



## Standard Test Method for Freezing Point of Aviation Fuels<sup>1</sup>

This standard is issued under the fixed designation D 2386; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon ( $\epsilon$ ) indicates an editorial change since the last revision or reapproval.

*This test method has been approved by the sponsoring committees and accepted by the Cooperating Societies in accordance with established procedures.*

*This standard has been approved for use by agencies of the Department of Defense.*

### 1. Scope

1.1 This test method covers the determination of the temperature below which solid hydrocarbon crystals may form in aviation turbine fuels and aviation gasoline.

NOTE 1—The interlaboratory program that generated the precisions for this test method did not include aviation gasoline.

1.2 The values stated in acceptable metric units are to be regarded as the standard.

1.3 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.* For specific hazard statements see 5.4, Section 6, and 7.2.

### 2. Referenced Documents

#### 2.1 ASTM Standards:

- D 910 Specification for Aviation Gasolines<sup>2</sup>
- D 1655 Specification for Aviation Turbine Fuels<sup>2</sup>
- D 3117 Test Method for Wax Appearance Point of Distillate Fuels<sup>2</sup>
- E 1 Specification for ASTM Thermometers<sup>3</sup>
- E 77 Test Method for Inspection and Verification of Thermometers<sup>3</sup>

#### 2.2 IP Standard:

IP Standards for Petroleum and Its Products, Part 1<sup>4</sup>

### 3. Terminology

#### 3.1 Definitions of Terms Specific to This Standard:

3.1.1 *freezing point*—the fuel temperature at which solid

hydrocarbon crystals, formed on cooling, disappear when the temperature of the fuel is allowed to rise.

### 4. Significance and Use

4.1 The freezing point of an aviation fuel is the lowest temperature at which the fuel remains free of solid hydrocarbon crystals that can restrict the flow of fuel through filters if present in the fuel system of the aircraft. The temperature of the fuel in the aircraft tank normally falls during flight depending on aircraft speed, altitude, and flight duration. The freezing point of the fuel must always be lower than the minimum operational tank temperature.

4.2 Freezing point is a requirement in Specifications D 910 and D 1655.

### 5. Apparatus

5.1 *Jacketed Sample Tube*—A double-walled, unsilvered vessel, similar to a Dewar flask, the space between the inner and outer tube walls being filled at atmospheric pressure with dry nitrogen or air. The mouth of the sample tube shall be closed with a stopper supporting the thermometer and moisture-proof collar through which the stirrer passes (Fig. 1).

5.2 *Collars*—Moisture-proof collars as shown in Fig. 2 shall be used to prevent condensation of moisture.

5.3 *Stirrer*—Shall be made of 1.6-mm brass rod bent into a smooth three-loop spiral at the bottom.

NOTE 2—The stirrer may be mechanically actuated as described in the apparatus section of Test Method D 3117.

5.4 *Vacuum Flask*—An unsilvered vacuum flask (**Warning**— Implosion hazard.) having the minimum dimensions shown in Fig. 1 shall be used to hold an adequate volume of cooling liquid and permit the necessary depth of immersion of the jacketed sample tube.

5.5 *Thermometer*—A total immersion type, having a range from  $-80$  to  $+20^{\circ}\text{C}$ , designated as ASTM No. 114C/IP No. 14C. (See Specification E 1, or Appendix A, IP Standard Thermometers, Volume 2, IP Standard Methods for Analysis and Testing of Petroleum and Related Products.)

<sup>1</sup> This test method is under the jurisdiction of ASTM Committee D02 on Petroleum Products and Lubricants and is the direct responsibility of Subcommittee D02.07 on Flow Properties.

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<sup>2</sup> *Annual Book of ASTM Standards*, Vol 05.01.

<sup>3</sup> *Annual Book of ASTM Standards*, Vol 14.03.

<sup>4</sup> Available from: Institute of Petroleum, 61 New Cavendish Street, London, W1M 8AR, U.K.

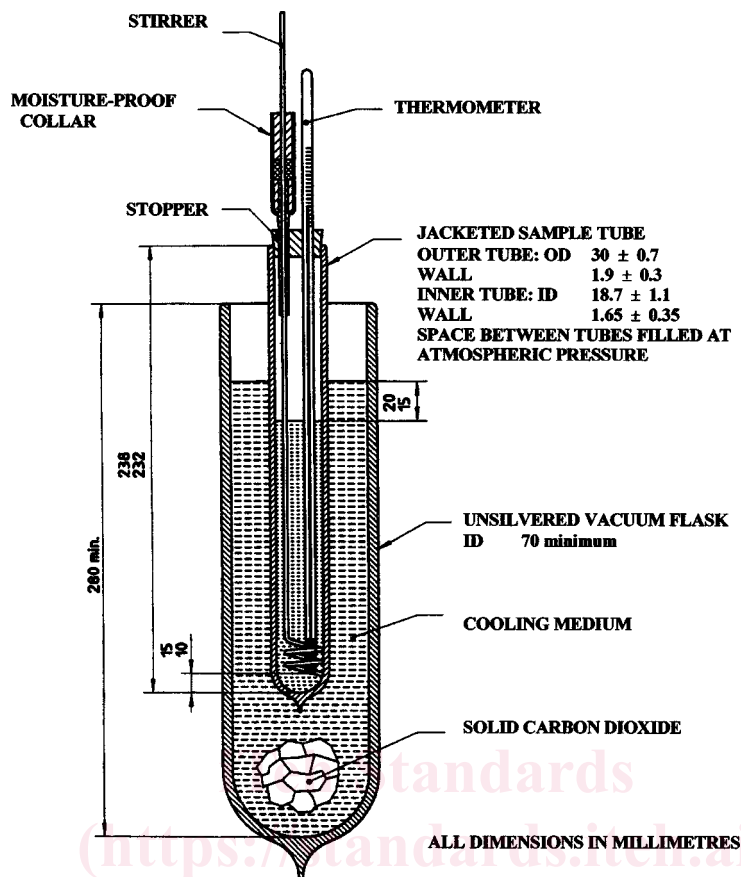


FIG. 1 Freezing Point Apparatus

NOTE 3—The accuracy of this thermometer is to be checked in accordance with Test Method E 77, at temperatures of 0, -40, -60, and -75°C.<sup>5</sup>

NOTE 4—Carbon dioxide (solid) and liquid nitrogen liberate gasses that can cause suffocation. Contact with skin causes burns, freezing, or both.

## 6. Reagents and Materials

6.1 *Acetone* (**Warning—Extremely flammable**)—Technical Grade acetone is suitable for the cooling bath, provided it does not leave a residue on drying.

6.2 *Ethanol or Ethyl Alcohol* (**Warning—Extremely flammable**)—A commercial or technical grade of dry ethanol is suitable for the cooling bath.

6.3 *Isopropyl Alcohol* (**Warning—Extremely flammable**)—A commercial or technical grade of dry isopropyl alcohol is suitable.

6.4 *Methanol or Methyl Alcohol* (**Warning—Extremely flammable and Toxic**)—A commercial or technical grade of dry methanol is suitable for the cooling bath.

6.5 *Carbon Dioxide (Solid) or Dry Ice* (**Warning—Extremely cold, -78°C**) (See Note 4)—A commercial grade of dry ice is suitable for use in the cooling bath.

6.6 *Liquid Nitrogen* (**Warning—Extremely cold, -196°C**)—A commercial or technical grade of liquid nitrogen is suitable for the cooling bath when the freezing point is lower than -65°C.

## 7. Procedure

7.1 Measure out  $25 \pm 1$  mL of the fuel and transfer it to the clean, dry, jacketed sample tube. Close the tube tightly with the cork holding the stirrer, thermometer, and moisture proof collar and adjust the thermometer position so that its bulb does not touch the walls of the tube flask and is approximately in the center. The bulb of the thermometer should be 10 to 15 mm from the bottom of the sample tube.

7.2 Clamp the jacketed sample tube so that it extends as far as possible into the vacuum flask (**Warning—Implosion hazard.**) containing the cooling medium (Note 5). The surface of the sample should be approximately 15 to 20 mm below the level of the coolant. Unless the medium is cooled by mechanical refrigeration, add solid carbon dioxide as necessary throughout the test to maintain the coolant level in the vacuum flask.

NOTE 5—Acetone and either methyl, ethyl, or isopropyl alcohols are suitable. All of these require cautious handling. Liquid nitrogen may also be used as a coolant instead of liquids cooled with solid carbon dioxide for fuel samples which have a freezing point below -65°C. Mechanical refrigeration is permitted. Where used the refrigerant temperature should be -70°C to 80°C.

7.3 Stir the fuel continuously, moving the stirrer up and down at the rate of 1 to 1.5 cycles/s, except when making

<sup>5</sup> The U.S. National Bureau of Standards, Gaithersburg, MD, and the British National Physical Laboratory, Teddington, England are able to certify thermometers at these temperatures.