UDC 629.7.046: 684.433

Ref. No.: ISO/R 837-1968 (E)

Withdrawn wi 1985
Replaced by 150 7166: 1985

ISO

INTERNATIONAL ORGANIZATION FOR STANDARDIZATION

(100)

ISO RECOMMENDATION R 837

(standards.iteh.ai)

ISO/R 837:1968
https://standards.iteh.ai/catalog/standards/sist/ac610e0b-8a66-4b14-b840-6b94ag4EDi7fon-837-1968
October 1968

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BRIEF HISTORY

The ISO Recommendation R 837, Aircraft seat rails and pins, was drawn up by Technical Committee ISO/TC 20, Aircraft, the Secretariat of which is held by the British Standards Institution (BSI).

Work on this question by the Technical Committee began in 1962 and led, in 1963, to the adoption of a Draft ISO Recommendation.

In June 1964, this Draft ISO Recommendation (No. 738) was circulated to all the ISO Member Bodies for enquiry. It was approved, subject to a few modifications of an editorial nature, by the following Member Bodies:

Australia	France	Sweden
Belgium	Israel	Switzerland
Brazil	Italy	Turkey
Canada e	STANJapan RD PRI	United/Kingdom
Chile	Netherlands	Yugoslavia
Colombia	(standormals.iteh.a	i)
Czechoslovakia	Spain	-)

Two Member Bodies opposed the approval of the Draft: https://standards.iteh.ai/catalog/standards/sist/ac610e0b-8a66-4b14-b840-6b94a04f8933/iso-r-837-1968

U.S.S.R.

The Draft ISO Recommendation was then submitted by correspondence to the ISO Council, which decided, in October 1968, to accept it as an ISO RECOMMENDATION.

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AIRCRAFT SEAT RAILS AND PINS

1. SCOPE

This ISO Recommendation relates to the dimensions and loading of aircraft seat rails and pins.

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The dimensions of aircraft seat rails should comply with those given in Figure 1.

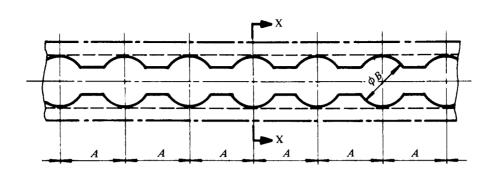
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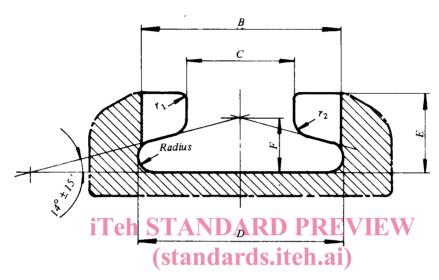
3. PINS

The dimensions of aircraft seat pins should comply with those given in Figure 2.

4. LOADING

Vertical down loads should be supported on the top of the rail and not on the base of the pin.





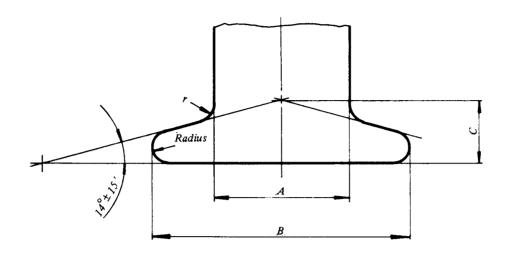
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Dimension	millimetres	inches
A	25.4*	1.0*
В	20.0 19.7	0.78 0.77
С	11.0 10.7	0.43 0.42
D	21.0 20.5	0.82 0.80
E	7.9 7.6	0.31 0.30
F	5.3 5.1	0.21 0.20
r ₁	0.5	0.02
r ₂	1.3	0.05

Tolerance on pitch = 0.2 mm (0.008 in). Tolerance on 20 pitches = 0.5 mm (0.020 in).

FIG. 1 - Aircraft seat rail



Eccentricity tolerance 0.2 mm (0.008 in)

Dimension Teh S	TANDmillimetres PRE	VIEWinches
A (standards.iteh.ai	0.39 0.38
Hetps://standards.it	ISO/R ₁ 82 <u>7</u> :1968 en.ai/catalog/stand <u>8</u> 23/sist/ac610e0b-	0.73 8a66-4b14-b86072
С	4.5 4.2	0.18 0.16
r	1.3	0.05

FIG. 2 - Aircraft seat pin

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