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**Cylindrical gears — Calculation of service
life under variable load — Conditions for
cylindrical gears in accordance with
ISO 6336**

*Engrenages cylindriques — Calcul de la durée de vie en service sous
charge variable — Conditions pour les engrenages cylindriques
conformément à l'ISO 6336*

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Foreword

ISO (the International Organization for Standardization) is a worldwide federation of national standards bodies (ISO member bodies). The work of preparing International Standards is normally carried out through ISO technical committees. Each member body interested in a subject for which a technical committee has been established has the right to be represented on that committee. International organizations, governmental and non-governmental, in liaison with ISO, also take part in the work. ISO collaborates closely with the International Electrotechnical Commission (IEC) on all matters of electrotechnical standardization.

The main task of technical committees is to prepare International Standards, but in exceptional circumstances a technical committee may propose the publication of a Technical Report of one of the following types:

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- type 1, when the required support cannot be obtained for the publication of an International Standard, despite repeated efforts;
 - type 2, when the subject is still under technical development or where for any other reason there is the future but not immediate possibility of an agreement on an International Standard;
 - type 3, when a technical committee has collected data of a different kind from that which is normally published as an International Standard (“state of the art”, for example).

Technical Reports of types 1 and 2 are subject to review within three years of publication, to decide whether they can be transformed into International Standards. Technical Reports of type 3 do not necessarily have to be reviewed until data they provide are considered to be no longer valid or useful.

ISO/TR 10495, which is a Technical Report of type 2, was prepared by Technical Committee ISO/TC 60, *Gears*, Subcommittee SC 2, *Gear capacity calculation*.

Annexes A and B of this Technical Report are for information only.

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Cylindrical gears — Calculation of service life under variable load — Conditions for cylindrical gears in accordance with ISO 6336

1 Scope

This Technical Report is concerned with the calculation of service life (or safety factors for a required life) of gears subject to variable loading. Clauses 4 and 5 give a general discussion of the subject; clauses 6 to 8 present a method which may be conveniently applied at the design stage. Whilst the method is presented in terms of ISO 6336, it is equally applicable to other gear stress calculations (e.g BS 436, DIN 3990, NF E23-015).

2 Normative references

The following standards contain provisions which, through reference in this text, constitute provisions of this Technical Report. At the time of publication, the editions indicated were valid. All standards are subject to revision, and parties to agreements based on this Technical Report are encouraged to investigate the possibility of applying the most recent editions of the standards indicated below. Members of IEC and ISO maintain registers of currently valid International Standards.

ISO 701:1976, *International gear rotation - Symbols for geometrical data*.

ISO 1122-1: 1983, *Glossary of gear terms - Part 1: Geometrical definitions*.

ISO 6336-1: 1996, *Calculation of load capacity of spur and helical gears - Part 1: Basic principles, introduction and general influence factors*.

ISO 6336-2: 1996, *Calculation of load capacity of spur and helical gears - Part 2: Calculation of surface durability (pitting)*.

ISO 6336-3: 1996, *Calculation of load capacity of spur and helical gears - Part 3: Calculation of tooth bending strength*.

ISO 6336-5: 1996, *Calculation of load capacity of spur and helical gears - Part 5: Strength and quality of material*.

3 Definitions, symbols, quantities and units

For the purposes of ISO TR 10495, the definitions given in ISO 1122-1 apply. Symbols are based on those given in ISO 701. Only symbols for quantities used in ISO TR 10495 are given in table 1.

Table 1 - Symbols used within ISO TR 10495

| Symbol | Quantity | Unit |
|----------------------------|--------------------------------------------------------------------------|-------------------------|
| b | Facewidth | mm |
| d_1 | Reference diameter of pinion | mm |
| e | Inclination of S-N curve | -- |
| i | Class | -- |
| l | Class interval | -- |
| K_A | Application factor | -- |
| $K_{F\alpha}$ | Transverse load distribution factor (bending stress) | -- |
| $K_{F\beta}$ | Face load distribution factor (bending stress) | -- |
| $K_{H\alpha}$ | Transverse load distribution factor (contact stress) | -- |
| $K_{H\beta}$ | Face load distribution factor (contact stress) | -- |
| K_V | Dynamic factor | -- |
| m_n | Normal module | mm |
| n_i | Number of cycles at i th stress level (number of counts in class i) | -- |
| n_l | Number of cycles at class interval level l | -- |
| N_l | Number of cycles to failure at class interval level l | -- |
| N_L | Number of cycles to failure | -- |
| S | Safety factor for stress | -- |
| $S_{F \text{ lim}}$ | Safety factor for bending stress (min.) | -- |
| $S_{H \text{ lim}}$ | Safety factor for contact stress (min.) | -- |
| T_i | Torque class | Nm |
| T_l | Pinion torque at top of class interval | Nm |
| u | Gear ratio | -- |
| U | Miner sum | -- |
| U_l | Individual damage part of class interval | -- |
| Y_F | Tooth form factor | -- |
| Y_{NT} | Tooth root stress life factor for standard test conditions | -- |
| $Y_{R \text{ rel T}}$ | Relative surface condition factor (root) | -- |
| Y_S | Stress correction factor | -- |
| Y_{ST} | Stress correction factor for the dimension of the standard test gears | -- |
| Y_X | Size factor (bending stress) | -- |
| Y_β | Helix angle factor (bending stress) | -- |
| $Y_{\delta \text{ rel T}}$ | Relative notch sensitivity factor | -- |
| $Z_{B,D}$ | Single pair tooth contact factor for pinion or gear | -- |
| Z_E | Elasticity factor | $(\text{N/mm}^2)^{1/2}$ |
| Z_H | Zone factor | -- |
| Z_L | Lubricant influence factor | -- |
| Z_{NT} | Contact stress life factor for standard test conditions | -- |
| Z_R | Roughness factor | -- |
| Z_V | Speed factor | -- |
| Z_W | Hardness ratio factor | -- |
| Z_β | Helix angle factor (contact stress) | -- |
| Z_c | Contact ratio factor (contact stress) | -- |
| $\sigma_{F \text{ lim}}$ | Nominal stress number (bending) | N/mm^2 |
| σ_{Fl} | Tooth root stress at class interval l | N/mm^2 |
| σ_{FP} | Permissible tooth root stress | N/mm^2 |
| $\sigma_{H \text{ lim}}$ | Allowable stress number (contact) | N/mm^2 |
| σ_{Hl} | Contact stress at class interval l | N/mm^2 |
| σ_{HO} | Nominal contact stress | N/mm^2 |
| σ_{HP} | Permissible contact stress | N/mm^2 |
| σ_l | Stress at class interval l | N/mm^2 |
| σ_{lim} | Allowable stress | N/mm^2 |
| σ_p | Permissible stress | N/mm^2 |

4 Introduction

4.1 Determination of load and stress spectra

Variable loads resulting from a working process, starting process or from operation at or near a critical speed will cause varying stresses at the gear teeth of a drive system. The magnitude and frequency of these loads depend upon the driven machine(s), the driver(s) or motor(s) and the mass elastic properties of the system.

These variable loads (stresses) may be determined by one or more of the procedures listed below:

- Experimental measurement of the operating loads at the machine in question;
- Estimation of the spectrum, if this is known, for a similar machine with similar operating mode;
- Calculation, using known external excitation and a mass elastic simulation of the drive system.

NOTE – Specific data, relevant for the method by which the load or torque measurements are performed, should be marked at the registered results.

To obtain the spectra, the range of the measured (evaluated) loads is divided into classes. A widely used number of classes is 64.

The cycle count for the load class corresponding to the load value for the highest loaded tooth is incremented at every load repetition. Table 2 shows as an example how to apply the torque classes defined in figure 1 to specific torque levels and correlated numbers of cycles.

Table 2 – Example (see figure 1): Classes 111 & 112

| Torque class, T_i , Nm | Number of cycles, n_i |
|--------------------------|-------------------------|
| $440 \leq T_{111} < 444$ | $n_{111} = 2338$ |
| $444 \leq T_{112} < 448$ | $n_{112} = 4318$ |

The torques used to evaluate tooth loading should include the dynamic effects at different rotational speeds.

This spectrum is only valid for the measured or evaluated time period. If the spectrum is extrapolated to represent the required life time, the possibility that there might be torque peaks not frequent enough to be evaluated in that measured spectrum must be considered. These peaks may have an effect on the gear life.

Stress spectra concerning bending and pitting can be obtained from the load (torque) spectrum by using Method II.

Scuffing resistance must be calculated from the worst combination of speed and load.

Wear is a continuous deterioration of the tooth flank and must be considered separately.

Tooth root stress can also be measured by means of strain gages in the fillet. In this case, the derating factors should be taken into account using the results of the measurements. The relevant contact stress can be calculated from the measurements.

4.2 General calculation of service life

The calculated service life is based on the theory that every load cycle (every revolution) is damaging to the gear. The amount of damage depends on the stress level and can be considered as zero for lower stress levels.

The calculated bending or pitting strength fatigue life of a gear is a measure of its ability to accumulate discrete damage until failure occurs.

The fatigue life calculation needs:

- The stress spectrum;
- Material fatigue properties;
- A damage accumulation method.

| T_l in Nm | [+0...+4[| [+4...+8[| [+8...+12[| [+12...+16[| [+16...+20[| [+20...+24[|
|-------------|-----------|-----------|------------|-------------|-------------|-------------|
| 0,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 24,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 48,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 72,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 96,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 120,00 | 0 | 706 | 3469 | 3081 | 5109 | 32 |
| 144,00 | 1 | 2 | 438 | 381 | 756 | 903 |
| 168,00 | 2 | 0 | 0 | 0 | 0 | 1 |
| 192,00 | 45 | 350 | 212 | 616 | 16 | 0 |
| 216,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 240,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 264,00 | 0 | 0 | 0 | 0 | 19 | 2108 |
| 288,00 | 2072 | 3933 | 4257 | 6 | 2 | 3 |
| 312,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 336,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 360,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 384,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 408,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 432,00 | 26 | 72 | 2338* | 4318* | 3665 | 1824 |
| 456,00 | 239 | 477 | 2553 | 3216 | 5576 | 2109 |
| 480,00 | 932 | 90 | 420 | 1913 | 2877 | 2891 |
| 504,00 | 1255 | 449 | 67 | 791 | 745 | 2166 |
| 528,00 | 651 | 518 | 23 | 1 | 0 | 0 |
| 552,00 | 0 | 0 | 8 | 24 | 127 | 520 |
| 576,00 | 751 | 713 | 295 | 42 | 0 | 0 |
| 600,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 624,00 | 0 | 0 | 0 | 0 | 0 | 3 |
| 648,00 | 218 | 187 | 329 | 469 | 34 | 0 |
| 672,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 696,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 720,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 744,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 768,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 792,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 816,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 840,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 864,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 888,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 912,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 936,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 960,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 984,00 | 0 | 0 | 0 | 0 | 0 | 0 |
| 1008,00 | 0 | 0 | 0 | 0 | 0 | 0 |

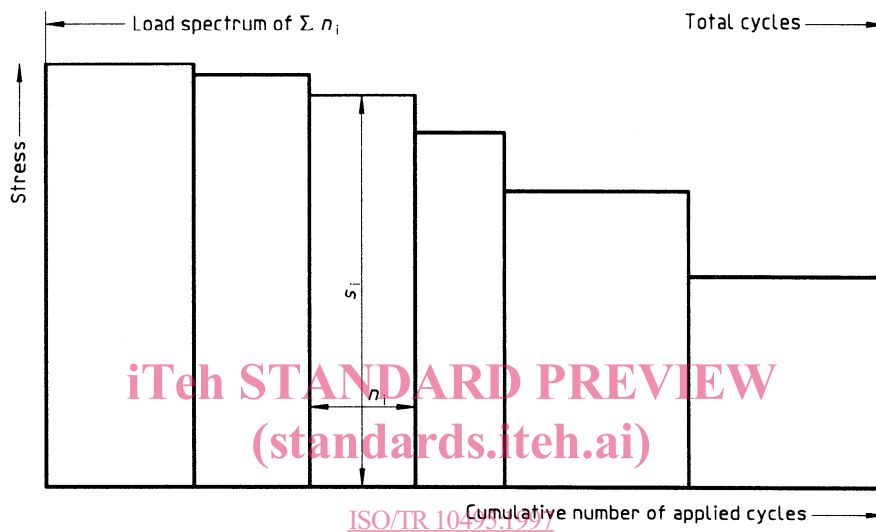
* example presented in table 2

Figure 1 – Torque spectrum (class number = 258)

The stress spectrum is discussed in clause 6.1.

Strength values based on material fatigue properties are chosen from applicable S-N curves. Many specimens must be tested by stressing them repeatedly at one stress level until failure occurs. This gives, after a statistical interpretation for a specific probability, a failure cycle number characteristic of this stress level. Repeating the procedure at different stress levels leads to an S-N curve.

An example of a cumulative stress spectrum is given in figure 2. Figure 3 shows measured cumulative stress spectra for tooth root stress. Figure 4 shows a cumulative contact stress spectrum with an S-N curve for specific material fatigue properties.



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Figure 2 – Example for a cumulative stress spectrum

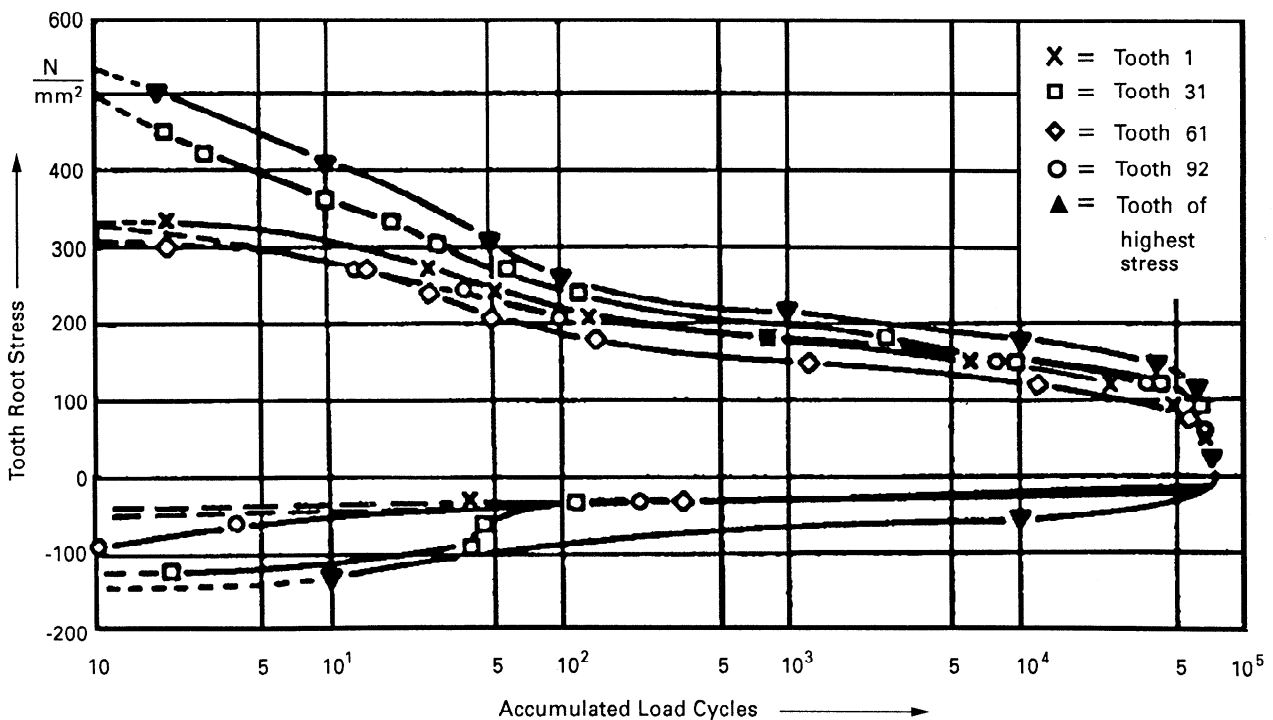
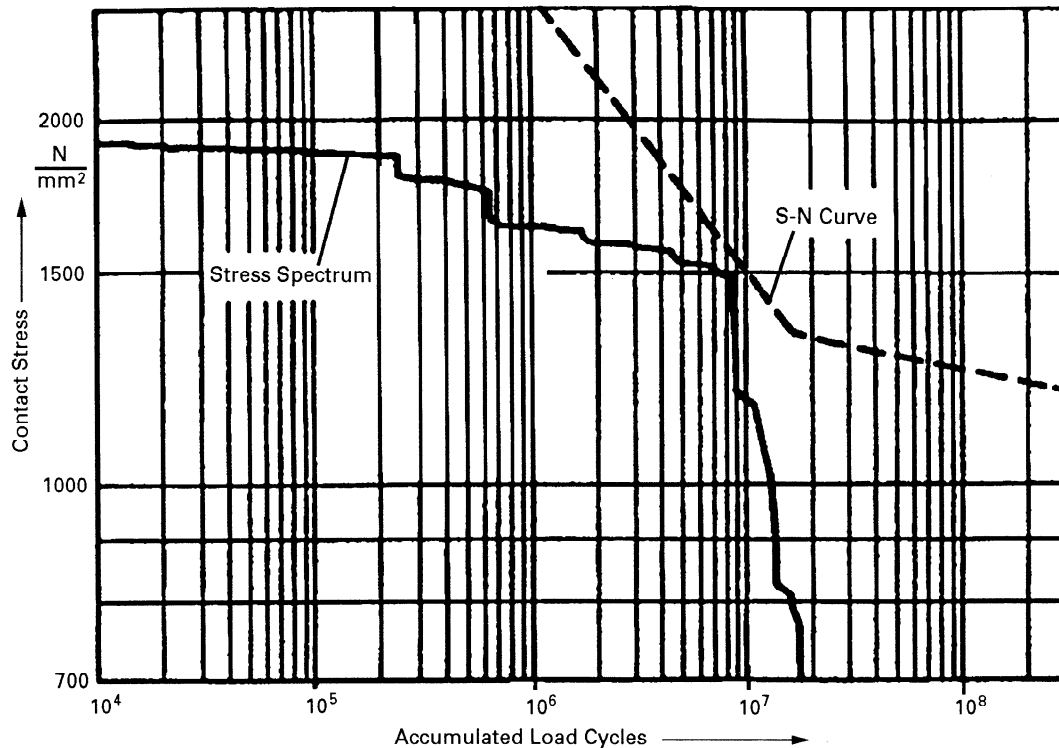


Figure 3 - Measured cumulative tooth root stress spectra for different teeth of one wheel



NOTE – The representation of the cumulative stress spectrum entirely below the S-N curve does not imply that the part will survive the total accumulative number of stress cycles. This information can be gained from a presentation as shown in figure 6.

Figure 4 – Cumulative contact stress spectrum with S-N curve

Linear, non-linear and relative methods are used.

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The literature presented in annex B gives a general account of the present state and application of damage accumulation.

4.3 Palmgren-Miner rule

The Palmgren-Miner rule – besides other rules or modifications – is a widely used linear damage accumulation method. It is assumed that the damaging effect of each stress repetition at a given stress level is equal, which means the first stress cycle at a given stress level is as damaging as the last.

The Palmgren-Miner rule operates on the hypothesis that the portion of useful fatigue life used by a number of repeated stress cycles at a particular stress is equal to the ratio of the total number of cycles during the fatigue life at a particular stress level according to the S-N curve established for the material. For example, if a part is stressed for 3000 cycles at a stress level which would cause failure in 100 000 cycles, 3% of the fatigue life would be expended. Repeated stress at another stress level would consume another similarly calculated portion of the total fatigue life.

NOTE – The used material fatigue characteristics and endurance data should be related to a specific and required failure probability, e.g. 1%, 5% or 10%.

When 100% of the fatigue life is expended in this manner, the part could be expected to fail. The order in which each of these individual stress cycles is applied is not considered significant in Palmgren-Miner analysis.

Failure could be expected when:

$$\sum_I \frac{n_I}{N_I} = 1,0 \quad \dots(1)$$

n_I Number of cycles at class interval level I

N_I Number of cycles to failure at interval level of class I (taken from the appropriate S-N curve)

If there is an endurance limit (upper, horizontal line beyond the knee in figure 5), the calculation is only done for stresses above this endurance limit.

If the appropriate S-N curve shows no endurance limit (lower line beyond the knee in figure 5), the calculation must be done for all stress levels. For each stress level, l , the number of cycles to failure, N_L , has to be taken from the corresponding part of the S-N-curve.

5 General calculation of service life, Method I

This method serves only for recalculation.

The load or stress spectrum of the gearing shall be determined by measurement, system analysis or experience.

For the calculation of cumulative damage, linear, non-linear, and relative methods may be used, provided that accuracy and reliability is appropriate to the application.

Annex B lists literature which gives data on the present state and application of calculation for service life.

6 Calculation of service strength on the basis of single-stage strength; calculation according to ISO 6336, Method II

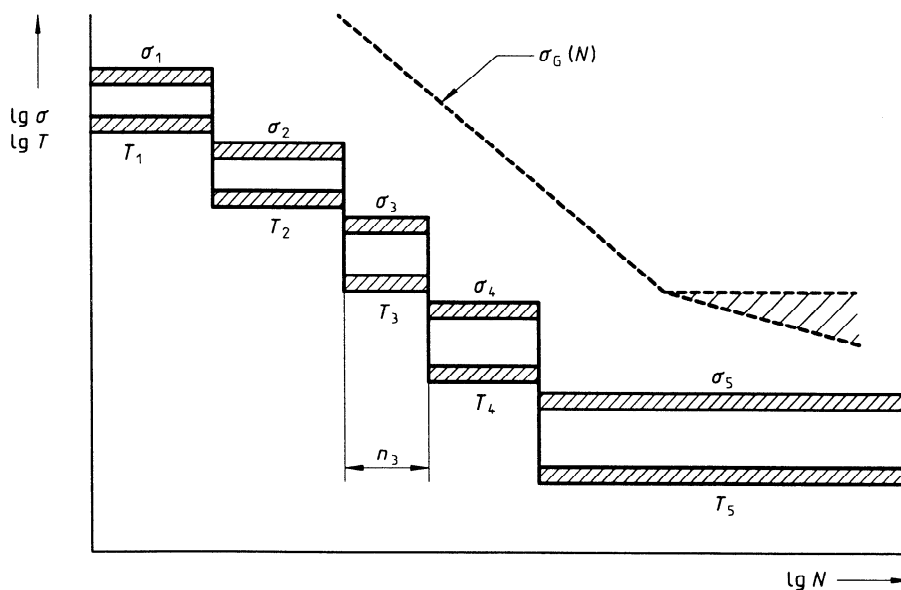
6.1 Basic principles

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This method is only valid for recalculation. It describes the application of linear cumulative damage calculations according to the Palmgren-Miner rule (see 4.3).

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This method has been chosen because it is widely known and easy to apply; the choice does not imply that the method is superior to others described in the literature.



NOTE – The representation of the cumulative stress spectrum entirely below the S-N curve does not imply that the part will survive the total accumulative number of stress cycles. This information can be gained from a presentation as shown in figure 6.

Figure 5 – Torque spectrum and associated stress spectrum with S-N curve