



Standard Test Methods for Structural Panels in Flexure¹

This standard is issued under the fixed designation D 3043; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

1. Scope

1.1 These test methods determine the flexural properties of strips cut from structural panels or panels up to 4 by 8 ft in size. Structural panels in use include plywood, waferboard, oriented strand board, and composites of veneer and of wood-based layers. Four methods of tests are included:

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The choice of method will be dictated by the purpose of the test, type of material, and equipment availability. All methods are applicable to material that is relative uniform in strength and stiffness properties. Only Method C should be used to test material suspected of having strength or stiffness variations within a panel caused by density variations, knots, knot-holes, areas of distorted grain, fungal attack, or wide growth variations. However, Method B may be used to evaluate certain features such as core gaps and veneer joints in plywood panels where effects are readily projected to full panels. Method C generally is preferred where size of test material permits. Moments applied to fail specimens tested by Method A, B or D in which large deflections occur can be considerably larger than nominal. An approximate correction can be made.

1.2 *Method A, Center-Point Flexure Test*—This method is applicable to material that is uniform with respect to elastic and strength properties. Total deflection, and modulus of elasticity computed from it, include a relatively constant component attributable to shear deformation. It is well suited to investigations of many variables that influence properties uniformly throughout the panel in controlled studies and to test small, defect-free control specimens cut from large panels containing defects tested by the large-specimen method.

1.3 *Method B, Two-Point Flexure Test*—This method, like Method A, is suited to the investigation of factors that influence strength and elastic properties uniformly throughout the panel, in controlled studies, and to testing small, defect free control

specimens cut from large specimens tested by Method C. However, it may be used to determine the effects of finger joints, veneer joints and gaps, and other features which can be placed entirely between the load points and whose effects can be projected readily to full panel width. Deflection and modulus of elasticity obtained from this method are related to flexural stress only and do not contain a shear component. Significant errors in modulus of rupture can occur when nominal moment is used (see Appendix X1).

1.4 *Method C, Pure Moment Test*—This method is ideally suited for evaluating effects of knots, knot-holes, areas of sloping grain, and patches for their effect on standard full-size panels. It is equally well suited for testing uniform or clear material whenever specimen size is adequate. Measured deformation and elastic constants are free of shear deformation effects; and panels can be bent to large deflections without incurring errors from horizontal force components occurring in other methods. Specimen size and span above certain minimums are quite flexible. It is preferred when equipment is available.

1.5 *Method D, Flexure Test for Quality Assurance*—This method, like Method A, is well suited to the investigation of factors that influence bending strength and stiffness properties. Also like Method A, this method uses small specimens in a center-point simple span test configuration. This method uses a span to depth ratio, specimen width, test fixture and test speed that make the method well suited for quality assurance. The method is frequently used for quality assurance testing of oriented strand board.

1.6 All methods can be used to determine modulus of elasticity with sufficient accuracy. Modulus of rupture determined by Methods A, B or D is subject to errors up to and sometimes exceeding 20 % depending upon span, loading, and deflection at failure unless moment is computed in the rigorous manner outlined in Appendix X1 or corrections are made in other ways. These errors are not present in Method C.

1.7 When comparisons are desired between results of specimen groups, it is good practice to use the same method of test for all specimens, thus eliminating possible differences relative to test method.

1.8 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the*

¹ These methods are under the jurisdiction of ASTM Committee D-7 on Wood and are the direct responsibility of Subcommittee D07.03 on Panel Products.

Current edition approved Apr. 10, 2000. Published July 2000. Originally published as D 3043 – 72. Last previous edition D 3043 – 95.

responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.

2. Referenced Documents

2.1 ASTM Standards:

- D 2395 Test Methods for Specific Gravity of Wood and Wood-Base Materials²
- D 4442 Test Methods for Direct Moisture Content Measurement of Wood and Wood-Base Materials²
- D 4761 Test Method for Mechanical Properties of Lumber and Wood-Base Structural Material²

3. Significance and Use

3.1 These methods give the flexural properties, principally strength and stiffness, of structural panels. These properties are of primary importance in most structural uses of panels whether in construction for floors, wall sheathing, roof decking, concrete form, or various space plane structures; packaging and materials handling for containers, crates, or pallets; or structural components such as stress-skin panels.

3.2 To control or define other variables influencing flexure properties, moisture content and time to failure must be determined. Conditioning of test material at controlled atmospheres to control test moisture content and determination of specific gravity are recommended. Comparisons of results of plywood, veneer composites, and laminates with solid wood or other plywood constructions will be greatly assisted if the thickness of the individual plies is measured to permit computation of section properties.

4. Control of Moisture Content

4.1 Structural panel samples to be tested at a specific moisture content or relative humidity shall be conditioned to approximate constant mass in controlled atmospheric conditions before testing. For structural panels used under dry conditions, a relative humidity of $65 \pm 5\%$ at a temperature of $68 \pm 6^\circ\text{F}$ ($20 \pm 3^\circ\text{C}$) is recommended.

5. Method A—Center-Point Flexure Test

5.1 *Summary*—A conventional compression testing machine is used to apply and measure a load at mid-span of a small flexure specimen; and the resulting deflection at mid span is measured or recorded. The test proceeds at a constant rate of head motion until either sufficient deflection data in the elastic range have been gathered or until specimen failure occurs. The specimen is supported on reaction bearings which permit the specimen and bearing plate to roll freely over the reactions as the specimen deflects.

5.2 *Test Specimen*—The test specimen shall be rectangular in cross section. The depth of the specimen shall be equal to the thickness of material, and the width shall be 1 in. (25 mm) for depths less than $\frac{1}{4}$ in. (6 mm) and 2 in. (50 mm) for greater depths (Note 1). When the principal direction of the face plies, laminations, strands, or wafers is parallel to the span, the length of the specimen (Note 2) shall be not less than 48 times the

depth plus 2 in.; when the principal direction of the face plies, laminations, strands, or wafers is perpendicular to the span, the specimen length shall be not less than 24 times the depth plus 2 in. (Note 3).

NOTE 1—In certain specific instances, it may be necessary or desirable to test specimens having a width greater than 1 or 2 in. (25 or 50 mm). To eliminate plate action when wider specimens are tested, the specimen width shall not exceed one third of the span length and precaution shall be taken to ensure uniform bearing across the entire width of the specimen at the load and reaction points.

NOTE 2—In cutting specimens to meet the length requirement, it is not intended that the length be changed for small variations in thickness. Rather, it is intended that the nominal thickness of the material under test should be used for determining the specimen length.

5.2.1 *Measurements*—Measure specimen thickness at mid-span at two points near each edge and record the average. Measure to the nearest 0.001 in. (0.02 mm) or 0.3 %. Measure width at mid-span to the nearest 0.3 %.

5.2.1.1 When needed for interpretation of test results for plywood, veneer composites, and laminates measure thickness of each layer to the nearest 0.001 in. (0.02 mm) at mid-span at each edge and record the average.

5.3 *Span*—The span shall be at least 48 times the nominal depth when the principal direction of the face plies, laminations, strands, or wafers of the test specimen is parallel to the span and at least 24 times the nominal depth when the principal direction of the face plies, laminations, strands, or wafers is perpendicular to the span (Note 3).

NOTE 3—Establishment of a span-depth ratio is required to allow an accurate comparison of test values for materials of different thicknesses. It should be noted that the span is based on the nominal thickness of the material and it is not intended that the spans be changed for small variations in thickness.

5.4 *End Supports*—Reaction points shall be capable of freely compensating for warp of the test specimen by turning laterally in a plane perpendicular to the specimen length so as to apply load uniformly across its width. Design of end supports shall place the center of rotation near the neutral axis of the specimen of average thickness. Construction is shown in detail in Fig. 1. Bearing points shall be rounded where they contact the specimen.

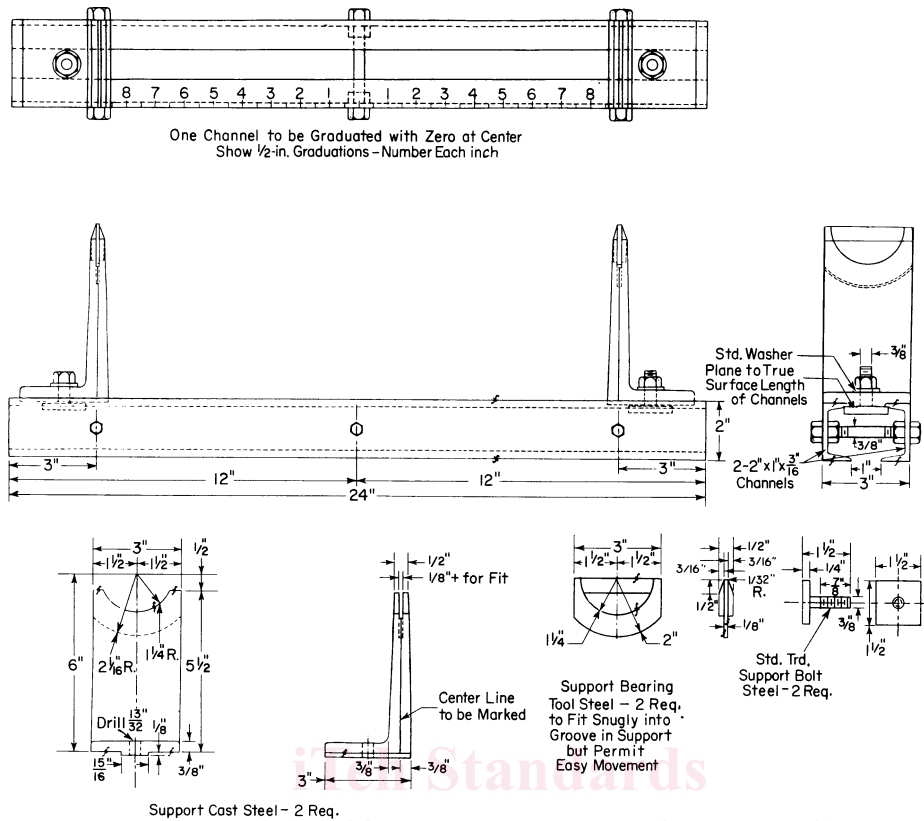
5.4.1 Use of bearing plates is generally recommended and is required wherever significant local deformation may occur.

5.4.2 Use of roller bearings or plates and rollers to preclude friction forces between end support and specimen is recommended in addition to the requirement of lateral compensation. Construction of a suitable end support using small roller bearings in conjunction with a plate which clips to the end of the specimen is illustrated in Fig. 2 and Fig. 3. The use of a large ball bearing to provide lateral compensation for warp is also illustrated. This method is particularly recommended for thin specimens and small loads.

5.4.3 As the specimen deflects during test, loads no longer act in the direction assumed in formulas for calculating properties. For a discussion of these errors, their effects, and methods for reducing them, refer to Appendix X1.

5.5 *Loading Block*—A loading block having a radius of curvature of approximately one and one-half times the depth of the test specimen for a chord length of not less than twice the

² Annual Book of ASTM Standards, Vol 04.10.



Inch-Pound (in.)	Metric Equiva- lents, (mm)	Inch-Pound (in.)	Metric Equiva- lents, (mm)
1/16	1.5	1/4	32
1/8	3	1/2	38
3/16	5	2	50
1/4	6	2 1/16	52
5/16	10	3	76
3/8	10.3	5 1/2	140
7/8	23	6	152
15/16	24	12	305
1	25	24	610

FIG. 1 Apparatus for Static Bending Test Showing Details of Laterally Adjustable Supports

depth of the specimen shall be used. In cases where excessive local deformation may occur, suitable bearing plates shall be used. Radius of curvature of bearing plate or block shall not be so large as to cause bridging as the specimen bends.

5.6 *Loading Procedure*—Apply the load with a continuous motion of the movable head throughout the test. The rate of load application shall be such that the maximum fiber strain rate is equal to 0.0015 in./in. (mm/mm) per min within a permissible variation of $\pm 25\%$. Load shall be measured to an accuracy of $\pm 1\%$ of indicated value or 0.4 percent of full scale, whichever is larger. Calculate the rate of motion of the movable head as follows:

$$N = zL^2/6d \quad (1)$$

where:

N = rate of motion of moving head, in./min (mm/min),

L = span, in. (mm),

d = depth of beam, in. (mm), and

z = unit rate of fiber strain, in./in.·min (mm/mm·min) of outer fiber length = 0.0015.

5.6.1 Measure the elapsed time from initiation of loading to maximum load and record to the nearest 1/2 min.

5.7 *Measurement of Deflection*—Take data for load-deflection curves to determine the modulus of elasticity, proportional limit, work to proportional limit, work to maximum load, and total work. Take deflections by the methods indicated in Fig. 4 or Fig. 5, and take readings to the nearest 0.001 in. (0.02 mm). Choose increments of load so that not less

LARGE DIAMETER BALL BEARING COMPENSATES FOR WARPED SPECIMENS

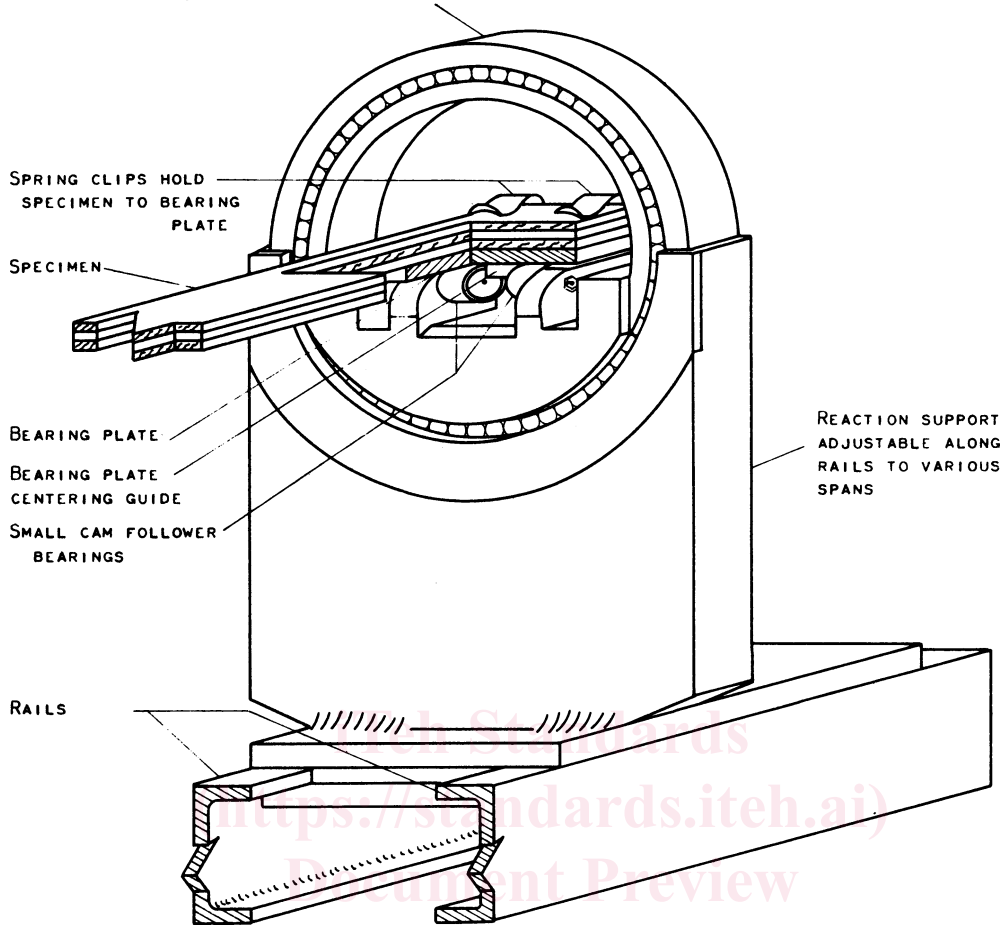


FIG. 2 Reaction Bearing for Small Flexure Test Specimens

ASTM D3043-00

than 12 and preferably 15 or more readings of load and deflection are taken to the proportional limit.

5.7.1 Deflections also may be measured with transducer-type gages and plotted simultaneously against load. In this case, record deflection to an accuracy of at least 1½ % of deformation at proportional limit and the recorded trace below the proportional limit shall be at least 2½ in. (64 mm) long or ¼ of full scale measured on the deformation axis, whichever is larger. Similar requirements apply to the load axis.

5.8 Calculations:

5.8.1 Calculate specimen bending stiffness as follows:

$$EI = (L^3/48)(P/\Delta) \tag{2}$$

where:

- EI = modulus of elasticity, psi (MPa) × moment of inertia, in.⁴ (or mm⁴),
- P/Δ = slope of load—deflection curve, lbf/in. (N/mm),
- I = moment of inertia, in.⁴ (mm⁴), and
- L = span, in. (mm).

5.8.1.1 Moment of inertia used in the computations in 5.8.1 may be calculated in several different ways depending upon the requirements of the investigation. It may be based on the entire cross section, may include only the moment of inertia of layers parallel to span, or may include all layers weighted in accordance with modulus of elasticity in the direction of bending

stress. State clearly the method employed in the report.

5.8.2 Calculate maximum moment ($S_b I/c$) by the following equation:

$$S_b I/c = PL/4 \tag{3}$$

where:

- $S_b I/c$ = maximum moment, lbf-in. (N-mm),
- S_b = modulus of rupture, psi (MPa),
- P = maximum load, lbf (N), and
- c = distance from neutral axis to extreme fiber, in. (mm).

6. Method B—Two-Point Flexure Test

6.1 Summary—The ends of a two-point flexure specimen are supported on special reaction bearings which in turn rest on the table of a conventional testing machine. A pivoted loading device applies equal loads at points ¼ of span from the reactions resulting from downward motion of the testing machine crosshead, and subjects the middle half of the specimen to conditions of nearly pure moment. Deflection of mid span relative to two points just inside the load points is measured with a dial gage or transducer thus giving deformation due to pure bending and unaffected by shear deformation.

6.2 Test Specimen—The test specimen shall be rectangular in cross section and its length shall exceed by 2 in. (50 mm) the



FIG. 3 Apparatus for Two-Point Loading and Measurement of Deflection (Method B)

span on which it is to be tested as determined in 6.3. Thickness shall be the thickness of the material. Width shall be 1 in. (25 mm) for material less than 1/4 in. (6 mm) thick and 2 in. for material 1/4 in. and over in thickness. The alternate width is 12 in. (300 mm).

6.2.1 *Measurements*—Measure specimen thickness at mid-span at two points near each edge and record the average. Measurements shall be to the nearest 0.001 in. (0.02 mm) or 0.3 % . Measure width at mid-span to the nearest 0.3 % .

6.2.1.1 When needed for interpretation of test results for plywood, veneer composites, and laminates, measure thickness of each layer to the nearest 0.001 in. (0.02 mm) at mid-span at each edge and record the average.

6.3 *Span*—Span-depth ratio has relatively little influence on the results of tests using two-point loading and the method of measuring deformation described for it in this standard. However, it is important that the distance between load point and adjacent support be sufficient to prevent rolling shear failures. The alternate 12-in. (300-mm) width will have a midlength (constant moment section) at least 12 in. in length.

6.3.1 Specimens tested for stiffness only shall have a span at least 48 times nominal thickness if the principal direction is parallel to span and 24 times nominal thickness if the principal direction is perpendicular to span.

6.3.2 It is recommended that two-point loading tests to failure be made on a span at least equal to the spacing between

load points plus 48 times specimen thickness or 24 times specimen thickness for the principal direction parallel or perpendicular respectively. Material having high rolling shear strength or having all its plies, laminations, strands, or wafers parallel to span may use closer spacing between loads and supports.

6.4 *Supports*—Reaction supports shall meet the requirements of 5.4 and 5.4.1. Other comments as well as those of 5.4.2 and 5.4.3 apply.

6.5 *Loading*—Apply two equal loads to the specimen equidistant from the supports by cylindrical surfaces having a radius of curvature of at least 1 1/2 times specimen thickness wherever it may contact the specimen. The axes of these surfaces shall remain parallel and at least one of them shall be free to turn about its axis or be loaded through rollers to prevent the application of friction forces to the surface of the specimen. Construction of a satisfactory loading head is illustrated in Fig. 6 and Fig. 3. Locate the pivot point that equalizes the two loads near the original neutral axis of the specimen.

6.5.1 Space load points sufficiently to provide a deflection which can be adequately measured. A spacing of at least 24 and 12 times specimen thickness is recommended for specimens with the principal direction parallel and perpendicular to span respectively.

6.5.2 Measure the sum of the two loads to an accuracy of at least 1 % of indicated value or 0.4 % of full scale, whichever is larger.

6.6 *Speed of Test*—Apply load at a continuous rate of motion of the load points with respect to the supports within a permissible range of 25 % of the rate determined as follows:

$$N = (za/3d)(3L - 4a) \quad (4)$$

where:

N = rate of motion, in./min (mm/min),

z = unit rate of fiber strain, in./in.·min (mm/mm·min) = 0.0015,

a = distance from support to adjacent load, in. (mm),

d = depth of beam, in. (mm), and

L = span, in. (mm).

6.6.1 Measure the elapsed time from initiation of loading to maximum load and record to the nearest 1/2 min.

6.7 *Measurement of Deflection*—Measure deflection of mid-span with respect to a line between two points equidistant from mid-span and just inside the two load points to an accuracy of at least 1 1/2 % of total deflection if tested for stiffness only, or 1 1/2 % of deflection at approximate proportional limit. All three points shall lie on the longitudinal axis of the specimen. Suitable equipment of the transducer type is illustrated in Fig. 6 and shown in Fig. 3. A dial gage could replace the transducer for manual reading. If individual gage readings are taken, at least 12 and preferably 15 or more load and deflection readings shall be taken below approximate proportional limit or for determining specimen stiffness.

6.8 *Calculations:*

6.8.1 Calculate the specimen bending stiffness as follows:

$$EI = [(L - L_1)L_2^2/32](P'/\Delta) \quad (5)$$

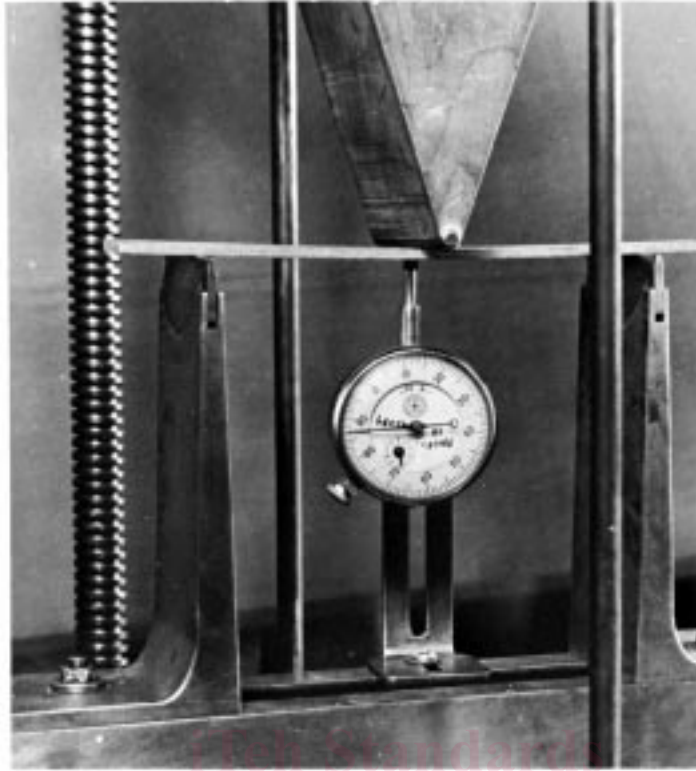


FIG. 4 Static Bending Test Showing Adjustable Supports and One Method of Attaching Dial Gage for Observing Deflection of Thin Material

where:

L_1 = span between load points, in. (mm)

L_2 = span between deflection measurement points, in. (mm),

P'/Δ = slope of load deflection curve where deflection is mid-span relative to ends of span L_2 , in. (mm), and other notation is as given in 5.8.1. Remarks of 5.8.2 apply.

6.8.2 Calculate maximum moment of the specimen as follows:

$$S_b I/c = P(L - L_i)/4 \quad (6)$$

where:

P = maximum load, lbf (N),

7. Method C—Pure Moment Test

7.1 *Summary*—A specially designed testing machine applies pure moments to opposite ends of the test panel through loading frames. Frames are free to move toward or away from each other during the test to preclude application of other than pure moments to the center span of the panel. Between loading frames deflection of the neutral axis follows a circular arc. Rotational deformation between points near the ends of the arc is measured during the test by special sensing gages resting on pins projecting from the face of the panel at these points. The test is simple and flexible, and results are directly relatable to basic properties at large deformations.

7.2 *Test Specimen*—Specimens shall be of a size comparable to that of the material in use, frequently consisting of the

entire panel. Limitation on size may be imposed by equipment size or moment capacity or size of available material. Except for effects of nonuniformity of properties within a panel, specimen dimensions do not tend to influence test results.

When nonuniform material containing density variation, knots, knot-holes, sloping grain or other sources of large variability is tested for general construction and industrial use, a minimum specimen width of 24 in. (610 mm) is recommended and in no case shall width be less than 12 in. (300 mm).

7.2.1 *Measurements*—Measure panel thickness at four points, two on each edge one fourth of panel length from each end, to the nearest 0.001 in. (0.02 mm) and record the average. Measure width to the nearest 0.3 % at two points one fourth of panel length from each end and record the average.

7.2.1.1 When needed for interpretation of test results for plywood, veneer composites, and laminates measure thickness of each layer to the nearest 0.001 in. (0.02 mm) at the same points at which total panel thickness is measured.

7.3 *Application and Measurement of Moments*—Fig. 7 illustrates application of pure moments to a specimen, by means of loading frames, and measurement of deformation. Apply equal and opposite pure moments to each end of the panel by frames. The frames shall be free to move toward or away from each other while under load to preclude application of direct tension or compression loads at large panel deformations. Support axes of the loading frames to remain in a parallel relationship throughout the test (Note 4). Space bars of the loading frames sufficiently to prevent shear failures between