

Designation: C 1341 - 00

Standard Test Method for Flexural Properties of Continuous Fiber-Reinforced Advanced Ceramic Composites¹

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1. Scope

- 1.1 This test method covers the determination of flexural properties of continuous fiber-reinforced ceramic composites in the form of rectangular bars formed directly or cut from sheets, plates, or molded shapes. Three test geometries are described as follows:
- 1.1.1 *Test Geometry I*—A three-point loading system utilizing center loading on a simply supported beam.
- 1.1.2 *Test Geometry IIA*—A four-point loading system utilizing two load points equally spaced from their adjacent support points with a distance between load points of one half of the support span.
- 1.1.3 Test Geometry IIB—A four-point loading system utilizing two load points equally spaced from their adjacent support points with a distance between load points of one third of the support span.
- 1.2 This test method applies primarily to all advanced ceramic matrix composites with continuous fiber reinforcement: uni-directional (1-D), bi-directional (2-D), tri-directional (3-D), and other continuous fiber architectures. In addition, this test method may also be used with glass (amorphous) matrix composites with continuous fiber reinforcement. However, flexural strength cannot be determined for those materials that do not break or fail by tension or compression in the outer fibers. This test method does not directly address discontinuous fiber-reinforced, whisker-reinforced, or particulate-reinforced ceramics. Those types of ceramic matrix composites are better tested in flexure using Test Methods C 1161 and C 1211.
- 1.3 Tests can be performed at ambient temperatures or at elevated temperatures. At elevated temperatures, a suitable furnace is necessary for heating and holding the specimens at the desired testing temperatures.
 - 1.4 This test method includes the following:

¹ This test method is under the jurisdiction of ASTM Committee C-28 o	n
Advanced Ceramics and is the direct responsibility of Subcommittee C28.07 o	n
Ceramic Matrix Composites.	

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	Section
Scope	1
Referenced Documents	2
Terminology	3
Summary of Test Method	4
Significance and Use	5
Interferences	6
Apparatus	7
Precautionary Statement	8
Specimens	9
Procedures	10
Calculation of Results	11
Report	12
Precision and Bias	13
Keywords	14
References	
CFCC Surface Condition and Finishing	A1
Conditions and Issues in Hot Loading of	A2
Specimens into Furnaces	
Toe Compensation on Stress-Strain	A3
Curves	
Corrections for Thermal Expansion in	A4
Flexural Equations	
Example of Test Report	X1

- 1.5 The values stated in SI units are to be regarded as the standard per Practice E 380.
- 1.6 This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.

2. Referenced Documents

- 2.1 ASTM Standards:
- C 1145 Terminology of Advanced Ceramics²
- C 1161 Test Method for Flexural Strength of Advanced Ceramics at Ambient Temperature²
- C 1211 Test Method for Flexural Strength of Advanced Ceramics at Elevated Temperatures²
- C 1239 Practice for Reporting Uniaxial Data and Estimating Weibull Distribution Parameters for Advanced Ceramics²

² Annual Book of ASTM Standards, Vol 15.01.

- C 1292 Test Method for Shear Strength of Continuous Fiber-Reinforced Advanced Ceramics at Ambient Temperatures²
- D 790 Test Methods for Flexural Properties of Unreinforced and Reinforced Plastics and Electrical Insulating Materials³
- D 2344 Test Method for Apparent Interlaminar Shear Strength of Parallel Fiber Composites by Short Beam Method⁴
- D 3878 Terminology for High-Modulus Reinforcing Fibers and Their Composites⁴
- E 4 Practices for Force Verification of Testing Machines⁵
- E 6 Terminology Relating to Methods of Mechanical Testing⁵
- E 177 Practice for Use of the Terms Precision and Bias in ASTM Test Methods⁶
- E 220 Test Method for Calibration of Thermocouples by Comparison Techniques⁷
- E 337 Test Method for Measured Humidity with Psychrometer (the Measurement of Wet- and Dry-Bulb Temperatures)⁸
- E 380 Practice for Use of International System of Units (SI) (The Modernized Metric System)⁶
- E 691 Practice for Conducting an Interlaboratory Study to Determine the Precision of a Test Method⁶

3. Terminology

- 3.1 Definitions—The definitions of terms relating to flexure testing appearing in Terminology E 6 apply to the terms used in this test method. The definitions of terms relating to advanced ceramics appearing in Terminology C 1145 apply to the terms used in this test method. The definitions of terms relating to fiber-reinforced composites appearing in Terminology D 3878 apply to the terms used in this test method. Pertinent definitions as listed in Test Method C 1161, Test Method D 790, Terminology C 1145, Terminology D 3878, and Terminology E 6 are shown in the following with the appropriate source given in brackets. Additional terms used in conjunction with this test method are also defined in the following.
- 3.1.1 advanced ceramic, n—a highly engineered, high-performance, predominately nonmetallic, inorganic, ceramic material having specific functional attributes. [C 1145]
- 3.1.2 breaking load, n [F]—the load at which fracture occurs. (In this test method, fracture consists of breakage of the test bar into two or more pieces or a loss of at least 20 % of the maximum load carrying capacity.) [E 6]
- 3.1.3 *ceramic matrix composite*, *n*—a material consisting of two or more materials (insoluble in one another) in which the major, continuous component (matrix component) is a ceramic, while the secondary component(s) (reinforcing component) may be ceramic, glass-ceramic, glass, metal, or organic in nature. These components are combined on a macroscale to

form a useful engineering material possessing certain properties or behavior not possessed by the individual constituents.

- 3.1.4 continuous fiber-reinforced ceramic composite (CFCC), n—a ceramic matrix composite in which the reinforcing phase consists of a continuous fiber, continuous yarn, or a woven fabric.
- 3.1.5 *flexural strength*, $n [FL^{-2}]$ —a measure of the ultimate strength of a specified beam in bending. [C 1161]
- 3.1.6 *four-point-1/3* point flexure, *n*—a configuration of flexural strength testing where a specimen is symmetrically loaded at two locations that are situated one third of the overall span away from the outer two support bearings.
- 3.1.7 four-point-1/4 point flexure, n—a configuration of flexural strength testing where a specimen is symmetrically loaded at two locations that are situated one quarter of the overall span away from the outer two support bearings.

[C 1161]

- 3.1.8 fracture strength, n [FL^{-2}]—the calculated flexural stress at the breaking load.
- 3.1.9 modulus of elasticity, $n [FL^{-2}]$ —the ratio of stress to corresponding strain below the proportional limit. **[E 6]**
- 3.1.10 proportional limit stress, n [FL^{-2}]—the greatest stress that a material is capable of sustaining without any deviation from proportionality of stress to strain (Hooke's law).
- 3.1.10.1 Discussion—Many experiments have shown that values observed for the proportional limit vary greatly with the sensitivity and accuracy of the testing equipment, eccentricity of loading, the scale to which the stress-strain diagram is plotted, and other factors. When determination of proportional limit is required, the procedure and sensitivity of the test equipment shall be specified. [E 6]
- 3.1.11 *slow crack growth*, *n*—subcritical crack growth (extension) that may result from, but is not restricted to, such mechanisms as environmentally assisted stress corrosion or diffusive crack growth.
- 3.1.12 span-to-depth ratio, n [nd]—for a particular specimen geometry and flexure test configuration, the ratio (L/d) of the outer support span length (L) of the flexure test specimen to the thickness/depth (d) of specimen. (As used and described in Test Method D 790.)
- 3.1.13 *three-point flexure*, *n*—a configuration of flexural strength testing where a specimen is loaded at a location midway between two support bearings. [C 1161]

4. Summary of Test Method

- 4.1 A bar of rectangular cross section is tested in flexure as a beam as in one of the following three load geometries:
- 4.1.1 *Test Geometry I*—The bar rests on two supports and is loaded by means of a loading roller midway between the supports (see Fig. 1.)
- 4.1.2 *Test Geometry IIA*—The bar rests on two supports and is loaded at two points (by means of two loading rollers), each an equal distance from the adjacent support point. The inner loading points are situated one quarter of the overall span away from the outer two support bearings. The distance between the loading rollers (that is, the load span) is one half of the support span (see Fig. 1).

³ Annual Book of ASTM Standards, Vol 08.01.

⁴ Annual Book of ASTM Standards, Vol 15.03.

⁵ Annual Book of ASTM Standards, Vol 03.01.

Annual Book of ASTM Standards, Vol 14.02.
Annual Book of ASTM Standards, Vol 14.03.

⁸ Annual Book of ASTM Standards, Vol 11.03.

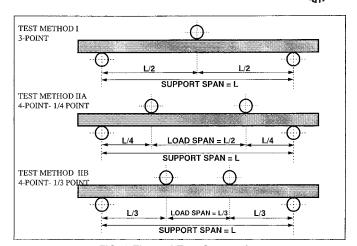


FIG. 1 Flexural Test Geometries

- 4.1.3 *Test Geometry IIB*—The bar rests on two supports and is loaded at two points (by means of two loading rollers), situated one third of the overall span away from the outer two support bearings. The distance between the loading rollers (that is, the load span) is one third of the support span (see Fig. 1).
- 4.2 The specimen is deflected until rupture occurs in the outer fibers or until there is a 20 % decrease from the peak load.
- 4.3 The flexural properties of the specimen (flexural strength and strain, fracture strength and strain, modulus of elasticity, and stress-strain curves) are calculated from the load and deflection using elastic beam equations.

5. Significance and Use

- 5.1 This test method is used for material development, quality control, and material flexural specifications. Although flexural test methods are commonly used to determine design strengths of monolithic advanced ceramics, the use of flexure test data for determining tensile or compressive properties of CFCC materials is strongly discouraged. The nonuniform stress distributions in the flexure specimen, the dissimilar mechanical behavior in tension and compression for CFCCs, low shear strengths of CFCCs, and anisotropy in fiber architecture all lead to ambiguity in using flexure results for CFCC material design data (1-4). Rather, uniaxial-loaded tensile and compressive tests are recommended for developing CFCC material design data based on a uniformly stressed test condition.
- 5.2 In this test method, the flexure stress is computed from elastic beam theory with the simplifying assumptions that the material is homogeneous and linearly elastic. This is valid for composites where the principal fiber direction is coincident/ transverse with the axis of the beam. These assumptions are necessary to calculate a flexural strength value, but limit the application to comparative type testing such as used for material development, quality control, and flexure specifications. Such comparative testing requires consistent and standardized test conditions, that is, specimen geometry/thickness, strain rates, and atmospheric/test conditions.
- 5.3 Unlike monolithic advanced ceramics which fracture catastrophically from a single dominant flaw, CFCCs generally

experience "graceful" fracture from a cumulative damage process. Therefore, the volume of material subjected to a uniform flexural stress may not be as significant a factor in determining the flexural strength of CFCCs. However, the need to test a statistically significant number of flexure specimens is not eliminated. Because of the probabilistic nature of the strength of the brittle matrices and of the ceramic fiber in CFCCs, a sufficient number of specimens at each testing condition is required for statistical analysis, with guidelines for sufficient numbers provided in 9.7. Studies to determine the exact influence of specimen volume on strength distributions for CFCCs are not currently available.

- 5.4 The four-point loading geometries (Geometries IIA and IIB) are preferred over the three-point loading geometry (Geometry I). In four-point loading, a larger portion of the test specimen is subjected to the maximum tensile and compressive stresses, as compared to the three-point geometry. If there is a statistical/Weibull character failure in the particular composite system being tested, the size of the maximum stress region will play a role in determining the mechanical properties. The four-point geometry may then produce more reliable statistical data.
- 5.5 Flexure tests provide information on the strength and deformation of materials under complex flexural stress conditions. In CFCCs nonlinear stress-strain behavior may develop as the result of cumulative damage processes (for example, matrix cracking, matrix/fiber debonding, fiber fracture, delamination, etc.) which may be influenced by testing mode, testing rate, processing effects, or environmental influences. Some of these effects may be consequences of stress corrosion or subcritical (slow) crack growth which can be minimized by testing at sufficiently rapid rates as outlined in 10.3 of this test method.
- 5.6 Because of geometry effects, the results of flexure tests of specimens fabricated to standardized test dimensions from a particular material or selected portions of a component, or both, cannot be categorically used to define the strength and deformation properties of the entire, full-size end product or its in-service behavior in different environments. The effects of size and geometry shall be carefully considered in extrapolating the test results to other configurations and performance conditions.
- 5.7 For quality control purposes, results from standardized flexure test specimens may be considered indicative of the response of the material lot from which they were taken with the given primary processing conditions and post-processing heat treatments.
- 5.8 The flexure behavior and strength of a CFCC are dependent on its inherent resistance to fracture, the presence of fracture sources, or damage accumulation processes or combination thereof. Analysis of fracture surfaces and fractography, though beyond the scope of this test method, is highly recommended.

6. Interferences

6.1 A CFCC material tested in flexure may fail in a variety of distinct fracture modes, depending on the interaction of the nonuniform stress fields in the flexure specimen and the local mechanical properties. The specimen may fail in tension,

compression, shear, or in a mix of different modes, depending on which mode reaches the critical stress level for failure to initiate. To obtain a valid flexural strength by this test method, the material must fail in the outer fiber surface in tension or compression, rather than by shear failure. The geometry of the specimen must be chosen so that shear stresses are kept low relative to the tension and compression stresses. This is done by maintaining a high ratio between the support span (L) and the thickness/depth (d) of the specimen. This L/d ratio is generally kept at values of ≥ 16 for 3-point testing and ≥ 30 for 4-point testing. If the span-to-depth ratio is too low, the specimen may fail in shear, invalidating the test. If the desired mode of failure is shear, then an appropriate shear test method should be used, such as Test Method C 1292 or D 2344.

6.2 Time-dependent phenomena, such as stress corrosion and slow crack growth, can interfere with the determination of the flexural strength at room and elevated temperatures. Creep phenomena also become significant at elevated temperatures. Both mechanisms can cause stress relaxation in flexure specimens during a strength test, thereby causing the elastic formula calculations to be in error. Test environment (vacuum, inert gas, ambient air, etc.) including moisture content (for example, relative humidity) may have an accelerating effect on stress corrosion and slow crack growth. Testing to evaluate the maximum strength potential of a material should be conducted in inert environments or at sufficiently rapid testing rates, or both, so as to minimize slow crack growth effects. Conversely, testing can be conducted in environments and testing modes and rates representative of service conditions to evaluate material performance under use conditions. When testing is conducted in uncontrolled ambient air with the intent of evaluating maximum strength potential, monitor and report the relative humidity and temperature.

6.3 Surface preparation of test specimens, although normally not considered a major concern in CFCCs, can introduce fracture sources on the surface which may have pronounced effects on flexural mechanical properties and behavior (for example, elastic and nonelastic regions of the stress-strain curve, flexural strength and strain, proportional limit stress and strain, etc.). Machining damage introduced during specimen preparation can be either a random interfering factor in the determination of flexure strength of specimen or an inherent part of the strength characteristics being measured. Surface preparation can also lead to the introduction of residual stresses. Universal or standardized test methods of surface preparation for CFCCs do not exist. It should be understood that final machining steps may or may not negate machining damage introduced during the initial machining. Thus, specimen fabrication history may play an important role in the measured strength distributions and should be reported. In addition, the nature of fabrication used for certain composites (for example, chemical vapor infiltration, hot pressing, and preceramic polymer lamination) may require the testing of specimens in the as-processed condition (that is, it may not be possible or appropriate to machine the specimen faces).

6.4 Fractures that initiate outside the uniformly stressed region of a flexure specimen (between the inner loading points in four-point and under the center load in three-point) may be

due to factors such as stress concentrations or strength limiting features in the microstructure of the specimen. Fractures which do occur outside the uniformly stressed sections will normally constitute invalid tests. If the flexure data is used in the context of estimating Weibull parameters then appropriate computational methods shall be used for such censored data. These methods are outlined in Practice C 1239.

6.5 Flexural strength at elevated temperature may be strongly dependent on loading rate as consequence of creep, stress corrosion, or slow crack growth effects. This test method measures the flexural strength at high loading rates in order to minimize these effects.

7. Apparatus

7.1 Testing Machine—Test the flexure specimens in a properly calibrated testing machine that can be operated at constant rates of cross-head motion over the range required. The error in the load measuring system shall not exceed ±1% of the maximum load being measured. The load-indicating mechanism shall be essentially free from inertial lag at the cross-head rate used. Although not recommended, if the cross-head displacement is used to determine the specimen deflection for the three-point loading geometry, determine the compliance of the load train (see Appendix X1), so that appropriate corrections can be made to the deflection measurement. Equip the system with a means for retaining the readout of the maximum load as well as a record of load versus time. Verify the accuracy of the testing machine in accordance with Practice E 4.

7.2 Loading Fixtures—The outer loading span and the desired test geometry determine the dimensions and geometry of the loading fixture. Select the fixture geometry from one of three configurations: 3-point, 4-point-1/4 point, and 4-point-1/3 point. The thickness of specimen to be tested determines the critical outer span dimension (L) of the loading fixture. The overall dimensions of the specimen and the required loading span are selected based on the specimen thickness, the desired test geometry, and the required span-to-depth ratio. Table 1, Table 2, and Table 3 give the recommended loading spans for different span/depth ratios, test specimen thicknesses, and the three test geometries. Loading fixtures shall be wide enough to support the entire width of the selected specimen geometry.

7.2.1 Ensure that the design and construction of the fixtures produces even and uniform loads along the bearing-to-specimen surfaces. A rigid loading fixture is permitted, if it is designed and aligned so that loads are evenly applied to the test specimen, particularly for four-point loading geometries. It is preferred, however, that load fixtures with an articulating geometry be used. An articulated loading fixture reduces or eliminates uneven loading caused by geometry variations of the specimen or misalignment of the test fixtures.

7.2.2 Semi-Articulating Fixtures—Specimens prepared in accordance with and meeting the parallelism requirement of 9.4 may be tested in a semi-articulating fixture. The bearing cylinders shall be parallel to each other within 0.1 mm over their length. (A representative design for a four-point fixture is illustrated in Fig. 2.).

7.2.3 Fully Articulating Fixture—Specimens with slight warp, twist, or bowing may not meet the parallelism requirements of 9.4. It is recommended that such specimens be tested

TABLE 1 Recommended Dimensions for Test Specimens of 9.1 for Various Support Span-to-Depth Ratios—Test Geometry I (3-Point)

TABLE 2 Recommended Dimensions for Test Specimens of 9.1 for Various Support Span-to-Depth Ratios—Test Geometry II-A (4 Point-1/4 Point)

(3-Point)				(4 Point-74 Point)						
Nominal Specimen Depth/ Thickness (mm)	Specimen Width (mm)	Specimen Length (mm)	Support Span (mm)	Rate of Cross-Head ^A Motion (mm/s)	Nominal Specimen Depth/ Thickness	Specimen Width (mm)	Specimen Length (mm)	Support Span (mm)	Load Span (mm)	Rate of Cross-Head ² Motion (mm/s)
		L/d = 16 to 1			(mm)					
1	3	26	16	0.04			L/d =	16 to 1		
2	6	45	32	0.09	1	3	26	16	8	0.04
3	9	60	48	0.13	2	6	45	32	16	0.09
4	12	75	64	0.17	3	9	60	48	24	0.13
5	15	90	80	0.21	4	12	75	64	32	0.17
6	18	105	96	0.26	5	15	90	80	40	0.21
10	30	180	160	0.43	6	18	105	96	48	0.26
15	45	270	240	0.64	10	30	180	160	80	0.43
20	60	360	320	0.86	15	45	270	240	120	0.64
		L/d = 32 to 1			20	60	360	320	160	0.86
1	3	42	32	0.17			L/d = 3	32 to 1		
2	6	75	64	0.34	1	3	42	32	16	0.17
3	9	105	96	0.51	2	6	75	64	32	0.34
4	12	145	128	0.68	3	9	105	96	48	0.51
5	15	180	160	0.86	4	12	145	128	64	0.68
6	18	210	192	1.03	5	15	180	160	80	0.86
10	30	360	320	1.71	6	18	210	192	96	1.03
15	45	530	480	2.57	10	30	360	320	160	1.71
20	60	710	640	3.42	15	45	530	480	240	2.57
		L/d = 40 to 1			20	60	710	640	320	3.42
1	3	50	40	0.27			L/d = -	40 to 1		
2	6	90	80	0.53		3	50	40	20	0.27
3	9	135	120	0.80	2	6	90	80	40	0.53
4	12	180	160	1.07	3	9	135	120	60	0.80
5	15	220	200	/ / 1.34	4	12	180	160	80	1.07
6	18	265	240	1.60	2 5	15	220	200	100	1.34
10	30	440	400	2.67	6	18	265	240	120	1.60
15	45	660	600	4.01	10	30	440	400	200	2.67
20	60	880	800	5.34	15	45	660	600	300	4.01
		L/d = 60 to 1			20	60	880	800	400	5.34
1	3	70	60	0.60			L/d =	60 to 1		
2	6	135	120	1.20	11341-100	3	70	60	30	0.60
3	9	200	180	1.80	2	6	135	120	60	1.20
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5	15	330	300	3.01	4	12	265	240	120	2.40
6	18	400	360	3.61	5	15	330	300	150	3.01
10	30	660	600	6.01	6	18	400	360	180	3.61
15	45	1000	900	9.02	10	30	660	600	300	6.01
20	60	1350	1200	12.02	15	45	1000	900	450	9.02
A Datas indicated	ore for a etre	in rate of 0.00	1 mm/mm o		20	60	1350	1200	600	12.02

in a fully articulating fixture. (A representative design for a four-point fixture is illustrated in Fig. 3.)

7.2.4 The test fixture shall be made of a material that is suitably rigid and resistant to permanent deformation at the loads and temperatures of testing. The test fixture material shall be essentially inert at the desired test temperatures.

7.3 Load Bearings—In both the three-point and four-point flexure test fixtures, use cylindrical bearings for support of the test specimen and for load application. The cylinders shall be made of a tool steel or a ceramic with an elastic modulus between 200 and 400 GPa and a flexural strength no less than 275 MPa. The load bearing cylinders shall remain elastic over the load and temperature ranges used.

7.3.1 Ensure that the load bearings have cylindrical surfaces that are smooth and parallel along their length to an accuracy of ± 0.05 mm. In order to avoid excessive indentation or crushing failure directly under the loading surface, the bearingsurface diameter shall be at least 3.0 mm. The bearing-surface diameter shall be approximately 1.5 times the beam depth of the test specimen size used. If the specimen has low throughthickness compressive strength, the cylinder diameter shall be four times the beam thickness to prevent crushing at the load points.

Note 1—In such circumstances, however, there is a possible error due to contact-point tangency shift due to the change in loading point as the specimen deflects during loading. The magnitude of this error can be estimated from Ref. 5.

7.3.2 Position the outer support bearing cylinders carefully such that the support span distance is accurate to a tolerance of 1 %. The load application bearing for the three-point configuration shall be positioned midway between the support bearings to an accuracy of 1 % of the outer span length. The load application (inner) bearings for the four-point configurations

A Rates indicated are for a strain rate of 0.001 mm/mm·s.

TABLE 3 Recommended Dimensions for Test Specimens of 9.1 for Various Support Span-to-Depth Ratios—Test Geometry II-B (4 Point-1/3 Point)

		(4 Point-	73 Point)							
Nominal Specimen Depth/ Thickness (mm)	Specimen Width (mm)	Specimen Length (mm)	Support Load Span Span (mm) (mm)		Rate of Cross-Head ^A Motion (mm/s)					
<i>L/d</i> = 16 to 1										
1	3	26	16	5.3	0.05					
2	6	45 32 10.			0.09					
3	9	60	48	16.0	0.14					
4	12	75	64	21.3	0.19					
5	15	90	80	26.7	0.24					
6	18	105	96	32.0	0.28					
10	30	180	160	53.3	0.47					
15	45	270	240	80.0	0.71					
20	60	360	320	106.7	0.95					
		L/d = 3	32 to 1							
1	3	42	32	10.7	0.19					
2	6	75	64	21.3	0.38					
3	9	105	96	32.0	0.57					
4	12	145	128	42.7	0.76					
5	15	180	160	53.3	0.95					
6	18	210	192	64.0	1.14					
10	30	360	320	106.7	1.89					
15	45	530	480	160.0	2.84					
20	60	710	640	213.3	3.79					
		L/d = 4	0 to 1							
1	3	50	40	13.3	0.30					
2	6	90	80	26.7	0.59					
3	9	135	120	40.0	0.89					
4	12	180	160	53.3	/ / 1.18					
5	15	220	200	66.7	1.48					
6	18	265	240	80.0	1.78					
10	30	440	400	133.3	2.96					
15	45	660	600	200.0	4.44					
20	60	880	800	266.7	5.92					
		L/d = 6	60 to 1							
1	3	70	60	20.0	0.67 TM					
2	6	135	120		1.33					
3ttps	://sta9dar	ds. 200 ai	cat 180	40.0 60.0	ds/si2.003a06c					
4	12	265	240	80.0	2.66					
4 5	15	330	300	100.0	3.33					
6	18	400	360	120.0	4.00					
10	30	400 660	600	200.0	4.00 6.66					
15	30 45	1000			9.99					
20	60	1350	1200	400.0	13.32					
25	75	1650	1500	500.0	16.65					
20	10	1000	1000	550.0	10.00					

A Rates indicated are for a strain rate of 0.001 mm/mm·s.

shall be properly positioned with respect to the support (outer) bearings to an accuracy of 1 % of the outer span length.

7.3.3 For articulating fixtures, the bearing cylinders shall be free to rotate in order to relieve frictional constraints (with the exception of the center-load bearing cylinder in three-point flexure, which need not rotate). This can be accomplished as shown in Fig. 2 and Fig. 3. Note that the outer support bearings roll outward, and the inner loading bearings roll inward.

Note 2—In general, fixed-pin fixtures have frictional constraints that have been shown to cause a systematic error on the order of 5 to 15 % in flexural strength for monolithic ceramics. Since this error is systematic, it will lead to a bias in estimates of mean strength. Rolling-pin fixtures are required for articulating fixtures by this test method. It is recognized that they may not be feasible for rigid fixtures, in which case fixed-pin fixtures may be used. But this shall be stated explicitly in the report.

- 7.4 Deflection Measurement—The test system shall have a means of measuring specimen deflection, appropriate for the load geometry and the test temperature. The preferred device measures actual deflection at the centerline of the test specimen load span, using direct contact or optical function. The calibrated range of the deflectometer shall be such that the linear strain region of the material tested will represent a minimum of 20 % of the calibrated range. The deflectometer shall have an accuracy of 1 % of the maximum deflection measured.
- 7.5 Strain Measurement—The use of strain gages for ambient testing is acceptable provided that the test material surface is smooth with little open porosity and that the applied strain gage is large enough to cover a representative area of the composite specimen. Follow the manufacturer's recommendations regarding application and performance. Strain gages shall not interfere with the deflection measuring device.
- 7.6 *Heating Apparatus*—For elevated-temperature testing, any furnace that meets the temperature uniformity and control requirements described below shall be acceptable. A furnace whose heated cavity is large enough to accept the entire test fixture is preferred.
- 7.6.1 The furnace shall be capable of establishing and maintaining a constant temperature (within $\pm 5^{\circ}$ C) during each test period. Measure the temperature uniformity of the test specimen across the load span section extending from the center to 5 mm inside the outer support points. The temperature uniformity along the load span shall be within $\pm 5^{\circ}$ C test temperatures up to and including 500°C and ± 1 % for test temperatures above 500°C.
- 7.6.1.1 In order to determine conformance to the temperature control and uniformity requirements, determine a temperature profile using thermocouples to measure the specimen temperature at three locations—the specimen center point and two points 5 mm inside the outer support points.
- 7.6.1.2 Determine temperature uniformity for all elevated-temperature testing and recheck the uniformity if any of the following parameters are changed: heating method, specimen material, sample geometry, or test temperature, or combination thereof.
- 7.6.2 Temperature Measurement—The use of thermocouples (TC) is recommended and preferred; however, the use of optical pyrometery is acceptable. For TC measurement, elevated-temperature tests require the placement of one TC at the specimen center. The sheathed TC should be within 1 mm of the test specimen. The use of two additional thermocouples at locations 5 mm inside the outer support points is recommended to check for temperature uniformity. Thermocouples shall be calibrated in accordance with Test Method E 220 with a verified accuracy of $\pm 5^{\circ}$ C.
- 7.6.3 Atmosphere Control—The furnace may have an air, inert, or vacuum environment, as required. If an inert or vacuum environment is used, and it is necessary to apply load through a bellows, fitting, or seal, verify that load losses or errors do not exceed 1 % of the expected failure loads.
- 7.7 Data Acquisition—At the minimum, obtain an autographic record of the applied load and center-point deflection or sample strain versus time for the specified cross-head rate. Either analog chart recorders or digital data acquisition systems