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Designation: B928/B928M-04a Designation: B 928/B 928M - 07

Standard Specification for High Magnesium Aluminum-Alloy Sheet and Plate for Marine Service and Similar Environments¹

This standard is issued under the fixed designation B 928/B 928M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

1. Scope*

1.1 This specification covers high magnesium (Note 1) marine application aluminum-alloy (Note 2), in those alloy-tempers shown in Table 2 [Table 3] and Table 4 [Table 5], for flat sheet, coiled sheet, and plate, in the mill finish that are intended for marine and similar environments:

Note 1-The term high magnesium in the general sense includes those alloys containing 3 % or more nominal magnesium.

NOTE 2-Throughout this specification use of the term alloy in the general sense includes aluminum as well as aluminum alloy.

1.2 Alloy and temper designations are in accordance with ANSI H35.1/H35.1(M) [H35.1M]. . The equivalent Unified Numbering System alloy designations are those of Table 1 preceded by A9, for example, A95083 for 5083 in accordance with Practice E 527.

1.3 The values stated in either SI units (Table 3 and Table 5) or inch-pound units (Table 2 and Table 4) are to be regarded separately as standard. The values stated in each system may not be exact equivalents; therefore each system shall be used independently of each other. Combining values from the two systems may result in non-conformance with the standard.

1.4 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.

1.5 This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.

2. Referenced Documents

2.1 The following documents of the issue in effect on the date of material purchase, unless otherwise noted, form a part of this specification to the extent referenced herein:

2.2 ASTM Standards:²

B 557 Test Methods offor Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products

B 557M Test Methods offor Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products Metric B 660 Practices for Packaging/Packing of Aluminum and Magnesium Products

- B 666/B 666M Practice for Identification Marking of Aluminum and Magnesium Products
- B 881 Terminology RelatedRelating to Aluminum- and Magnesium-Alloy Products

E 3 Practice for Preparation of Metallographic Specimens

E29Guide for Preparation of Metallographic Specimens

E 29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications

E 34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Alloys

E 55 Practice for Sampling Wrought Nonferrous Metals and Alloys for Determination of Chemical Composition E407Test

Methods for Microetching Metals and Alloys

E 527 Practice for Numbering Metals and Alloys (UNS)

*A Summary of Changes section appears at the end of this standard.

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¹ This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum-Alloy Wrought Products.

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² For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For Annual Book of ASTM Standards volume information, refer to the standard's Document Summary page on the ASTM website.



 TABLE 1 Chemical Composition Limits^{A,B,C}

Alloy	Silicon	Iron	Copper	Manganese	Magnesium	Chromium	Zinc	<u>Titanium</u> -	Other El Each	ements ^D Total ^E	Aluminum
5050	0.45	0.50	0.05	0.0 += 1.0	E 0 4- 0 0	0.05	0.4.4-0.0	0.00			un un a lus al a u
5059	0.45	0.50	0.25	0.6 to 1.2	5.0 to 6.0	0.25	0.4 to0.9	0.20	0.05 ^F	<u>0.15</u>	remainder
5083	0.40	0.40	0.10	0.40 to 1.0	4.0 to 4.9	0.05 to 0.25	0.25	0.15	0.05	0.15	remainder
5086	0.40	0.50	0.10	0.20 to 0.7	3.5 to 4.5	0.05 to 0.25	0.25	0.15	0.05	0.15	remainder
5383	0.25	0.25	0.20	0.7 to 1.0	4.0 to 5.2	0.25	0.40	0.15	0.05 ^G	0.15	remainder
5456	0.25	0.40	0.10	0.50 to 1.0	4.7 to 5.5	0.05 to 0.20	0.25	0.20	0.05	0.15	remainder

^A Limits are in weight percent maximum unless shown as a range or stated otherwise

^B Analysis shall be made for the elements for which limits are shown in this table.

^C For purposes of determining conformance to these limits, an observed value or a calculated value attained from analysis shall be rounded to the nearest unit in the last right-hand place of figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E 29.

^D Others include listed elements for which no specific limit is shown, as well as unlisted metallic elements, but doesn't include elements shown with composition limits in the footnotes. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic Others elements. Should any analysis by the producer or the purchaser establish that an Others element exceeds the limit of Each or that the aggregate of several Others elements exceeds the limit of Total, the material shall be considered nonconforming.

^E Other Elements—Total shall be the sum of unspecified metallic elements 0.010 % or more, rounded to the second decimal before determining the sum. ^F 0.05 to 0.25 Zr.

^G 0.20 Zr max.

E607Test Method for Atomic Emission Spectrometric Analysis Aluminum Alloys by the Point-to-Plane Technique, Nitrogen Atmosphere-Practice for Numbering Metals and Alloys in the Unified Numbering System (UNS)

E 716 Practices for Sampling Aluminum and Aluminum Alloys for Spectrochemical Analysis

E 1251 Test Method for Optical Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Argon Atmosphere, Point-to-Plane, Unipolar Self-Initiating Capacitor Discharge Test Method for Analysis of Aluminum and Aluminum Alloys by Atomic Emission Spectrometry

G 66 Test Method for Visual Assessment of Exfoliation Corrosion Susceptibility of 5xxx5XXX Series Aluminum Alloys (ASSET Test)

G 67 Test Method for Determining the Susceptibility to Intergranular Corrosion of 5xxx5XXX Series Aluminum Alloys by Mass Loss After Exposure to Nitric Acid (NAMLT Test)

2.3 ANSI Standards:³

H35.1H35.1/H35.1(M) Alloy and Temper Designation Systems for Aluminum H35.1(M)Alloy and Temper Designation Systems for Aluminum

H35.2 Dimensional Tolerances for Aluminum Mill Products

H35.2(M) Dimensional Tolerances for Aluminum Mill Products

3. Terminology

3.1 Definitions—Refer to Terminology B 881 for definitions of product terms used in this specification.

3.2 Definitions of Terms Specific to This Standard: d7efad8-0260-4ec1-9dbb-58e0049efc26/astm-b928-b928m-07

3.2.1 *exfoliation*—corrosion that proceeds laterally from the sites of initiation along planes parallel to the original rolling surface, generally at grain boundaries, forming corrosion products that force metal away from the body of the material, giving rise to a layered appearance.

3.2.2 intergranular corrosion—corrosion that preferentially occurs at, or adjacent to, the grain boundaries of a metal or alloy.

3.2.3 *sensitization*—the development of a continuous or nearly continuous grain boundary precipitate in 5xxx alloy-temper material, that causes the material to be susceptible to intergranular forms of corrosion.

3.2.4 *stress-corrosion cracking*—a cracking process that requires the simultaneous action of a corrodent, and sustained tensile stress. (This excludes corrosion-reduced sections, which fail by fast fracture. It also excludes intercrystalline or transcrystalline corrosion which can disintegrate an alloy without either applied or residual stress.)

4. Ordering Information

4.1 Orders for material to this specification shall include the following information:

4.1.1 This specification designation (which includes the number, the year, and the revision letter, if applicable),

- 4.1.2 Quantity in pieces or pounds [kilograms],
- 4.1.3 Alloy (see 7.1 and Table 1),
- 4.1.4 Temper (see 8.1 and Table 2 [and Table 4 [Table 3 and Table 5]),
- 4.1.5 For sheet, whether flat or coiled, and
- 4.1.6 Dimensions (thickness, width, and length or coil size).

4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:

^a Available from American National Standards Institute (ANSI), 25 W. 43rd St., 4th Floor, New York, NY 10036.

³ Available from Aluminum Association, Inc., 1525 Wilson Blvd., Suite 600, Arlington, VA 22209, http://www.aluminum.org.

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A.U	Specilfied Thicekness,	ITensile St	reength, ksi	GoppYield St	rength (0.2 % offset), ksi	MaElongation in 2 in.		
AlloyTemper	<u>in.</u>	Magnesiumin	Chro mium		Zmine	or 4× Diamester, min, %		Titanium <u>ax</u>
			Alloy 5059	<u>ax</u>				
H116	0.078 to 0.249	54.0		39.0		<u>10</u>		-
<u></u>	0.250 to 0.787	54.0	 	39.0	 	10		
	0.788 to 1.575	52.0		38.0		10		
H321	0.078 to 0.249	54.0		39.0		10 10 10 10		
	0.250 to 0.787	54.0	<u> </u>	39.0	<u> </u>	10		
	0.788 to 1.575	<u>52.0</u>	<u></u>	38.0	<u></u>	10		
			Alloy 5083					
H116	O ther Elements ^D	AL : 0.400	44.0		31.0		<u>10</u>	
H116	0.063 ther Elements ^D	Aluminumo 0.499		44.0		31.0		10
	Each		Aluminumo 0.499		31.0		<u></u> 12	
	0.500 to 1.250	44.0	<u></u>	<u>31.0</u>	<u></u>	<u>12</u> 12		
	To tal ^E o 1.500			31.0		12		
	1.251 to 1.500	44.0	<u></u>	31.0	<u></u>	<u>12</u> - 0.5012		
	1.501 to 3.000	5059		-0.45				
	<u>1.501 to 3.000</u>	41.0		29.0		<u>12</u> -0.4 to0.9		
H321	0 405 4: 0 407	<u></u>	0.6 to 1.2	-5.0 to6.0	0.25			
H321	0.125 to 0.187	44.0	56.0	31.0	· · ·	10		
	0 100 1- 1 500	<u></u>		- 0.15	-remainder	12		
	0.188 to 1.500 1.501 to 3.000	<u>44.0</u> 5083	<u>56.0</u> - 0.40	<u>31.0</u> 	0.10	<u>12</u> - 0.40to1.2		
	1.501 to 3.000	41.0	56.0	29.0		12		
			Alloy 50 <u>86</u>					
H116		<u>4.0 to4.9</u>	0.05 to 0.25	-0.25		- 0.158		
H116	0.063 to 0.249	40.0		28.0				
				-0.15	remainder	<u>8</u> 10		
	0.250 to 0.499	40.0		28.0		<u>10</u>		
	0.500 to 1.250	5086		-0.40	1 • 4 1 •	_		
0.50	0.500 to 1.250	40.0 	 0 .20 to 0 .7	28.0 - 3.5 to4.5	0 .05 to 0 .25	10 - 0.		
0.00	1.251 to 2.000	40.0	0.2010 0.1	28.0		<u>10</u>		
05			Cumle		eview			
25		<u>-0.15</u>	50.0	-0.05		-0.158		
<u>H321</u>	0.063 to 0.249 remainder0.250 to 0.320	<u>40.0</u> <u>40.0</u>	<u>52.0</u> 52.0	28.0 28.0	<u></u>	<u>8</u> -9		
	0.250 to 0.320	40.0	A <u>52.0</u> / B9	2 <u>28.0</u> 39281	1-07	_9		
ttps://sta	ndards.iteh.ai/catal	log/standards	sist/7 Alloy 5383	8-0260-4e	c1-9dbb-58e0049	efc26/astm-b928-		
H116	0.118 to 0.500	48.0	<u></u>	33.0	<u></u>	10 10 10		
	0.501 to 2.000	48.0	<u></u>	33.0	<u></u>	10		
H321	0.118 to 0.500	$\frac{48.0}{48.0}$	<u></u>	33.0		<u>10</u> 10		
	0.501 to 2.000	48.0	<u></u>	33.0	<u> </u>	10		
H116	0.063 to 0.499	5383	Alloy 5456	-0.25		-0.25		
H116	0.063 to 0.499	46.0		33.0		10		
11110	0.000 10 0.433	<u> </u>	0.7 to 1.0	-4.0 to5.2	0.25	- 0.4012		
	0.500 to 1.250	46.0	<u></u>	33.0	<u></u>	12		
	1 0E1 to 1 500	<u></u>		-0.05 ^G 21.0		0.152		
	1.251 to 1.500	44.0		31.0		1 <u>2</u> 12		
	remainder1.501 to 3.000		····	29.0 29.0	····			
	<u>1.501 to 3.000</u> 3.001 to 4.000	<u>41.0</u> 5456	<u></u>	<u>29.0</u> 0.25	<u></u>	<u>12</u> 		
	3.001 to 4.000	5456 40.0		- 0.25 25.0		- 0.4012 12		
H321	0.001 10 4.000	<u> </u>	- 0.50-to1.0	-4.7 to5.5	0 .05 to0 .20	<u>-0.25</u>		
H321	0.100 to 0.187	48.0	59.0	34.0	0.00.00.20	10		
11021	0.188 to 0.499	46.0	59.0	33.0	46.0	<u>12</u> 12		
	0.100 10 0.700	10.0	00.0	30.0		<u></u>		
	0.500 to 1.500	<u> </u>	- 0.05	-0.15	remainder44 0	12		
	0.500 to 1.500 0.500 to 1.500	0.20 44.0	- 0.05 56.0	- 0.15 31.0	remainder44.0 44.0	12 12		

^A LiTo determine conformance to this-ar specification-w, eigach value for t-pensile streengt-mh aximum und for yield strength ss-shall be rowunded to the nearest 0.1 ksi a-rnd each value for elongeation to the nearest 0.5 %, both in accordance with thed rounding metheod of Prwactisce E 29. ^B Analysis sThall be m bade for the elements for which limits are shown in this table.

CFor purposes o f determining conformance to these limits, an observed value or a calculated value attained from analysis shall be rounded to the nearest unit in the last right hand place of figures used in expressing the specified limit, in accordance with the rounding off meth od of Practice E29. ^DOthers include listed ete ments for which no specific limit is shown, as well as unlisted metallic elements, but doesn't include elements shown with composition limits

in the footnotes. The producer may analyze sam ples for trace elements not specified in the specification. However, such analysis is not required and may not cover a l metallic Others elements. Should any analysis by the producer or the purchaser establ ish that an Other s element exceeds the limit of Each or that the aggregate of several Others elements. Should any analysis by the material shall be considered Anoneonforming.

FOther Elements—Total shall be the sum of unspecified metallic elements 0.010% or more, rounded to the second decimal before determining the sum.

^F0.05 to 0.25 Zr.

^G0.20 Zr max A1.

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TABLE-2.3 Longitudinal Mechanical Property Limits, [SInch-Pound Units]^{A,B}

	Specified Thicknes	s, -in. mm	Te	ensile Sti	rength, ksi MF	PaYield Strength	(0.2 % offset),-k			jation in 2 in. ameter , min, %	<u>c</u>
lemper	r <u>over</u>	through		min	max	min	max	— <u>MPa</u>	in 50 mm	$\frac{\text{in } 5 \times \text{Diamete}}{(5.65 \sqrt{A})}$	<u>r</u>
					Alloy 5059					· · · · ·	_
H116 H116 H321 H321	0.078 to 0.249 <u>1.99</u> 0.250 to 0.787 <u>6.30</u> 0.788 to1.575 <u>12.50</u> <u>20.00</u> 0.078 to 0.249 <u>1.99</u> 0.250 to 0.787	54.0 6.30 54.0 12.50 52.0 20.00 40.00 54.0 54.0 12.50 52.0 20.00 40.00 54.0 54.0 54.0 6.30 54.0 10.50		370 370		39.0 270 39.0 270 38.0 270 39.0 270 260 39.0 270 39.0 270 39.0 270 270 270 270 270 39.0 270 270 270		+0 10 +0 10 +0 10 +0 10	 10 10 <u>10</u> 	-	
	<u>6.30</u> <u>12.50</u> 0.788 to 1.575	<u>12.50</u> 20.00 52.0		<u>370</u> <u>370</u> 360	 	<u>270</u> <u>38.0</u>	 	10 	 <u>10</u> 10		
	20.00	<u>40.00</u>		360		260			10	-	
					Alloy 5083					-	
04.0	1.60 12:50 0.500 to 1.250 0.500 to 1.250 	$ \begin{array}{r} 12.50 \\ H116 \\ 0 \\ 30.00 \\ \underline{44.0} \\ 30.00 \\ \underline{-12} \\ \end{array} $	1000000000000000000000000000000000000	<u>305</u> -44.0 305	<u></u> <u></u> <u>31.0</u> <u>305</u>	215 	<u></u> <u>12</u> <u>215</u>	10 10 1.251 to 1.500 1.251 to 1.500	 10 <u>44.0</u> 	<u></u>	<u>10</u>
80.00 H321 H321	40.00 1.501 to 3.000 -0.125te0.187 3.20 0.188 to 1.500 -1.501 to 3.000 1.500 1.500 1.500	$ \begin{array}{r} -12 \\ 41.0 \\ 285 \\ -44.0 \\ 5.00 \\ 44.0 \\ 5.00 \\ -41.0 \\ 40.00 \\ \end{array} $	56.0 12.50	29.0 200 -31.0 305 -56.0 305	43.0 385 31.0 305 385	$S_{\frac{15}{215}}^{\frac{15}{15}}$	\mathbf{d}_{10}^{12}		$\begin{array}{c} \dots \\ \underline{12} \\ \underline{12} \\ \underline{10} \\ 10 \end{array}$		
43.0		_		÷							
	TZ										
		_	40.00		Alloy 5086		281480.00		<u>285</u>	385	200
H116 H116	<u>-0.063te0.249</u> <u>1.60</u> <u>-0.250</u> te0.499 <u>6.30</u> <u>-0.500te1.250</u> <u>12.50</u> <u>-1.251</u> te2.000	$ \frac{12.50}{-40.0} 30.00}{-40.0} $	<u>40.00</u> 'catalog/sta	275 275 275	Alloy 5086	<u>A B928/B9</u> <u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u>	$\frac{201}{1-0}$	80-58 00-490 <u>8</u> 10 10 <u></u>	elc26/a	<u>385</u> istm-b928	<u>200</u> -b928m-(
H116 H116	-0.063to0.249 1.60 -0.250 -0.250 -0.500to1.250 12.50 -1.251 to2.000 30.00	$ \begin{array}{r} $	<u> </u>	275 275 275 275 275	IS/SIST <u>/7.0</u> 7 	<u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u>	0-4ec <u>1_9</u> db 	8 10 10 10 <u></u>	efc2.6/8 9 9 9 9	1.000	
H116 H116	$\begin{array}{r} -0.063 \text{to} 0.249 \\ \hline 1.60 \\ \hline 0.250 \\ \hline 10.500 \text{to} 0.499 \\ \hline 0.500 \text{to} 1.250 \\ \hline 12.50 \\ \hline 1.251 \\ \hline 1.251 \\ \hline 1.60 \\ \hline 1.60 \\ \hline \end{array}$	$ \begin{array}{r} $	<u> </u>	275 275 275 275 275 275 275	S/SISI <u>C/C</u> I7 355 355	<u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> 195 <u>195</u>	0-4ec <u>1.9</u> db 	8 10 10 10 8 8	elc26/a	1.000	
H116 H116 H321	0.063to0.249 1.60 0.250 to0.490 6.30 0.500to1.250 1.251 to2.000 30.00 1.60	$ \begin{array}{r} $	<u> </u>	275 275 275 275 275 275 275	S/SIST	<u>195</u> 28.0 195 28.0 195 28.0 195 195 195	0-4ec <u>1.9</u> db 	8 10 10 10 8	elc26/a 9 9 9 9 9	1.000	
H116 H116 H321	0.063te0.249 1.60 0.250 to0.490 6.30 0.500to1.250 1.251 to2.000 1.60 1.60 1.60 0.118 te 0.500 0.501 te2.000	<u>6.30</u> <u>40.0</u> <u>12.50</u> <u>40.0</u> <u>30.00</u> <u>50.00</u> <u>-10</u> <u>6.30</u> <u>8.00</u> <u>48.0</u> <u>12.50</u> <u>48.0</u>	<u> </u>	275 275 275 275 275 275 275 275 275 275	S/SISI <u>C/C</u> I7 355 355	<u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>230</u> 33.0	0-4ec <u>1.9</u> db 	8 10 10 10 10 10 10 10 10		1.000	
H116 H116 H321 H321 H116	-0.063te0.249 1.60 -0.250 te0.491 6.30 -0.500to1.250 12.50 1.251 te2.000 30.00 1.60 1.60 0.00 1.60 1.60 0.00 1.60 1.60 0.00 0.00 1.60 0.00 0.0118 to 0.500 0.118 to 0.500 3.00	6.30 9 40.0 12.50 40.0 30.00 9 40.0 50.00 -10 6.30 8.00 -10 6.30 8.00 -10 6.30 8.00 -10 6.30 8.00 -12.50 48.0 12.50 48.0 12.50 -12.50 -12.50 -12.50 -12.50 -12.50 -12.50 -12.50 -10 -10 -10 -10 -10 -10 -10 -1	<u> </u>	275 275 275 275 275 275 275 275 330 330 330 330 330 330 330 330	S/SIST	<u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>28.0</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>230</u> 33.0 230 33.0 230 33.0 230	2001-00	8 10 10 10 10 10 10 10 10 10 10		1.000	
H116 H116 H321 H321 H116 H116 H321	$\begin{array}{c} -0.063tc0.249 \\ \hline 1.60 \\ -0.250 \\ \hline 10.250 \\ \hline 10.500to1.250 \\ \hline 12.50 \\ \hline 1.251 \\ \hline 1.251 \\ \hline 1.60 \\ \hline 1.60 \\ \hline 1.60 \\ \hline 0.118 to 0.500 \\ \hline 0.501 to2.000 \\ \hline 0.501 to2.000 \\ \hline 12.50 \\ \hline 0.118 to 0.500 \\ \hline 0.118 to 0.500 \\ \hline \end{array}$	<u>6.30</u> <u>40.0</u> <u>12.50</u> <u>40.0</u> <u>30.00</u> <u>50.00</u> <u>10</u> <u>6.30</u> <u>8.00</u> <u>48.0</u> <u>12.50</u> <u>48.0</u> <u>12.50</u> <u>48.0</u> <u>50.00</u> <u>48.0</u>	<u> </u>	275 275 275 275 275 275 275 275 275 275	S/SISU	<u>195</u> <u>28.0</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>195</u> <u>230</u> 230 230 33.0 230 33.0	2001-0	8 8 10 10 10 8 8 9 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 		1.000	
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Temper										
	- over	through	min	max	min	max in 50 mm	i n 5× Diameter (5.65 √A)		or ×4 Diameter, min, %	`e
		through	min	max	min	<u>max</u> in 50 mm	$\frac{\text{in 5} \times \text{Diameter}}{(5.65 \sqrt{A})}$			
				Alloy 5059						
H116	1.99	6.30	370		270		10			
	6.30	12.50	370		270		10			
	12.50	20.00	370		270			10		
	20.00	40.00	360		260			10		
H321	1.99	6.30	370		270		10			
	6.30	12.50	370		270		10			
	12.50	20.00	370		270			10		
	20.00	40.00	360		260			10		
	20.00								-	
				Alloy 5083			0.15		-	
H116		to 0.249	1.60	12.50	305	<u></u>	215	<u></u>	10	
<u>H116</u>	<u>0.118</u>	to 0.249	44.0	12.50	305	<u></u>	<u>31.0</u>	<u></u>	10	
			12.50	30.00	305		215			1
	0.250	to 0.499	44.0	30.00	305	<u></u>	<u>31.0</u>	<u></u>		1
	30.00	-40.00	305		215	· · · ·		10		
	40.00	-80.00	285		200			10		
H321	-3.20	-5.00	305	385	215	295	10			
	-5.00	-12.50	305	385	215	295	12			
	12.50	-40.00	305	385	215	295	····	10		
	40.00	-80.00	285	385	200	295		10		
				Alloy 5086	;				-	
H116	-1.60	-6.30	275	l len	195	laafas	8	····	-	
	-6.30	-12.50	275		195	····	10	····		
	12.50	-30.00	275		195			-9		
	30.00	-50.00	275	S:/#SU	195	<u>Iras.Ite</u>		-9		
				Alloy 5383	3				-	
H116	3.00	12.70	330	ocen	230	Preview	10	10		
	12.70	50.80	330		230			10		
-1321 3.00	12.70	330		230		10	10			
			12.70	50.80	330		230			1
	<u>0.118</u>	to 0.236	44.0	55.0	<u> B 31.0</u> B	<u>928M-07</u>	230			1
nttps://s	tandards.i	teh.ai/catalo	g/standard	Alloy 5456	efad8-026	0-4ec1-9dbb-5	58e0049efc20	5/as		
				Alloy 5086						
H116	-1.60	-12.50	315		230	····	10	····		
	12.50	-30.00	315		230	····	· · · ·	10		
	30.00	-40.00	305		215	····		10		
	40.00	-80.00	285		200			10		
	80.00	110.00	275		170			10		
H321	4	.00	12.50	315	405	230	315	12	<u></u>	
H321		to 0.320	40.0	52.0	28.0	230	315	12	<u></u>	
	12.50	-40.00	305	385	215	305		10		

TABLE-3_4 Long Transverse Mechanical Property Limits-[S, Inch-Pound Units]^{A,B}

^A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest <u>0.1-MP ksi</u> a-and each value for elongation to the nearest 0.5 %, both in accordance with the rounding method of Practice E 29.

^B The basis for establishment of mechanical property limits is shown in Annex A1. CElongations in 50 mm apply for thicknesses up through 12.50 mm and in 5× diameter (5.65 \sqrt{A}) for thicknesses over 12.50 mm where A is the cross-sectional area of the specimen.

4.2.1 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (see 11.1),

4.2.2 Whether Practices B 660 applies and, if so, the levels of preservation, packaging, and packing required (see 15.3), and),

4.2.3 Whether certification is required (see Section 13), and.

4.2.4 Whether tensile testing should be in the longitudinal or long transverse direction (see 8.5).

5. Responsibility for Quality Assurance

5.1 *Responsibility for Inspection and Tests*—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use his own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order or at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set

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Temper	Specified Th	ickness, mm	Tensile Strength, MPa		Yield Strengt	th (0.2 % offset), MPa	Elongat	Elongation, min, % ^C		
	over	through	min	max	min	max	<u>in 50 mm</u>	$\frac{\text{in 5x Diamete}}{(5.65 \sqrt{A})}$		
				Alle	oy 5083					
<u>H116</u>	<u>3.00</u> 6.00	<u>6.00</u> 12.50	<u>305</u> 305	<u></u>	215 215 215	<u></u> 	<u>10</u> 10	<u></u>		
<u>H321</u>	3.00	<u>6.00</u>	<u>305</u>	<u>380</u>	<u>215</u>	<u></u>	<u>10</u>	<u></u>		
				Alle	oy 5086					
H321	6.00	8.00	275	355	195	<u></u>	10	<u></u>		

^ATo determine conformance to this specification, each value for tensile strength shall be rounded to the nearest 1 MPa and each value for elongation to the nearest 0.5 %

both in accordance with the rounding method of Practice E 29.

^BThe basis for establishment of mechanical property limits is shown in Annex A1.

^CElongations in 50 mm apply for thicknesses up through 12.50 mm and in 5× diameter (5.65 V A) for thicknesses over 12.50 mm where A is the cross-sectional area of the specimen.

forth in this specification where such inspections are deemed necessary to ensure that material conforms to prescribed requirements.

5.2 Lot Definition—An inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, cast or melt lot, and thickness, subjected to inspection at one time.

6. General Quality

6.1 Unless otherwise specified, the material shall be supplied in the mill finish, shall be uniform as defined by the requirements of this specification and shall be commercially sound. Any requirement not so covered is subject to negotiation between producer and purchaser.

6.2 Each coil, sheet and plate shall be examined to determine conformance to this specification with respect to general quality and identification marking. On approval of the purchaser, however, the producer may use a system of statistical quality control for such examinations.

7. Chemical Composition

7.1 Limits—The sheet and plate shall conform to the chemical composition limits specified in Table 1. Conformance shall be determined by the producer, by the analysis of samples taken at the time the ingots are cast or samples taken from the finished or semifinished product. If the producer has determined the chemical composition of the material during the course of manufacture, additional sampling and analysis of the finished product shall not be required.

7.2 Number of Samples—The number of samples taken for determination of chemical composition shall be as follows:

7.2.1 When samples are taken at the time the ingots are cast, at least one sample shall be taken for each group of ingots cast simultaneously from the same source of molten metal.

7.2.2 When samples are taken from the finished or semifinished product, a sample shall be taken to represent each 4000 lb [2000 kg] or fraction thereof, of material in the lot, except that not more than one sample shall be required per piece.

7.3 Methods of Sampling—Samples for determination of chemical composition shall be taken in accordance with one of the following methods:

7.3.1 Samples for chemical analysis shall be taken by drilling, sawing, milling, turning, or clipping a representative piece or pieces to obtain a prepared sample of not less than 75 g. Sampling shall be in accordance with Practice E 55.

7.3.2 Sampling for spectrochemical analysis shall be in accordance with Practices E 716. Samples for other methods of analysis shall be suitable for the form of material being analyzed and the type of analytical method used.

7.4 Methods of Analysis—The determination of chemical composition shall be made in accordance with suitable chemical (Test Methods E 34), or spectrochemical (Test Methods E607 and E1251), or spectrochemical (Test Method E 1251) methods. Other methods may be used only when no published ASTM method is available. In case of dispute, the methods of analysis shall be as agreed upon between the producer and purchaser.

8. Tensile Properties of Material as Supplied

8.1 Limits—The sheet and plate shall conform to the requirements for tensile properties as specified in Table 2 [Table 3]. 8.1.1Tensile property limits for sizes not covered in] or Table 4 [Table 5]. Table 2 [Table 3] includes specification limits for tensile properties in the longitudinal direction. Table 4 [Table 5] includes specification limits for tensile properties in the long transverse direction.

8.1.1 Tensile property limits for sizes not covered in Table 2 or Table 4 [Table 3 or Table 5] shall be as agreed upon between the producer and purchaser and shall be so specified in the contract or purchase order.

8.2 Number of Samples—One sample shall be taken from each end of each parent coil, or parent plate, but no more than one sample per 2000 lb [1000 kg] of sheet or 4000 lb [2000 kg] of plate, or part thereof, in a lot shall be required. Other procedures for selecting samples may be employed if agreed upon between the producer and purchaser.