



Designation: D 4255/D 4255M – 01

Standard Test Method for In-Plane Shear Properties of Polymer Matrix Composite Materials by the Rail Shear Method¹

This standard is issued under the fixed designation D 4255/D 4255M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense.

1. Scope

1.1 This test method determines the in-plane shear properties of high-modulus fiber-reinforced composite materials by either of two procedures. In Procedure A, laminates clamped between two pairs of loading rails are tested. When loaded in tension the rails introduce shear forces in the specimen. In Procedure B, laminates clamped on opposite edges with a tensile or compressive load applied to a third pair of rails in the center are tested.

1.2 Application of this test method is limited to continuous-fiber or discontinuous-fiber-reinforced polymer matrix composites in the following material forms:

1.2.1 Laminates composed only of unidirectional fibrous laminae, with the fiber direction oriented either parallel or perpendicular to the fixture rails.

1.2.2 Laminates composed only of woven fabric filamentary laminae with the warp direction oriented either parallel or perpendicular to the fixture rails.

1.2.3 Laminates of balanced and symmetric construction, with the 0° direction oriented either parallel or perpendicular to the fixture rails.

1.2.4 Short-fiber-reinforced composites with a majority of the fibers being randomly distributed.

NOTE 1—Additional test methods for determining in-plane shear properties of polymer matrix composites may be found in Test Methods D 5379/D 5379M and D 5448/D 5448M, and Practice D 3518/D 3518M.

1.3 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.*

1.4 The values stated in either SI units or inch-pound units are to be regarded separately as standard. Within the text the inch-pounds units are shown in brackets. The values stated in

each system are not exact equivalents; therefore, each system must be used independently of the other. Combining values from the two systems may result in nonconformance with the standard.

2. Referenced Documents

2.1 ASTM Standards:

- D 792 Test Methods for Density and Specific Gravity (Relative Density) of Plastics by Displacement²
- D 883 Terminology Relating to Plastics²
- D 2584 Test Method for Ignition Loss of Cured Reinforced Resins³
- D 2734 Test Method for Void Content of Reinforced Plastics³
- D 3171 Test Method for Constituent Content of Composite Materials⁴
- D 3518/D 3518M Practice for In-Plane Shear Stress-Strain Response of Unidirectional Polymer Matrix Composite Materials by Tensile Test of $\pm 45^\circ$ Laminate⁴
- D 3878 Terminology for Composite Materials⁴
- D 5229/D 5229M Test Method for Moisture Absorption Properties and Equilibrium Conditioning of Polymer Matrix Composite Materials⁴
- D 5379/D 5379M Test Method for Shear Properties of Composite Materials by the V-Notched Beam Method⁴
- D 5448/D 5448M Test Method for In-Plane Shear Properties of Hoop Wound Polymer Matrix Composite Cylinders⁴
- E 4 Practices for Force Verification of Testing Machines⁵
- E 6 Terminology Relating to Methods of Mechanical Testing⁵
- E 111 Test Method for Young's Modulus, Tangent Modulus, and Chord Modulus⁵
- E 122 Practice for Choice of Sample Size to Estimate a Measure of Quality for a Lot or Process⁶
- E 177 Practice for Use of the Terms Precision and Bias in

¹ This test method is under the jurisdiction of ASTM Committee D30 on Composite Materials and is the direct responsibility of Subcommittee D30.04 on Lamina and Laminate Test Methods.

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² Annual Book of ASTM Standards, Vol 08.01.

³ Annual Book of ASTM Standards, Vol 08.02.

⁴ Annual Book of ASTM Standards, Vol 15.03.

⁵ Annual Book of ASTM Standards, Vol 03.01.

⁶ Annual Book of ASTM Standards, Vol 14.02.

ASTM Test Methods⁶

E 251 Test Methods for Performance Characteristics of Metallic Bonded Resistance Strain Gages⁵

E 456 Terminology Relating to Quality and Statistics⁶

E 1237 Guide for Installing Bonded Resistance Strain Gages⁵

E 1309 Guide for Identification of Composite Materials in Computerized Material Property Databases⁴

E 1434 Guide for Development of Standard Data Records for Computerization of Mechanical Test Data for High-Modulus Fiber-Reinforced Composite Materials⁴

E 1471 Guide for Identification of Fibers, Fillers, and Core Materials in Computerized Material Property Databases⁴

2.2 ASTM Adjunct:

Adjunct No. ADJD4255, Rail Shear Fixtures Machining Drawings⁷

3. Terminology

3.1 Terminology D 3878 defines terms relating to high-modulus fibers and their composites. Terminology D 883 defines terms relating to plastics. Terminology E 6 defines terms relating to mechanical testing. Terminology E 456 and Practice E 177 define terms relating to statistics. In the event of a conflict between terms, Terminology D 3878 shall have precedence over the other terminology standards.

NOTE 2—If the term represents a physical quantity, its analytical dimensions are stated immediately following the term (or letter symbol) in fundamental dimension form, using the following ASTM standard symbology for fundamental dimensions, shown within square brackets: [M] for mass, [L] for length, [T] for time, [θ] for thermodynamic temperature, and [nd] for nondimensional quantities. Use of these symbols is restricted to analytical dimensions when used with square brackets, as the symbols may have other definitions when used without the brackets.

3.2 Definitions of Terms Specific to This Standard:

3.2.1 *in-plane shear, n*—shear associated with shear forces applied to the edges of the laminate so that the resulting shear deformations occur in the plane of the laminate rather than through the thickness.

3.2.2 *offset shear stress* [M/(LT²)], *n*—the shear stress associated with an offset of the shear chord modulus of elasticity line along the strain axis (see 12.4).

3.2.3 *shear strength* [M/(LT²)], *n*—the shear stress carried by a material at failure under a pure shear condition.

3.2.4 *transition region, n*—a strain region of a stress-strain or strain-strain curve over which a significant change in the slope of the curve occurs within a small strain range.

3.2.4.1 *Discussion*—Many filamentary composite materials exhibit a nonlinear response during loading, such as seen in plots of either longitudinal stress versus longitudinal strain or transverse strain versus longitudinal strain. In certain cases, the nonlinear response may be conveniently approximated by a bilinear fit. There are several physical reasons for the existence of a transition region. Common examples include matrix cracking under tensile loading and ply delamination.

3.2.5 *traveler, n*—a small piece of the same material as, and processed similarly to, the test specimen, used for example to measure moisture content as a result of conditioning. This is also sometimes termed as a reference sample.

3.3 Symbols:

A = cross-sectional area of test specimen

B_y = percent bending of specimen

CV = coefficient of variation statistic of a sample population for a given property, %

F₁₂^o = offset shear stress, the value of the shear stress at the intersection of the stress-strain plot with a line passing through the offset strain value at zero stress and with a slope equal to the shear chord modulus of elasticity

F^u = ultimate shear stress

G = shear modulus of elasticity

h = specimen thickness

l = specimen length, the dimension parallel to the rails in the gage section

n = number of specimens

P_i = load carried by test specimen at *i*th data point

P^{max} = load carried by a test specimen that is the lesser of (1) the maximum load before failure, (2) the load at 5 % shear strain, or (3) the load at the bending limit (see 11.8.1)

s_{n-1} = sample standard deviation

x_i = measured or derived property for an individual specimen from the sample population

\bar{x} = sample mean (average)

γ = shear strain

ϵ = indicated normal strain from strain transducer

$\mu\epsilon$ = 10⁻⁶ m/m (10⁻⁶ in./in.)

τ_i = shear stress at *i*th data point

4. Summary of Test Method

4.1 *Procedure A: Two-Rail Shear Test*—A flat panel with holes along opposing edges is clamped, usually by through bolts, between two pairs of parallel steel loading rails, see Figs. 1 and 2. When loaded in tension, this fixture introduces shear forces in the specimen that produce failures across the panel. This test method is typical but not the only configuration usable. The two-rail shear fixtures can also be compression loaded. The load may be applied to failure.

4.1.1 If load-strain data are required, the specimen may be instrumented with strain gages. Two three-element strain gage rosettes are installed at corresponding locations on both faces of the specimen.

4.2 *Procedure B: Three-Rail Shear Test*—A flat panel, clamped securely between pairs of rails on opposite edges and in its center, is loaded by supporting the side rails while loading the center rails. See Figs. 3-5. A load on the center rail of either tension or compression produces a shear load in each section of the specimen. The load may be applied to failure.

4.2.1 The test fixture consists of three pairs of parallel rails usually bolted to the test specimen by through bolts. The two outside pairs of rails are attached to a base plate which rests on the test machine. A third pair (middle rails) is guided through a slot in the top of the base fixture. The unit is normally loaded

⁷ A copy of the detailed drawing for the construction of the fixtures shown in Figs. 1 and 2 is available at a nominal cost from ASTM Headquarters. Request Adjunct No. ADJD4255.

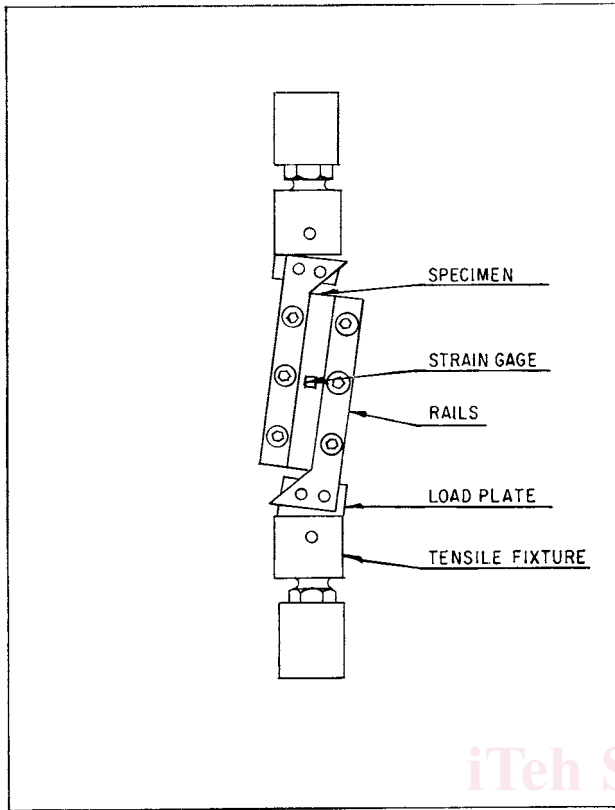


FIG. 1 Procedure A Assembly Rail Shear Apparatus

in compression. It is also permissible to load the middle rails in tension, but this requires attaching the base fixture to the test machine.

4.2.2 If load-strain data are required, the specimen may be instrumented with strain gages. Three-element strain gage rosettes are to be installed at corresponding locations on opposite faces of the specimen.

4.3 Detailed fixture drawings are available as ASTM Adjunct No. ADJD4255.

5. Significance and Use

5.1 These shear tests are designed to produce in-plane shear property data for material specifications, research and development, and design. Factors that influence the shear response and should therefore be reported include: material, methods of material preparation and lay-up, specimen stacking sequence, specimen preparation, specimen conditioning, environment of testing, specimen alignment and gripping, speed of testing, time at temperature, void content, and fiber volume reinforcement content. Properties that may be measured by this test method include:

- 5.1.1 In-plane shear stress versus shear strain response,
- 5.1.2 In-plane shear chord modulus of elasticity,
- 5.1.3 Offset shear stress, and

5.1.4 Maximum in-plane shear stress. In cases in which the strain at failure is greater than 5 %, the shear stress corresponding to 5 % shear strain should be reported.

6. Interferences

6.1 There are no standard test methods capable of producing a perfectly pure and uniform shear stress condition to failure for every material, although some test methods can come acceptably close for a specific material for a given engineering purpose. The off-axis load of the two-rail method introduces a comparatively small tensile load in the panel.

6.2 *Material and Specimen Preparation*—Poor material fabrication practices, lack of control of fiber alignment, and damage induced by improper specimen machining are known causes of high material data scatter in composites.

6.3 *Determination of Failure*—Rail shear specimens, especially thin ones, can buckle during load application. Buckling can be detected by measuring surface strains on opposite faces of the specimens with three-element strain gage rosettes. Data measured with the specimen in a buckled state are not representative of the material shear properties. Modulus data must be checked to confirm that buckling has not occurred in the modulus measurement range. Strength measurements must be checked to confirm that shear strength has not been influenced by specimen buckling. Failure by buckling should not be interpreted as indicating the maximum shear strength.

6.3.1 Ply delamination is another possible failure mode for laminates containing a large number of 45° plies. This failure reflects instability of 45° plies loaded in compression as contrasted to the overall buckling failure previously described. Differences in strain gage readings will not be noticeable, but the failure can be identified by delaminated plies in contrast to fiber breakage.⁸

6.4 *Gripping*—Failure through bolt holes indicates inadequate gripping. Alternate gripping methods are discussed in 7.2.3.

6.5 *End Effects*—This test method assumes a state of pure shear throughout the length of the specimen gage section. However, the gage section ends have zero shear stress because no traction and no constraints are applied there. A stress transition region exists between the ends and interior portions of the gage section. The length of this transition region determines the error induced in the material shear data.

7. Apparatus

7.1 *Micrometers*—The micrometer(s) shall use a suitable size diameter ball interface on irregular surfaces such as the bag-side of a laminate and a flat anvil interface on machined edges or very-smooth tooled surfaces. The accuracy of the instruments shall be suitable for reading to within 1 % of the sample length and thickness. For typical specimen geometries, an instrument with an accuracy of $\pm 2.5 \mu\text{m}$ [$\pm 0.0001 \text{ in.}$] is desirable for thickness measurement, while an instrument with an accuracy of $\pm 25 \mu\text{m}$ [$\pm 0.001 \text{ in.}$] is desirable for length measurement.

7.2 Rail Shear Fixtures

7.2.1 *Two-Rail Shear*—A two-rail shear fixture is shown in Figs. 1 and 2. Detailed fixture drawings are available as ASTM

⁸ A. K. Hussain and D. F. Adams, "The Wyoming-Modified Two-Rail Shear Test Fixture for Composite Materials," *Journal of Composites Technology and Research*, Vol 21, No. 4, October 1999, pp. 215-223.

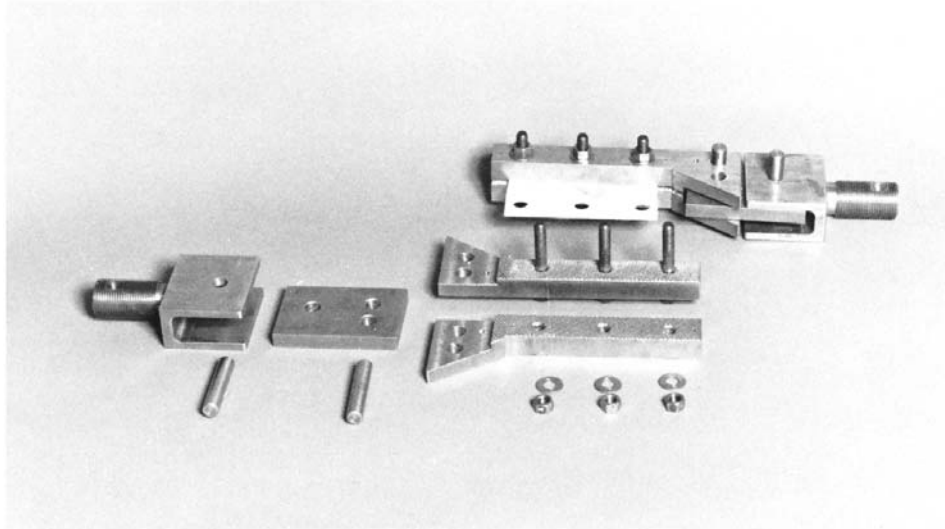


FIG. 2 Procedure A Partially Assembled Typical Test Fixture

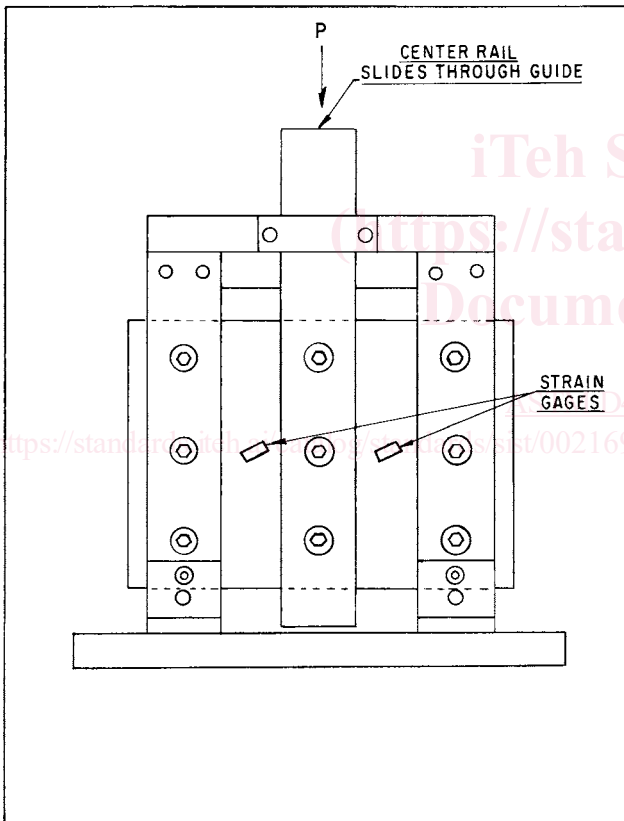


FIG. 3 Procedure B Assembly Rail Shear Fixture

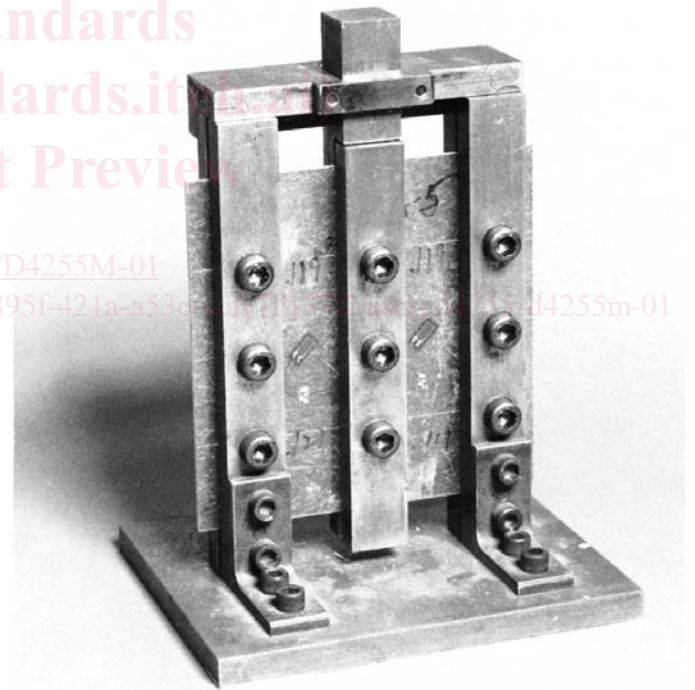


FIG. 4 Procedure B Assembled Typical Test Fixture

Adjunct No. ADJD4255. The test fixture consists of two pairs of rails which can clamp the test specimen with through bolts. The rails are then attached to the test machine through pins, a load plate that also aligns the rails with each other, and a clevis that connects directly to the test machine. This equipment is typical but not the only configuration usable. The two-rail shear fixture can be compression loaded. Also see 7.2.3 for rail modifications.

7.2.2 *Three-Rail Shear*—A three-rail shear fixture is shown in Figs. 3-5. Detailed fixture drawings are available as ASTM Adjunct ADJD4255. The test fixture consists of three pairs of rails that clamp the test specimen with through bolts. The two

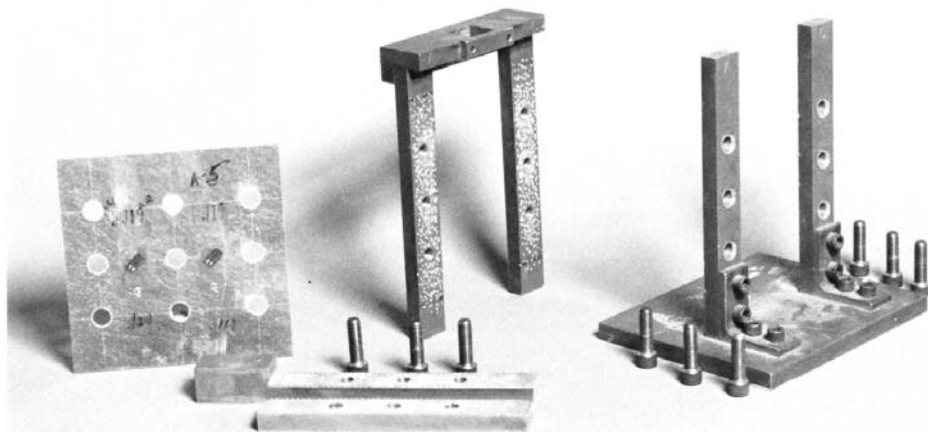


FIG. 5 Procedure B Disassembled Typical Test Fixture

outside pairs of rails are attached to a base plate that rests on the test machine. The third (middle) pair of rails are guided through a slot in the top of the base fixture. The unit shown is loaded in compression. The middle rails can be tensile loaded, which requires fastening the base fixture to the test machine. This equipment is typical but not the only configuration that is usable. Also see 7.2.3 for rail modifications.

7.2.3 Rail Modifications—The following list is not inclusive but is typical of methods used by various laboratories to meet the requirements of specific materials. Techniques that work for one material may be unacceptable for another. If these modifications are to be used as part of a specification, the rail grip system shall be completely specified and these modifications noted in the test report. These modifications have been used to grip the following specimens:

- 7.2.3.1 Abrasive paper or cloth bonded to the rails,
- 7.2.3.2 Machining V grooves in the rails,
- 7.2.3.3 Center punching rails in a random pattern,
- 7.2.3.4 Changing the number of bolt holes from three up to eight per rail and using smaller holes,
- 7.2.3.5 Soft metal shims,
- 7.2.3.6 Tabbing specimens in rail areas, and
- 7.2.3.7 Thermal spray surfaces.

7.3 Testing Machine—The testing machine shall conform with Practices E 4 and shall satisfy these requirements:

7.3.1 Testing Machine Heads—The testing machine shall have two loading heads with at least one movable head along the testing axis.

7.3.2 Platens/Adapter—One of the testing machine heads shall be capable of being attached to the lower half of the two-rail shear test fixture (described in 7.2.1) or of supporting the base of the three-rail fixture (described in 7.2.2) using an adapter or platen interface as required. The other head shall be

capable of being attached to the upper half of the fixture or of loading the center rail of the fixture. If required, one of the interfaces may be capable of relieving minor misalignments between heads, such as with a universal or a hemispherical ball joint.

7.3.3 Drive Mechanism—The testing machine drive mechanism shall be capable of imparting to the movable head a controlled displacement rate with respect to the stationary head. The displacement of the movable head shall be capable of regulation as specified in 11.3.

7.3.4 Load Indicator—The testing machine load-sensing device shall be capable of indicating the total load applied to the test specimen. This device shall be essentially free from response lag at the specified testing rate and shall indicate the load with an accuracy over the load range(s) of interest of within $\pm 1\%$ of the indicated value, as specified by Practices E 4. The load range(s) of interest may be fairly low for modulus evaluation or much higher for strength evaluation, or both, as required.

NOTE 3—Obtaining precision load data over a large range of interest in the same test, such as when both elastic modulus and maximum load are being determined, places extreme requirements on the load cell and its calibration. For some equipment a special calibration may be required. For some combinations of material and load cell, simultaneous precision measurement of both elastic modulus and maximum strength may not be possible, and measurement of modulus and strength may have to be performed in separate tests using a different load cell range for each test.

7.4 Strain-Indicating Device—Bonded resistance strain gages shall be used to measure strain. A minimum of two three-element strain gage rosettes are required, at corresponding locations on opposite faces of the specimen at the center of the gage section, as illustrated in Fig. 1, Fig. 3, and Figs. 6-9.

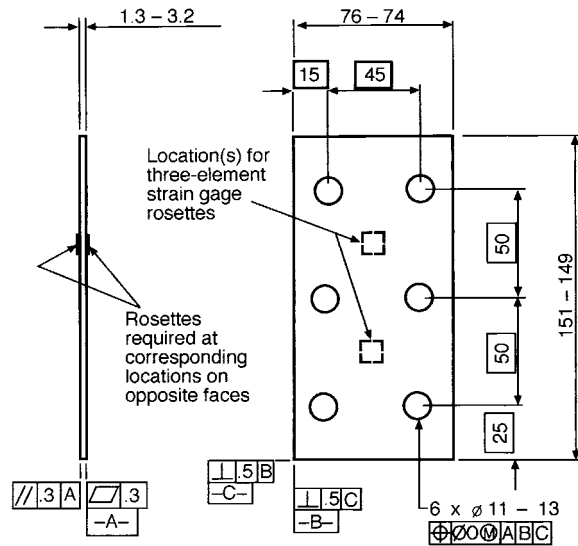


FIG. 6 Procedure A, Two-Rail Shear Specimen, SI Units

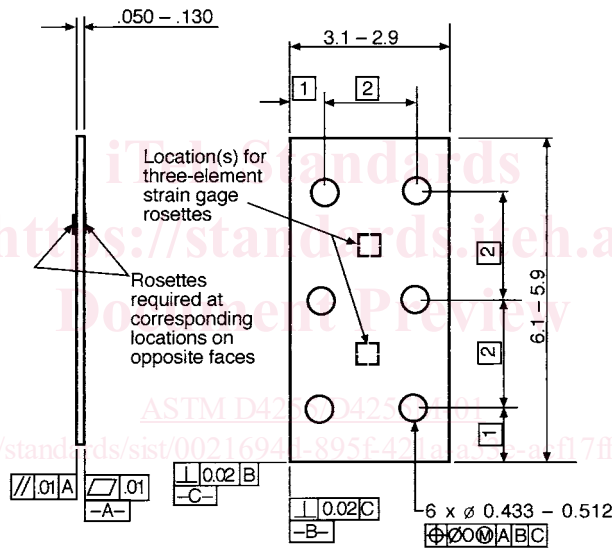


FIG. 7 Procedure A, Two-Rail Shear Specimen, Inch-Pound Units

7.4.1 *Bonded Resistance Strain Gages*—Strain gage selection is a compromise based on the type of material. An active gage length of 3 mm [0.125 in.] is recommended for most materials, although larger gages may be more suitable for some woven fabrics. The gage should not be so large that it lies within four specimen thicknesses of a rail. Gage calibration certification shall comply with Test Methods E 251. Strain gage rosettes with a minimum normal strain range of approximately 3 % (measuring 6 % shear strain) are recommended. When testing woven fabric laminates, gage selection should consider the use of an active gage length that is at least as large as the characteristic repeating unit of the weave. Some guidelines on strain gage use on composites follow. Additional general information can be found in the literature.^{9,10}

7.4.1.1 Surface preparation of fiber-reinforced composites in accordance with Guide E 1237 can penetrate the matrix material and cause damage to the reinforcing fibers, resulting in improper specimen failures. Reinforcing fibers should not be exposed or damaged during the surface preparation process. Consult the strain gage manufacturer regarding surface preparation guidelines and recommended bonding agents for composites, pending the development of a set of standard practices for strain gage installation surface preparation of fiber-reinforced composite materials.

¹⁰ *Manual on Experimental Methods of Mechanical Testing of Composites*, C. H. Jenkins, Ed., second edition, Society for Experimental Mechanics, Section II, Strain Measurement, 1998, pp. 25-84.

⁹ M. E. Tuttle and H. F. Brinson, "Resistance-Foil Strain Gage Technology as Applied to Composite Materials," *Experimental Mechanics*, 1984, Vol 24, No. 1, pp. 54-65, Errata noted in Vol. 26, No. 2, June 1986, pp. 153-154.