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Part 5: Submarine specials

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Page

Contents

Fore	eword	iv
1	Scope	
2	Normative references	
3	Terms and definitions	
4	Test-related physical quantities	2
5	General test conditions	4
6	Test 5.1 — Meander test6.1Description6.2Analysis and presentation of results of a meander test6.3Evaluation for supercritical or high damping6.4Evaluation of the time to half-value6.5Designation of a meander test	
7	 Test 5.2 — Vertical overshoot test 7.1 Description 7.2 Analysis and presentation of results of a vertical overshoot test 7.3 Designation of a vertical overshoot test 	
8	 Test 5.3 — Neutral level flight test 8.1 Description 8.2 Analysis and presentation of results of a neutral level flight test 8.3 Designation of a neutral level flight test 	
9	Test 5.4 — Critical speed test 9.1 General ISO: 13643-5:2013 9.2 Description Description 9.3 Analysis and presentation of results of a critical speed test 9.4 Designation of a critical speed test	
Bibl	liography	

Foreword

ISO (the International Organization for Standardization) is a worldwide federation of national standards bodies (ISO member bodies). The work of preparing International Standards is normally carried out through ISO technical committees. Each member body interested in a subject for which a technical committee has been established has the right to be represented on that committee. International organizations, governmental and non-governmental, in liaison with ISO, also take part in the work. ISO collaborates closely with the International Electrotechnical Commission (IEC) on all matters of electrotechnical standardization.

The procedures used to develop this document and those intended for its further maintenance are described in the ISO/IEC Directives, Part 1. In particular the different approval criteria needed for the different types of ISO documents should be noted. This document was drafted in accordance with the editorial rules of the ISO/IEC Directives, Part 2. www.iso.org/directives

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The committee responsible for this document is ISO/TC 8, *Ships and marine technology*, Subcommittee SC 6, *Navigation and ship operations* **STANDARD PREVIEW**

ISO 13643 consists of the following parts, under the general title *Ships and marine technology* — *Manoeuvring of ships*:

- Part 1: General concepts, quantities and test conditions5:2013
- https://standards.iteh.ai/catalog/standards/sist/0850e3fd-6c6e-4441-8eba-
- Part 2: Turning and yaw checking 4f01dbbef7bb/iso-13643-5-2013
- Part 3: Yaw stability and steering
- Part 4: Stopping, acceleration, traversing
- Part 5: Submarine specials
- Part 6: Model test specials

Ships and marine technology — Manoeuvring of ships —

Part 5: **Submarine specials**

1 Scope

This part of ISO 13643 defines symbols and terms and provides guidelines for the conduct of tests to give evidence about the manoeuvring ability in the vertical plane of submarines and models. It is intended to be read in conjunction with ISO 13643-1.

2 Normative references

The following documents, in whole or in part, are normatively referenced in this document and are indispensable for its application. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

ISO 13643-1, Ships and marine technology — Manoeuvring of ships — Part 1: General concepts, quantities and test conditions **Teh STANDARD PREVIEW**

3 Terms and definitions (standards.iteh.ai)

For the purposes of this document, the following terms and definitions apply.

https://standards.iteh.ai/catalog/standards/sist/0850e3fd-6c6e-4441-8eba-4f01dbbef7bb/iso-13643-5-2013

3.1

meander test

manoeuvring test to establish a submarine's manoeuvring characteristics and to verify the submarine's dynamic stability in the vertical plane

3.2

vertical overshoot test

manoeuvring test to determine the effectiveness of the stern planes when initiating and terminating changes of depth

3.3

neutral level flight test

manoeuvring test to determine the trim angle and the hydroplane angles at which the submarine maintains a constant dived depth at any given speed during submerged operation

Note 1 to entry: Neutral level flight is obtained

- for submarines with retracted bow planes by using a definite trim angle and a definite angle of stern planes, and
- for submarines with non-retractable bow planes, by using definite angles of the bow and stern planes for arbitrary trim angles (preferably $\theta_S = 0^\circ$).

3.4

critical speed test

manoeuvring test to determine the speed at which the effect of the hydroplanes is reversed during submerged operation

4 Test-related physical quantities

Test-related physical quantities are according to <u>Table 1</u>; general quantities and concepts are according to ISO 13643-1.

Symbol	CC-	CI	Concept		
Symbol	code	SI unit	Term	Definition or explanation	
<i>a</i> ₁	-	rad m ² s ^{-2 a}	Coefficient	For regression approximation	
<i>a</i> ₂	_	m ² s ⁻²	Coefficient	For regression approximation	
b_1	-	rad m ² s ^{-2 a}	Coefficient	For regression approximation	
<i>b</i> ₂	-	m ² s ⁻²	Coefficient	For regression approximation	
Cc	CCR	1	Damping ratio	Ratio between damping constant $\ln 2/t_{\frac{1}{2}}$ and eigenfrequency $2\pi/T_0$ of the undamped oscillation	
<i>c</i> ₁	_	rad m ² s ^{-2 a}	Coefficient	For regression approximation	
<i>d</i> ₁	_	rad m ² s ⁻² a	Coefficient	For regression approximation	
$F_{\rm V}$	FVC	N	Vertical force	_	
MA	MAX	-	Main axis	(See ISO 13643-1)	
M _T	MYT	Nm	Trim moment	_	
Т	TIP	s iTe	h STANDA Period of oscillation	Period of the damped oscillation,	
			(standard		
T ₀	TIP0	S	Period of oscillation	Period of the undamped oscillation	
t _A	TIA	https://star s	ISO 1364 adards.iteh.ai/catalog/standa Response(time)ef7bb/iso	Times to achieve the trim amplitudes, θ_{Ai} , i = 1, 2,	
t _C	TIC	S	Overshoot time	Time from putting the stern planes into the oppo- site direction until reaching maximum trim angle	
tt	TIT	S	Levelling-off time	Time from putting the stern planes into the opposite direction until reaching maximum depth change	
t _{1/2}	TI05	S	Time to half-value	Time elapsed before the envelope of time-depend- ent trim variation has decreased by half	
V _{CR}	VCR	m s ^{-1 b}	Critical speed	Speed at which the effect of the hydroplanes is reversed	
$V_{\rm F}$	VF	m s ^{-1 b}	Final speed	Speed at the end of test (run)	
V_0	V0	m s ^{-1 b}	Initial speed	(See ISO 13643-1)	
V _{0i}	VOI	m s ^{-1 b}	Initial speed	For neutral level flight and vertical overshoot test: For individual runs of the test	
V _{0m}	V0M	m s-1 b	Mean test speed	_	
z ₀	ZO	m	Dived depth	Vertical coordinate in the earth-fixed axis system of the origin of the submarine (see ISO 13643-1, Table 2) at any given time	
Z00	Z00	m	Initial dived depth	At the commencement of the test (run)	
^a For an	gles, the unit	° (degree) may b	e used .		
b The ur	nit kn, commo	on in navigation,	may be used.		

Table 1 — Test-related physical quantities

Symbol	CC- code	SI unit	Concept		
Symbol			Term	Definition or explanation	
ż _{0F}	DZDTF	m s ^{-1 b}	Rate of depth change	When constant trim angle, $ heta_{ extsf{F}}$, has been reached	
$\Delta z_{0\rm E}$	DZ0E	m	Response depth change	Change of depth relative to z_{00} when trim angle is changed by $\Delta \theta_{\rm E}$	
$\Delta z_{0\rm F}$	DZ0F	m	Final change of dived depth	Under steady final conditions, only defined for a dynamically stable boat	
Δz_{0M}	DZ0M	m	Levelling-off depth change	Maximum change of depth, relative to z_{00} + Δz_{0E}	
$\Delta \delta_{ m Si}$	DANSI	rad ^a	Test stern plane angle	Relative to δ_{S0} ; if necessary, an equivalent stern plane angle shall be given, e.g. for submarines with X-planes: $\frac{1}{4} (\Delta \delta_{Ai1} + \Delta \delta_{Ai2} + \Delta \delta_{Ai3} + \Delta \delta_{Ai4})$.	
$\Delta heta_{ m E}$	DTETPE	rad ^a	Execute change of trim angle	For meander test: $\theta_{\rm E} - \theta_{\rm S0}$ Specified change of trim angle relative to $\theta_{\rm S0}$ at which the stern planes are returned to their initia settings $\delta_{\rm S0}$ For vertical overshoot test: $\theta_{\rm E} - \theta_{\rm S0}$ Specified change of trim angle relative to $\theta_{\rm ext}$ at	
			FANDARD	Specified change of trim angle relative to θ_{S0} at which the stern planes are applied in the opposite direction (δ_{Si})	
$\delta_{ m B}$	ANB	rad a 🕓	Bow plane angle	(See ISO 13643-1)	
$\delta_{ m B0}$	ht ANB0	tps://standards.ite rad ^a	ISO 13643-5:2013 Initial boxaplanes/sist/0 angle Ubbef7bb/iso-13643-	For meander, vertical overshoot, and critical speed tests: SUe3fd-6c6e-4441-8eba- Bow plane angle at the commencement of the test (valid for neutral level flight)	
			Bow plane angle for neutral level flight	Result of neutral level flight test	
$\delta_{\rm S}$	ANS	rad ^a	Stern plane angle	(See ISO 13643-1)	
δ_{SX}	ANSX	rad ^a	Angle of stabilising fin or of the fixed post of a stern plane	Relative to the horizontal plane in MA, positive when leading edge tilts upwards	
δ_{SX0}	ANSX0	rad ^a	Angle of stabilising fin or of the fixed post of a stern plane, for neutral level flight	_	
$\delta_{ m Si}$	ANSI	rad ^a	Test stern plane angle	Relative to δ_{S0} ; if necessary, an equivalent stern plane angle shall be given, e.g. for submarines wit X-planes: $\frac{1}{4} (\delta_{Ai1} + \delta_{Ai2} + \delta_{Ai3} + \delta_{Ai4})$.	
	ANSO	VSO rad ^a	Initial stern plane angle	For meander, vertical overshoot and critical speed test:	
$\delta_{ m S0}$				Stern plane angle at the commencement of the tes (valid for neutral level flight)	
~50			Stern plane angle for neutral level flight	Result of neutral level flight test:	
				If necessary, an equivalent stern plane angle shall be given, e.g. for submarines with X-planes: $\frac{1}{4} (\delta_{A01} + \delta_{A02} + \delta_{A03} + \delta_{A04}).$	

Table 1 (continued)

Crumb al	CC- code	SI unit	Concept	
Symbol			Term	Definition or explanation
$ heta_{\mathrm{A}}$	TRIMSA	rad ^a	Trim amplitude	Absolute value of the respective extreme θ_{Ai} , i = 1, 2, 3, of the change of trim with reference to θ_0
$ heta_{ m E}$	TRIMSE	rad ^a	Execute trim angle	$\theta_{\rm S0} + \Delta \theta_{\rm E}$
$ heta_{ m F}$	TRIMSF	rad ^a	Trim angle at the end of run	Constant trim angle for the respective hydroplane settings
$\theta_{\rm S}$	TRIMS	rad ^a	Trim angle	(See ISO 13643-1)
$ heta_{ m SS}$	TRIMSS	rad ^a	Overshoot angle	In the vertical plane after applying the stern planes in the opposite direction
	TRIMS0rad aInitial trim angletests: Trim angle at the commencement for neutral level flight)TRIMS0rad aTrim angle for neu-		Initial trim angle	For meander, vertical overshoot, and critical speed tests:
$ heta_{ m S0}$		Trim angle at the commencement of the test (valid for neutral level flight)		
				Result of neutral level flight test:
				Trim angle at which the submarine maintains a constant dived depth

 Table 1 (continued)

b The unit kn, common in navigation, may be used.

5 General test conditions (standards.iteh.ai)

In addition to the general test conditions outlined in JSQ_13643-1, the following specific test conditions shall be complied with. https://standards.iteh.ai/catalog/standards/sist/0850e3fd-6c6e-4441-8eba-

- During the test, including the approach phase, each successive position of the ship shall be recorded at suitable time intervals (usually every second).
- The submarine shall be trimmed according to the results of the neutral level flight test (see <u>Clause 8</u>).
- Dived depth and water depth shall be sufficient (a clearance of at least one boat's length to the surface and to the bottom shall be maintained). For model tests, surface and bottom effects shall be excluded by the use of suitable measures.
- The bow plane angle shall remain unaltered.
- There shall be no relocation of mass (e.g. due to movements of the crew) during the conduct of any test. Unavoidable shifts of mass are to be compensated and recorded.

6 Test 5.1 — Meander test

6.1 Description

A series of tests for different initial speeds must be conducted since damping and time constants of the motion of the submarine are speed dependent, and a boat that proves to be stable at low speeds may become unstable at higher speeds.

For safety reasons, the series of tests shall be commenced with a low initial speed, V_0 .

The submarine shall approach on a steady speed, V_0 , before commencing the test. During the test, the propulsion plant settings must remain unaltered and the heading kept as constant as possible. Heading and rudder movements shall be recorded throughout the test (ideally, at intervals of 1 s). If the submarine is equipped with planes acting simultaneously in the horizontal and the vertical directions

(e.g. X-planes), these planes should be controlled in such a way that a steady heading is maintained as a matter of priority.

After the submarine has been moving ahead for at least two minutes without significant movements of rudder and planes, the stern planes are set to the specified test stern plane angle, $\Delta \delta_{Si}$, as fast as possible and must be held there until the trim angle has deviated from the initial trim angle, θ_{S0} , by the specified execute change of trim angle, $\Delta \theta_E$. At this point, the stern planes are reversed to the initial position and held until the test is completed.

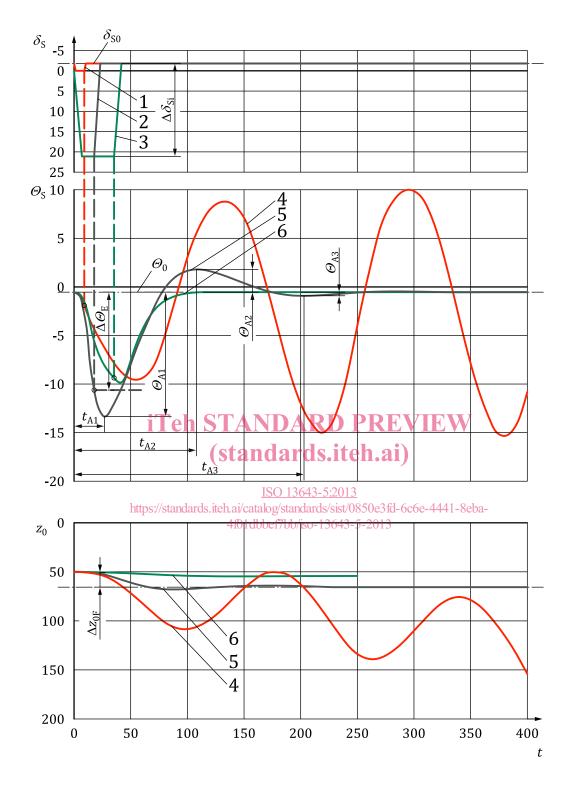
The stern plane impetus moves the submarine from its equilibrium condition. Test stern plane angle, $\Delta \delta_{Si}$, and execute change of trim angle, $\Delta \theta_E$, shall be selected in such a way that the stern plane impetus acts as quickly and powerfully as possible, and the submarine has at least three measurable trim amplitudes, θ_A , in the case of a subsequent oscillation. Only data after completion of the simulation are to be evaluated.

Because of the stern plane impetus, the submarine turns about its transverse axis and, in doing so, changes its trim and dived depth in the direction in which the planes were acting. Submarines with high damping approach a different dived depth without oscillation about the transverse axis. If the damping is less, the submarine starts to oscillate about the transverse axis. As long as the oscillation is damped, the submarine is stable and approaches a different constant dived depth. On the other hand, if the trim amplitude, θ_A , increases, the submarine is dynamically unstable. The mean dived depth may alter also.

If the submarine demonstrates pronounced instability, the test is to be stopped immediately for safety reasons.

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Key

1 V_0 = 14,0 kn, $\Delta \delta_{\rm Si}$ = 2°, $\Delta \theta_{\rm E}$ = 2° (boat dynamically supercritically damped 6 unstable) 2 V_0 = 6,0 kn, $\Delta \delta_{\rm Si}$ = 23°, $\Delta \theta_{\rm E}$ = 10° (boat dynamically $\delta_{\rm S}, \Theta_{\rm S}$ in ° unstable) $3 V_0 = 3.9 \text{ kn}, \Delta \delta_{\text{Si}} = 23^\circ, \Delta \theta_{\text{E}} = 9^\circ$ (boat supercritically t in s damped) 4 dynamically unstable in m Z_0 5 dynamically unstable

Figure 1 — Time history

6.2 Analysis and presentation of results of a meander test

6.2.1 Evaluation for subcritical damping

The following data are obtained from the test:

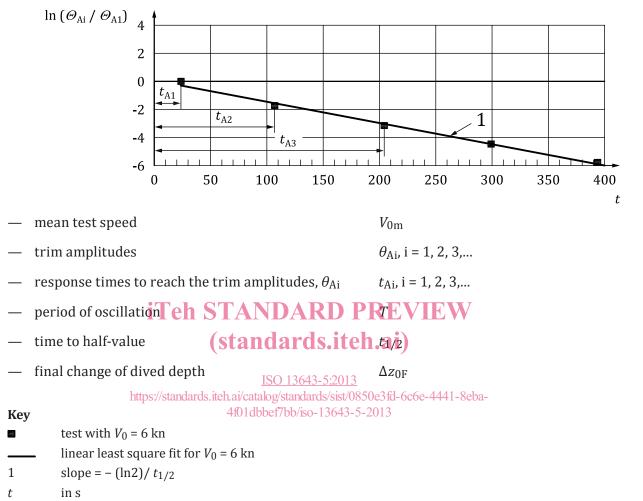


Figure 2 — Evaluation

6.2.2 Damping ratio

The damping ratios shall be plotted against the mean test speed, V_{0m} , as shown in Figure 3.

$$C_{\rm c} = \frac{1}{\sqrt{\left(\frac{2 \pi t_{\frac{1}{2}}}{T \ln 2}\right)^2 + 1}}$$