

### SLOVENSKI STANDARD SIST EN 3872:2005

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#### Aerospace series - Aluminium alloy AL-R39002-H112 - Die forgings - a <200 mm

Aerospace series - Aluminium alloy AL-R39002-H112 - Die forgings - a <200 mm

Série aérospatiale - Alliage d'aluminium AL-R39002-H112 - Pieces matricées - a <200 mm

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Ta slovenski standard je istoveten z: EN 3872-2005

ICS:

49.025.20 Aluminij Aluminium

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EUROPEAN STANDARD NORME EUROPÉENNE EUROPÄISCHE NORM **EN 3872** 

June 2005

ICS 49.025.20

#### **English version**

## Aerospace series - Aluminium alloy AL-R39002-H112 - Die forgings - a ≤200 mm

Série aérospatiale - Alliage d'aluminium AL-R39002-H112 - Pièces matricées - a ≤200 mm Luft- und Raumfahrt - Aluminiumlegierung AL-R39002-H112 - Gesenkschmiedestücke - a ≤200 mm

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This European Standard exists in three official versions (English, French, German). A version in any other language made by translation under the responsibility of a CEN member into its own-language and notified to the Central Secretariat has the same status as the official versions.

CEN members are the national standards bodies of Austria, Belgium, Cyprus, Czech Republic, Denmark, Estonia, Finland, France, Germany, Greece, Hungary, Iceland, Ireland, Italy, Latvia, Lithuania, Luxembourg, Malta, Netherlands, Norway, Poland, Portugal, Slovakia, Slovenia, Spain, Sweden, Switzerland and United Kingdom.

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EUROPEAN COMMITTEE FOR STANDARDIZATION COMITÉ EUROPÉEN DE NORMALISATION EUROPÄISCHES KOMITEE FÜR NORMUNG

Management Centre: rue de Stassart, 36 B-1050 Brussels

#### **Foreword**

This document (EN 3872:2005) has been prepared by the European Association of Aerospace Manufacturers - Standardization (AECMA-STAN).

After enquiries and votes carried out in accordance with the rules of this Association, this Standard has received the approval of the National Associations and the Official Services of the member countries of AECMA, prior to its presentation to CEN.

This European Standard shall be given the status of a national standard, either by publication of an identical text or by endorsement, at the latest by December 2005, and conflicting national standards shall be withdrawn at the latest by December 2005.

Attention is drawn to the possibility that some of the elements of this document may be the subject of patent rights. CEN [and/or CENELEC] shall not be held responsible for identifying any or all such patent rights.

According to the CEN/CENELEC Internal Regulations, the national standards organizations of the following countries are bound to implement this European Standard: Austria, Belgium, Cyprus, Czech Republic, Denmark, Estonia, Finland, France, Germany, Greece, Hungary, Iceland, Ireland, Italy, Latvia, Lithuania, Luxembourg, Malta, Netherlands, Norway, Poland, Portugal, Slovakia, Slovenia, Spain, Sweden, Switzerland and the United Kingdom.

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#### Introduction

This standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This standard has been prepared in accordance with EN 4500-2.

#### 1 Scope

This standard specifies the requirements relating to:

Aluminium alloy AL-R39002 H112 Die forgings  $a \le 200 \text{ mm}$ 

for aerospace application.

## 2 Normative references PREVIEW

The following referenced documents are indispensable for the application of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies. EN 3872:2005

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EN 2082-3, Aerospace series — Aluminium4alloy, forging7stock5and forgings — Technical specification — Part 3: Pre-production and production forgings.

EN 4258, Aerospace series — Metallic materials — General organization of standardization — Links between types of EN standards and their use.

EN 4292, Aerospace series — Aluminium alloy AL-R39002 — Forging stock. 1)

EN 4500-2, Aerospace series — Metallic materials — Rules for drafting and presentation of material standards — Part 2: Specific rules for aluminium, aluminium alloys and magnesium alloys. 1)

EN 6018, Aerospace series — Test methods for metallic materials — Determination of density according to displacement method. 1)

<sup>1)</sup> Published as AECMA Prestandard at the date of publication of this standard.

1	Material designation		Aluminium alloy AL-R39002										
2	Chemical	Element		С	Si	0	Fe	Mg	Li	Others		Al	
	composition			O	OI .	J	10	IVIG	Li	Each	Total	Al	
	%	min.		0,25	-	0,20	ı	4,0	1,2	1	ı	Base	
		max.		0,45	0,20	0,7	0,30	5,5	1,4	0,05	0,15	Dasc	
3	Method of melting						-						
4.1	Form		Die forgings										
4.2	Method of production			Forged from forging stock EN 4292									
4.3	Limit dimension(s) mm			a ≤ 200									
5	5 Technical specification			EN 2082-3									

6.1	Delivery condition	H112
	Heat treatment	-
6.2	Delivery condition code	U
7	Use condition	H112
	Heat treatment	Delivery condition

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8.1	1 Test sample(s)				See EN 2082-3.					
8.2	Te	est piece(s)		https	SIST EN 3872:2005 See EN 2082-3. ://standards.iteh.ai/catalog/standards/sist/26aa9893-6bf1-465b-ab96-					
8.3	He	eat treatment		***17*	0a0eb5114ee1/sist-en-38 <b>/9se 200d</b> ition.					
9	Dimensions concerned mm			mm	a ≤ 200					
10	Thickness of cladding on each face %			%	-					
11	Di	rection of test piece	;		L	LT	ST			
12		Temperature	θ	°C	Ambient	Ambient	Ambient			
13		Proof stress	R <sub>p0,2</sub>	MPa	≥ 370	≥ 360	≥ 350			
14	Т	Strength	R <sub>m</sub>	MPa	≥ 455	≥ 445	≥ 435			
15		Elongation	Α	%	≥ 6	≥ 5	≥ 3			
16		Reduction of area	Z	%		-				
17	17 Hardness				-					
18	8 Shear strength R <sub>c</sub> MPa			MPa	-					
19	19 Bending k –			-	<del>-</del>					
20	20 Impact strength				<del>-</del>					
21		Temperature $\theta$		°C	-					
22		Time h			-					
23	С	Stress	$\sigma_{\text{a}}$	MPa		-				
24	0	Elongation	а	%		-				
25		Rupture stress	$\sigma_{R}$	MPa		-				
26		Elongation at rupture	Α	%		-				
27	No	otes (see line 98)				-				

Stress corrosion	2 3			See EN 2				
	3			The "capability of				
			The "capability clause" applies					
	_	a > 25 mm						
	6	σ = 290 MPa						
	7	t ≥ 20 d						
racture toughness	-	See EN 2082-3.						
ς <sub>ις</sub>	7	Dimensions mm	L-T MPa $\sqrt{m}$		T-L MPa√ <i>m</i>	S-L MPa√ <i>m</i>		
		<i>a</i> ≤ 200 ≥ 26 ≥			≥ 25	≥ 24		
External defects	-	See EN 2082-3.						
nternal defects	-	See EN 2082-3.						
Density	1	EN 6018						
	2	The "capability clause" applies						
	7	$\rho \le 2,60 \text{ kg/dm}^3$						
Batch uniformity	_							
	7			150 (typical value)				
		Hardness	НВ	δ ≤ 10 per		$\Delta \le$ 15 per batch		
https://stanc	ards.				65b-ab96-			
5 Marking inspection				See EN 2	082-3.			
B Notes				-	-			
9 Typical use								
Ω (A)	atch uniformity  iTe  https://stance	atch uniformity  ITeh S  https://stand.ards.	tensity  1 2 7 atch uniformity  - 7 Hardness  ITeh STANDAF (standard)  SISTEN 38 https://standards.iteh.ai/catalog/standard 0a0eb5114ce1/sist	atch uniformity  Thardness  ITeh STANDARD I (standards.ite)  SIST EN 3872:2006  https://standards.iteh.ai/catalog/standards/sist/200a0eb5114ee1/sist-en-387.	1	1		

100	_	Product qualification	-	See EN 2082-3.
				Qualification programme to be agreed between manufacturer and purchaser.
		l	iTe	Ph STANDARD PREVIEW (standards.iteh.ai)  SIST EN 3872-2005  ndards.iteh.ai/catalog/standards/sist/26ag/9893-6bf)-465b-ab96-0a0eb5114ee1/sist-en-3872-2005