from linear elastic analysis of the beam column.

7.2.8 Using the resultant forces from the analysis, the design of the liner is checked against the appropriate material properties of the laminate. The resistance in a certain direction is based on the strength of the laminate in that direction, determined either experimentally or derived from known properties of the constituents.

7.2.9 Most material properties used in the analysis and design are results of short-duration tests at room temperature. Factors are recommended to reduce these property values to a design life of 35 years of the liner in the range of operating temperatures given in Section 5.

7.2.10 The minimum structural-wall thickness of the liner shall be ³⁄₈ in. (10 mm).

7.2.11 Wherever the design requirements for the concrete chimney are cited in accordance with ACI 307, it is implied that requirements of ACI 307, or of the project specification, whichever are more critical, shall be used.

7.2.12 Nothing in this section shall be deemed to prohibit the use of other properly substantiated technical data and procedures for the analysis and design of fiber reinforced plastic liners.

7.3 *Dead Loads (D):*

7.3.1 The dead load shall include the estimated weight of all permanent structures, including stiffeners and attachments. A unit weight of 140 lb/ft³ (2243 kg/m³) is recommended for the calculation of the dead load of the liner (including stiffeners and attachments) using its nominal dimensions.

7.3.2 Where fly-ash is expected to be deposited on the inside surface of the liner, the dead load of the fly-ash shall be considered in the design. Where a wet scrubber is in operation, or can be in operation upstream of the chimney liner, a fly-ash deposit shall be considered to be attached to the inside of the liner. The variation of the thickness of the fly-ash deposit along the height of the liner shall be as shown in Fig. 1. The dead load of the wet fly-ash shall be based on a unit weight of 80 lb/ft³ (1281 kg/m³).

7.3.2.1 Where a wet scrubber is not in operation and no quenching system is used, or used briefly or rarely, the dry
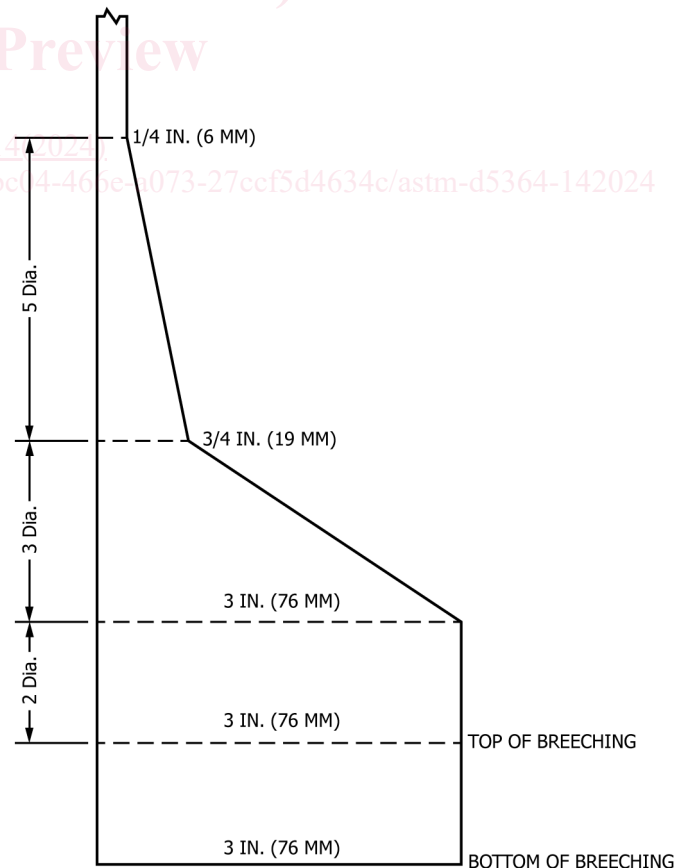

FIG. 1 Fly-Ash Deposit