



Designation: B211 – 03

Standard Specification for Aluminum and Aluminum-Alloy Bar, Rod, and Wire¹

This standard is issued under the fixed designation B211; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense.

1. Scope*

1.1 This specification² covers rolled or cold-finished bar, rod, and wire in alloys (Note 1) and tempers as shown in Table 2.

NOTE 1—Throughout this specification use of the term *alloy* in the general sense includes aluminum as well as aluminum alloy.

NOTE 2—The term *cold finished* is used to indicate the type of surface finish, sharpness of angles, and dimensional tolerances produced by drawing through a die.

NOTE 3—See Specification B221 for aluminum and aluminum-alloy extruded bars, rods, wire, shapes, and tubes; and Specification B316/B316M for aluminum and aluminum-alloy rivet and cold-heading wire and rods.

1.2 Alloy and temper designations are in accordance with ANSI H35.1. The equivalent UNS alloy designations are those of Table 1 preceded by A9, for example, A91100 for aluminum 1100 in accordance with Practice E527.

1.3 A complete metric companion to Specification B211 has been developed—B211M; therefore, no metric equivalents are presented in this specification.

1.4 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.

2. Referenced Documents

2.1 The following documents of the issue in effect on date of material purchase form a part of this specification to the extent referenced herein:

2.2 *ASTM Standards*:³

B221 Specification for Aluminum and Aluminum-Alloy Extruded Bars, Rods, Wire, Profiles, and Tubes

B316/B316M Specification for Aluminum and Aluminum-Alloy Rivet and Cold-Heading Wire and Rods

B557 Test Methods for Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products

B594 Practice for Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Aerospace Applications

B660 Practices for Packaging/Packing of Aluminum and Magnesium Products

B666/B666M Practice for Identification Marking of Aluminum and Magnesium Products

B881 Terminology Relating to Aluminum- and Magnesium-Alloy Products

B918 Practice for Heat Treatment of Wrought Aluminum Alloys

E29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications

E34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Alloys

E55 Practice for Sampling Wrought Nonferrous Metals and Alloys for Determination of Chemical Composition

E290 Test Methods for Bend Testing of Material for Ductility

E527 Practice for Numbering Metals and Alloys in the Unified Numbering System (UNS)

E607 Test Method for Atomic Emission Spectrometric Analysis Aluminum Alloys by the Point to Plane Technique Nitrogen Atmosphere

E716 Practices for Sampling Aluminum and Aluminum Alloys for Spectrochemical Analysis

E1004 Test Method for Determining Electrical Conductivity Using the Electromagnetic (Eddy-Current) Method

E1251 Test Method for Analysis of Aluminum and Aluminum Alloys by Atomic Emission Spectrometry

G47 Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

2.3 *ANSI Standards*:

H35.1 Alloy and Temper Designation Systems for Aluminum

H35.2 Dimensional Tolerances for Aluminum Mill Products

2.4 *Federal Standard*:

Fed. Std. No. 123 Marking for Shipment (Civil Agencies)⁴

¹ This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum Alloy Wrought Products.

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² For ASME Boiler and Pressure Vessel Code applications see related Specification SB-211 in Section II of that Code.

³ For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For Annual Book of ASTM Standards volume information, refer to the standard's Document Summary page on the ASTM website.

⁴ Available from Standardization Documents Order Desk, Bldg. 4 Section D, 700 Robbins Ave., Philadelphia, PA 19111-5094, Attn: NPODS.

*A Summary of Changes section appears at the end of this standard.

2.5 Military Standard:
MIL-STD-129 Marking for Shipment and Storage⁴

4.1.6 Geometry and Dimensions, Diameter for rounds; distance across flats for square-cornered squares, hexagons, or

TABLE 1 Chemical Composition Limits^{A,B,C}

Alloy	Silicon	Iron	Copper	Manganese	Magnesium	Chromium	Zinc	Bismuth	Lead	Titanium	Other Elements ^D		Aluminum
											Each	Total ^E	
1060	0.25	0.35	0.05	0.03	0.03	...	0.05	0.03	0.03 ^F	...	99.60 min ^G
1100	0.95 Si + Fe	...	0.05–0.20	0.05	0.10	0.05	0.15	99.00 min ^G
2011	0.40	0.7	5.0–6.0	0.30	0.20–0.6	0.20–0.6	...	0.05	0.15	remainder
2014	0.50–1.2	0.7	3.9–5.0	0.40–1.2	0.20–0.8	0.10	0.25	0.15	0.05	0.15	remainder
2017	0.20–0.8	0.7	3.5–4.5	0.40–1.0	0.40–0.8	0.10	0.25	0.15	0.05	0.15	remainder
2024	0.50	0.50	3.8–4.9	0.30–0.9	1.2–1.8	0.10	0.25	0.15	0.05	0.15	remainder
2219	0.20	0.30	5.8–6.8	0.20–0.40	0.02	...	0.10	0.02–0.10	0.05 ^H	0.15 ^H	remainder
3003	0.6	0.7	0.05–0.20	1.0–1.5	0.10	0.05	0.15	remainder
5052	0.25	0.40	0.10	0.10	2.2–2.8	0.15–0.35	0.10	0.05	0.15	remainder
5056	0.30	0.40	0.10	0.05–0.20	4.5–5.6	0.05–0.20	0.10	0.05	0.15	remainder
5154	0.25	0.40	0.10	0.10	3.1–3.9	0.15–0.35	0.20	0.20	0.05	0.15	remainder
6061	0.40–0.8	0.7	0.15–0.40	0.15	0.8–1.2	0.04–0.35	0.25	0.15	0.05	0.15	remainder
6110	0.7–1.5	0.8	0.20–0.7	0.20–0.7	0.50–1.1	0.04–0.25	0.30	0.15	0.05	0.15	remainder
6262	0.40–0.8	0.7	0.15–0.40	0.15	0.8–1.2	0.04–0.14	0.25	0.40–0.7	0.40–0.7	0.15	0.05	0.15	remainder
7075	0.40	0.50	1.2–2.0	0.30	2.1–2.9	0.18–0.28	5.1–6.1	0.20	0.05	0.15	remainder

^A Limits are in mass percent maximum unless otherwise shown.

^B Analysis shall be made for the elements for which limits are shown in this table.

^C For purposes of determining conformance to these limits, an observed value or a calculated value obtained from analysis shall be rounded to the nearest unit in the last right-hand place of figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E29.

^D *Others* includes listed elements for which no specific limit is shown as well as unlisted metallic elements. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic *Others* elements. Should any analysis by the producer or the purchaser establish that an *Others* element exceeds the limit of *Each* or that the aggregate of several *Others* elements exceeds the limit of *Total*, the material shall be considered non-conforming.

^E *Other elements*—Total shall be the sum of unspecified metallic elements 0.010 % or more each, rounded to the second decimal before determining the sum.

^F Vanadium 0.05 % max.

^G The aluminum content is the difference between 100.00 % and the sum of all other metallic elements and silicon present in amounts of 0.010 % or more each, rounded to the second decimal before determining the sum.

^H Vanadium 0.05–0.15 % zirconium 0.10–0.25 %. The total for other elements does not include vanadium and zirconium.

2.6 Aerospace Material Specification:

AMS 2772 Heat Treatment of Aluminum Alloy Raw Materials⁵

octagons; width and depth for square-cornered rectangles (orders for squares, hexagons, octagons, or rectangles with rounded corners usually require a drawing),

3. Terminology

3.1 *Definitions*: Refer to Terminology B881 for definitions of product terms in this specification.

3.1.1 *flatten and slit wire*—Flatten wire which has been slit to obtain square edges.

3.2 *Definitions of Terms Specific to This Standard*:

3.2.1 *capable of*—The term *capable of* as used in this specification means that the test need not be performed by the producer of the material. However, should subsequent testing by the purchaser establish that the material does not meet these requirements, the material shall be subject to rejection.

4. Ordering Information

4.1 Orders for material to this specification shall include the following information:

4.1.1 This specification designation (which includes the number, the year, and the revision letter, if applicable),

4.1.2 Quantity in pieces or pounds,

4.1.3 Alloy (Section 7),

4.1.4 Temper (Section 9),

4.1.5 *Product Form*, Rolled or cold finished bar, rolled or cold finished rod, or wire,

4.1.7 Length, and

4.1.8 Tensile property limits and dimensional tolerances for sized not covered in Table 2 and in ANSI H35.2, respectively.

4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:

4.2.1 Whether heat treatment in accordance with Practice B918 is required (8.2),

4.2.2 Whether 7075-O material is required to develop requirements for T73 temper (see 10.1.2),

4.2.3 Whether bend testing is required for 2017, 2024, or 3003 (Section 12),

4.2.4 When specified finish of bar and rod is not required (Section 15),

4.2.5 Whether marking for identification is required (Section 16),

4.2.6 Whether ultrasonic inspection is required (Section 17, Table 3),

4.2.7 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (Section 19),

4.2.8 Whether certification is required (Section 21), and

4.2.9 Whether Practices B660 apply, and if so, the levels of preservation, packaging, and packing required (Section 22).

⁵ Available from Society of Automotive Engineers (SAE), 400 Commonwealth Dr., Warrendale, PA 15096-0001.

5. Manufacture

5.1 The products covered by this specification shall be produced either by hot extruding and cold finishing or by hot rolling with or without cold finishing, at the option of the producer.

TABLE 2 Mechanical Property Limits^A

Temper	Specified Diameter or Thickness, in.	Tensile Strength, ksi		Yield Strength ^B (0.2 % offset), min, ksi	Elonga- tion ^B in 2 in. or 4 × Diam- eter, min, %
		min	max		
Aluminum 1060					
O	0.124 and under	8.0
	0.125 and over	8.0	...	2.5	25
H14	0.374 and under	12.0	...	10.0	...
H18	0.374 and under	16.0	...	13.0	...
Aluminum 1100					
O	0.124 and under	11.0	15.5
	0.125 and over	11.0	15.5	3.0	25
H12	0.374 and under	14.0
H14	0.374 and under	16.0
H16	0.374 and under	19.0
H18	0.374 and under	22.0
H112	all	11.0	...	3.0	...
F	all	^c	...	^c	...
Alloy 2011					
T3	0.125–1.500	45.0	...	38.0	10
	1.501–2.000	43.0	...	34.0	12
	2.001–3.500	42.0	...	30.0	12
T4 and T451 ^D	0.125–8.000	40.0	...	18.0	16
T6 and T651	0.375–6.500	54.0	...	40.0	10
T8	0.125–3.250	54.0	...	40.0	10
Alloy 2014 ^E					
O	0.124 and under	...	35.0
	0.125–8.000	...	35.0	...	12
T4, T42 ^F , and T451 ^D	0.124 and under	55.0
	0.125–8.000 ^G	55.0	...	32.0	16
T6, T62 ^F , and T651 ^D	0.124 and under	65.0
	0.125–8.000 ^G	65.0	...	55.0	8
Alloy 2017 ^E					
O	0.124 and under	...	35.0
	0.125–8.000	...	35.0	...	16
T4, T42 ^F , and T451 ^D	0.124 and under	55.0
	0.125–8.000 ^H	55.0	...	32.0	12
Alloy 2024 ^E					
O	0.124 and under	...	35.0
	0.125–8.000	...	35.0	...	16
T36	0.124 and under	69.0
	0.125–0.375	69.0	...	52.0	10
T4 ^I	0.124 and under	62.0
	0.125–0.499	62.0	...	45.0 ^J	10
	0.500–4.500 ^G	62.0	...	42.0 ^J	10
	4.501–6.500 ^J	62.0	...	40.0	10
	6.501–8.000 ^J	58.0	...	38.0	10
T42 ^F	0.124 and under	62.0
	0.125–1.000	62.0	...	40.0	10
	1.001–6.500 ^G	62.0	...	40.0	10
T351 ^D	0.500–6.500 ^G	62.0	...	45.0	10
	6.501–8.000	62.0	...	45.0	9
T6	0.124 and under	62.0
	0.125–6.500 ^G	62.0	...	50.0	5
T62 ^F	0.124 and under	60.0
	0.125–6.500 ^G	60.0	...	46.0	5
T851 ^D	0.500–6.500 ^G	66.0	...	58.0	5
Alloy 2219					
T851 ^D	0.500–2.000	58.0	...	40.0	4
	2.001–4.000	57.0	...	39.0	4

TABLE 2 *Continued*

Temper	Specified Diameter or Thickness, in.	Tensile Strength, ksi		Yield Strength ^B (0.2 % offset), min, ksi	Elonga- tion ^B in 2 in. or 4 × Diam- eter, min, %
		min	max		
Alloy 3003					
O	all	14.0	19.0	5.0	25
H12	0.374 and under	17.0
H14	0.374 and under	20.0
H16	0.374 and under	24.0
H18	0.374 and under	27.0
H112	all	14.0	...	5.0	...
F	all	^c	...	^c	...
Alloy 5052					
O	0.124 and under	...	32.0
	0.125 and over	25.0	32.0	9.5	25
H32	0.124 and under	31.0
	0.125–0.374	31.0	...	23.0	...
H34	0.374 and under	34.0	...	26.0	...
H36	0.124 and under	37.0
	0.125–0.374	37.0	...	29.0	...
H38	0.374 and under	39.0
F	all	^c	...	^c	...
Alloy 5056					
O	0.124 and under	...	46.0
	0.125 and over	...	46.0	...	20
H111	0.374 and under	44.0
H12	0.374 and under	46.0
H32	0.374 and under	44.0
H14	0.374 and under	52.0
H34	0.374 and under	50.0
H18	0.374 and under	58.0
H38	0.374 and under	55.0
H192	0.374 and under	60.0
H392	0.374 and under	58.0
Alloy 5154					
O	all	30.0	41.0	11.0	25
H32	0.374 and under	36.0
H34	0.374 and under	39.0
H36	0.374 and under	42.0
H38	0.374 and under	45.0
H112	all	30.0	...	11.0	...
Alloy 6061 ^E					
O	0.124 and under	...	22.0
	0.125–8.000	...	22.0	...	18
T4 and T451 ^D	0.124 and under	30.0
	0.125–8.000 ^H	30.0	...	16.0	18
T42 ^F	0.125–8.000 ^H	30.0	...	14.0	18
T6, T62 ^F , and T651 ^D	0.124 and under	42.0
	0.125–8.000 ^H	42.0	...	35.0	10
T89 and T94	0.374 and under	54.0	...	47.0	...
Alloy 6110					
T9	0.374 and under	65.0	...	63.0	2
Alloy 6262					
T6 and T651 ^D	0.125–8.000 ^G	42.0	...	35.0	10
T9	0.125–2.000	52.0	...	48.0	5
	2.001–3.000	50.0	...	46.0	5
Alloy 7075 ^F					
O	0.124 and under	...	40.0
	0.125–8.000	...	40.0	...	10
T6, T62	0.124 and under	77.0	...	66.0	...
	0.125–4.000 ^K	77.0	...	66.0	7
T651	0.124 and under	77.0	...	66.0	...
	0.125–4.000 ^K	77.0	...	66.0	7
	4.001–6.000	75.0	...	64.0	7
	6.001–7.000	73.0	...	62.0	7
T73 and T7351 ^D	0.124 and under	68.0
	0.125–4.000	68.0	...	56.0	10
	4.001–5.000	66.0	...	55.0	8

TABLE 2 *Continued*

Temper	Specified Diameter or Thickness, in.	Tensile Strength, ksi		Yield Strength ^B (0.2 % offset), min, ksi	Elonga- tion ^B in 2 in. or 4 × Diam- eter, min, %
		min	max		
	5.001–6.000	64.0		52.0	8
Temper	Specified Diameter or Thickness, in.			Bend Diameter Factor, N	
Alloy 2017					
T4, T42, and T451		0.124 and under 0.125–8.000 ^H		3 ^L 6 ^L	
Alloy 2024					
O		0.124 and under		1	
T351, T4, T42		0.124 and under		3	
		0.125–6.500 Alloy 3003		6	
all					
O		all		0	
H12		0.374 and under		2	
H14		0.374 and under		2	
H16		0.374 and under		8	

^A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest 0.1 ksi and each value for elongation to the nearest 0.5 %, both in accordance with the rounding-off method of Practice E29. The basis for establishment of tensile property limits is shown in Annex A1.

^B The measurement of yield strength and elongation is not required for wire less than 0.125 in. in thickness or diameter.

^C There are no tensile requirements for material in the F temper but it usually can be expected that material 1½ in. or less in thickness or diameter (except sections over 4 in. in width) will have a strength about equivalent to the H14 or H34 temper. As size increases the strength decreases to nearly that of the O temper.

^D For stress-relieved tempers, characteristics and properties other than those specified may differ somewhat from the corresponding characteristics and properties of material in the basic tempers.

^E Also available in the F temper for which no properties are specified and no tension tests are performed but for which tests are performed for confirmation of heat-treat response as required by Section 10.

^F Material in the T42 or T62 tempers is not available from the materials producers. These properties can usually be obtained by the user when material is properly solution heat treated or solution and precipitation heat treated from the O or F temper. These properties also apply to samples of material in the O or F temper that are solution heat treated or solution and precipitation heat treated by the producer to determine that the material will respond to proper heat treatment. Properties attained by the user, however, may be lower than those listed if the material has been formed or otherwise cold or hot worked, particularly in the O temper, prior to solution heat treatment.

^G Properties listed for this full size increment are applicable to rod. Properties listed are also applicable to square, rectangular, hexagonal, or octagonal bar having a maximum thickness of 4 in. and a maximum cross-sectional area of 36 in.².

^H For bar, maximum cross-sectional area is 50 in.².

^I Minimum yield strength for 2024-T4 wire and rod 0.125 in. and larger in thickness or diameter, produced in coil form for both straight length and coiled products, is 40.0 ksi.

^J Properties listed for this size increment are applicable to rod only.

^K For rounds, maximum diameter is 4 in.; for square, hexagonal, or octagonal bar, maximum thickness is 3½ in.; for rectangular bar, maximum thickness is 3 in. with corresponding maximum width of 6 in.; for rectangular bar less than 3 in. in thickness, maximum width is 10 in.

^L Bend diameter factor values stated for this full size increment apply to T4 product only. Values listed also apply to T451 product in the 0.500–8.000 in. size range.

6. Quality Assurance

6.1 *Responsibility for Inspection and Tests*—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use his own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order or at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set forth in this specification where such inspections are deemed necessary to ensure that material conforms to prescribed requirements.

6.2 *Lot Definition*—An inspection lot shall be defined as follows:

6.2.1 For heat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions traceable to a heat-treat lot or lots, and subjected to inspection at one time.

6.2.2 For nonheat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill

form, alloy, temper, and nominal dimensions subjected to inspection at one time.

7. Chemical Composition

7.1 *Limits*—The bars, rods, and wire shall conform to the chemical composition limits specified in Table 1. Conformance shall be determined by the producer by analyzing samples taken at the time the ingots are cast, or samples taken from the finished or semifinished product. If the producer has determined the chemical composition of the material during the course of manufacture, sampling and analysis of the finished product shall not be required.

NOTE 4—It is standard practice in the United States aluminum industry to determine conformance to the chemical composition limits prior to further processing of ingots into wrought products. Due to the continuous nature of the process, it is not practical to keep a specific ingot analysis identified with a specific quantity of finished material.

7.2 *Number of Samples*—The number of samples taken for determination of chemical composition shall be as follows:

7.2.1 When samples are taken at the time the ingots are cast, at least one sample shall be taken for each group of ingots cast simultaneously from the same source of molten metal.

7.2.2 When samples are taken from the finished or semifinished product, a sample shall be taken to represent each 4000 lb, or fraction thereof, in the lot, except that no more than one sample shall be required per piece.

7.3 *Methods of Sampling*—Samples for determination of chemical composition shall be taken in accordance with one of the following methods:

7.3.1 Samples for chemical analysis shall be taken from the material by drilling, sawing, milling, turning, clipping, and so forth, a representative piece or pieces to obtain a weight of prepared sample not less than 75 g. Sampling shall be in accordance with Practice E55.

7.3.2 Sampling for spectrochemical analysis shall be in accordance with Practices E716. Samples for other methods of analysis shall be suitable for the form of material being analyzed and the type of analytical method used.

NOTE 5—It is difficult to obtain a reliable analysis of each of the components of clad materials using material in its finished state. A reasonably accurate determination of the core composition can be made if the cladding is substantially removed prior to analysis. The cladding composition is more difficult to determine because of the relatively thin layer and because of diffusion of core elements to the cladding. The correctness of cladding alloy used can usually be verified by a combination of metallographic examination and spectrochemical analysis of the surface at several widely separated points.

7.4 *Method of Analysis*—The determination of chemical composition shall be made in accordance with suitable chemical (Test Methods E34), or spectrochemical (Test Methods E607 and E1251), methods. Other methods may be used only when no published ASTM method is available. In case of dispute, the methods of analysis shall be agreed upon between the purchaser and the producer.

8. Heat Treatment

8.1 Unless otherwise specified in 8.2, producer or supplier heat treatment for the applicable tempers in Table 2 shall be in accordance with AMS 2772.

8.2 When specified, heat treatment of applicable tempers in Table 2 shall be in accordance with Practice B918.

9. Tensile Properties of Material As Supplied

9.1 *Limits*—The bar, rod, and wire shall conform to the tensile requirements in Table 2.

9.2 *Number of Specimens*:

9.2.1 For material having a nominal weight of less than 1 lb/linear ft, one tension test specimen shall be taken for each 1000 lb or fraction thereof in the lot. Only one specimen shall be taken from any one piece when more than one piece is available.

9.2.2 For material having a nominal weight of 1 lb or more/linear ft, one tension test specimen shall be taken for each 1000 ft or fraction thereof in the lot. Only one specimen shall be taken from any one piece when more than one piece is available.

9.3 *Test Specimens*—Geometry of test specimens and the location in the product from which they are taken shall be as specified in Test Methods B557.

9.4 *Test Methods*—The tension tests shall be made in accordance with Test Method B557.

10. Producer Confirmation of Heat-Treat Response

10.1 In addition to the requirements of 9.1, material in Alloys 2014, 2017, 2024, and 6061 produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for T42 temper material. The heat-treated samples may be tested prior to four days natural aging, but if they fail to conform to the T42 temper properties, the tests may be repeated after completion of four days natural aging without prejudice.

10.1.1 Alloy 7075 material produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and precipitation heat treatment, conform to the properties specified in Table 2 for T62 temper material.

10.1.2 When specified, 7075-O material (within the size limits specified in Table 2) shall, after proper solution and precipitation heat treatment, conform to the properties specified for T73 temper in Table 2 and Section 13.

10.2 *Number of Specimens*—The number of specimens from each lot of O temper material and F temper material to verify conformance with 10.1 shall be as specified in 9.2.

11. Heat Treatment and Reheat Treatment Capability

11.1 As-received material in the O or F temper and in Alloys 2014, 2017, 2024, and 6061 (within the size limitation specified in Table 2 and without the imposition of cold work) shall, after proper solution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for T42 temper material.

TABLE 3 Ultrasonic Discontinuity Limits for Rolled or Cold-Finished Bar^A

Alloys	Size			Discontinuity Class ^B
	Thickness, in.	Maximum Weight per Piece, lb	Maximum Width to Thickness Ratio	
2014, 2219 2024, 7075 }	0.500–1.499	600	...	B
	1.500–3.000	600	...	A
	3.001–6.000	1000	...	B

^A Discontinuities in excess of those listed in this table shall be allowed if it is established that they will be removed by machining or that they are in noncritical areas.

^B The discontinuity class limits are defined in Section 11 of Practice B594.

11.2 As-received Alloy 7075 material in the O or F temper (within the size limitation specified in Table 2 and without the imposition of cold work) shall, after proper solution and precipitation heat treatment, conform to the properties specified in Table 2 for T6 and T62 tempers.

11.3 Material in Alloys and Tempers 2014-T4, T451, T6, T651; 2017-T4, T451; 2024-T4, T6, T351, and T851, shall, after proper resolution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for the T42 temper.