

Designation: B928/B928M - 09

StandardSpecification for High Magnesium Aluminum-Alloy Sheet and Plate for Marine Service and Similar Environments¹

This standard is issued under the fixed designation B928/B928M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (\$\epsilon\$) indicates an editorial change since the last revision or reapproval.

1. Scope*

1.1 This specification covers high magnesium (Note 1) marine application aluminum-alloy (Note 2), in those alloy-tempers shown in Table 2 [Table 3] and Table 4 [Table 5], for flat sheet, coiled sheet, and plate, in the mill finish that are intended for marine and similar environments:

Note 1—The term high magnesium in the general sense includes those alloys containing $3\,\%$ or more nominal magnesium.

Note 2—Throughout this specification use of the term alloy in the general sense includes aluminum as well as aluminum alloy.

- 1.2 Alloy and temper designations are in accordance with ANSI H35.1/H35.1(M). The equivalent Unified Numbering System alloy designations are those of Table 1 preceded by A9, for example, A95083 for 5083 in accordance with Practice E527.
- 1.3 The values stated in either SI units (Table 3 and Table 5) or inch-pound units (Table 2 and Table 4) are to be regarded separately as standard. The values stated in each system may not be exact equivalents; therefore, each system shall be used independently of each other. Combining values from the two systems may result in non-conformance with the standard.
- 1.4 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.
- 1.5 This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.

2. Referenced Documents

2.1 The following documents of the issue in effect on the date of material purchase, unless otherwise noted, form a part of this specification to the extent referenced herein:

¹ This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum Alloy Wrought Products.

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2.2 ASTM Standards:²

B557 Test Methods for Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products

B557M Test Methods for Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products (Metric)

B660 Practices for Packaging/Packing of Aluminum and Magnesium Products

B666/B666M Practice for Identification Marking of Aluminum and Magnesium Products

B881 Terminology Relating to Aluminum- and Magnesium-Alloy Products

E3 Guide for Preparation of Metallographic Specimens

E29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications

E34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Alloys

E50 Practices for Apparatus, Reagents, and Safety Considerations for Chemical Analysis of Metals, Ores, and Related Materials

E527 Practice for Numbering Metals and Alloys in the Unified Numbering System (UNS)

E607 Test Method for Atomic Emission Spectrometric Analysis Aluminum Alloys by the Point to Plane Technique Nitrogen Atmosphere (Withdrawn 2011)³

E716 Practices for Sampling and Sample Preparation of Aluminum and Aluminum Alloys for Determination of Chemical Composition by Spectrochemical Analysis

E1251 Test Method for Analysis of Aluminum and Aluminum Alloys by Spark Atomic Emission Spectrometry

G66 Test Method for Visual Assessment of Exfoliation Corrosion Susceptibility of 5XXX Series Aluminum Alloys (ASSET Test)

G67 Test Method for Determining the Susceptibility to Intergranular Corrosion of 5XXX Series Aluminum Alloys by Mass Loss After Exposure to Nitric Acid (NAMLT Test)

² For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For Annual Book of ASTM Standards volume information, refer to the standard's Document Summary page on the ASTM website.

³ The last approved version of this historical standard is referenced on www.astm.org.

TABLE 1 Chemical Composition Limits^{A,B,C}

| Alloy | Silicon | n Iron Coppe | Cannar | Manganasa | Magnesium | Chromium | Zinc | Titanium | Other Elements ^D | | Alumaimuma |
|-------|---------|--------------|--------|-------------|------------|--------------|-----------|----------|-----------------------------|--------------------|------------|
| | | | Copper | Manganese | | | | | Each | Total ^E | Aluminum |
| 5059 | 0.45 | 0.50 | 0.25 | 0.6 to 1.2 | 5.0 to 6.0 | 0.25 | 0.4 to0.9 | 0.20 | 0.05 ^F | 0.15 | remainder |
| 5083 | 0.40 | 0.40 | 0.10 | 0.40 to 1.0 | 4.0 to 4.9 | 0.05 to 0.25 | 0.25 | 0.15 | 0.05 | 0.15 | remainder |
| 5086 | 0.40 | 0.50 | 0.10 | 0.20 to 0.7 | 3.5 to 4.5 | 0.05 to 0.25 | 0.25 | 0.15 | 0.05 | 0.15 | remainder |
| 5383 | 0.25 | 0.25 | 0.20 | 0.7 to 1.0 | 4.0 to 5.2 | 0.25 | 0.40 | 0.15 | 0.05 ^G | 0.15 | remainder |
| 5456 | 0.25 | 0.40 | 0.10 | 0.50 to 1.0 | 4.7 to 5.5 | 0.05 to 0.20 | 0.25 | 0.20 | 0.05 | 0.15 | remainder |

^A Limits are in weight percent maximum unless shown as a range or stated otherwise.

TABLE 2 Longitudinal Mechanical Property Limits, Inch-Pound Units^{A,B}

| | Specified Thickness, | Tensi | le Strength, ksi | Yield Strength (0 | Elongation in 2 in. | | |
|-------------------|----------------------|--------------|--------------------|-------------------|---------------------|------------------------|--|
| Temper | in. | min | max | min | | or 4× Diameter, min, % | |
| | | | Alloy 5059 | | | | |
| H116 | 0.078 to 0.249 | 54.0 | | 39.0 | | 10 | |
| | 0.250 to 0.787 | 54.0 | | 39.0 | | 10 | |
| | 0.788 to 1.575 | 52.0 | | 38.0 | | 10 | |
| H321 | 0.078 to 0.249 | 54.0 | ··· | 39.0 | | 10 | |
| | 0.250 to 0.787 | 54.0 | | 39.0 | | 10 | |
| | 0.788 to 1.575 | 52.0 | | 38.0 | | 10 | |
| | | | Alloy 5083 | | | | |
| | | | | | | | |
| H116 | 0.063 to 0.499 | 44.0 | | 31.0 | | 10 | |
| | 0.500 to 1.250 | 44.0 | tondondo | 31.0 | | 12 | |
| | 1.251 to 1.500 | 44.0 | tanuarus. | 31.0 | | 12 | |
| | 1.501 to 3.000 | 41.0 | | 29.0 | | 12 | |
| H321 | 0.125 to 0.187 | 44.0 | 56.0 | 31.0 | | 10 | |
| | 0.188 to 1.500 | 44.0 | 56.0 | -31.0 | | 12 | |
| | 1.501 to 3.000 | 41.0 | 56.0 | 29.0 | | 12 | |
| | | | Alloy 5086 | | | | |
| H116 | 0.063 to 0.249 | 40.0 | | 28.0 | | 8 | |
| | 0.250 to 0.499 | 40.0 | ΓM B928/B928M-09 | 28.0 | | 10 | |
| | 0.500 to 1.250 | 40.0 | 00.500.1500.4001.1 | 28.0 | T 11 0/ . 1 | 10 | |
| //standards | 1.251 to 2.000 | ards/40.0/64 | 2ta596-1700-4391-b | 21 28.0 XX be5 | /adbt/astm-b | 928-692810-09 | |
| H321 ^C | 0.063 to 0.249 | 40.0 | 52.0 | 28.0 | | 8 | |
| | 0.250 to 0.320 | 40.0 | 52.0 | 28.0 | | 9 | |
| | | | Alloy 5383 | | | | |
| H116 | 0.118 to 0.500 | 48.0 | ••• | 33.0 | | 10 | |
| | 0.501 to 2.000 | 48.0 | ••• | 33.0 | | 10 | |
| H321 | 0.118 to 0.500 | 48.0 | ••• | 33.0 | | 10 | |
| | 0.501 to 2.000 | 48.0 | *** | 33.0 | | 10 | |
| | | | Alloy 5456 | | | | |
| H116 | 0.063 to 0.499 | 46.0 | • • • | 33.0 | | 10 | |
| | 0.500 to 1.250 | 46.0 | | 33.0 | | 12 | |
| | 1.251 to 1.500 | 44.0 | | 31.0 | | 12 | |
| | 1.501 to 3.000 | 41.0 | | 29.0 | | 12 | |
| | 3.001 to 4.000 | 40.0 | | 25.0 | | 12 | |
| H321 | 0.100 to 0.187 | 48.0 | 59.0 | 34.0 | | 10 | |
| | 0.188 to 0.499 | 46.0 | 59.0 | 33.0 | | 12 | |
| | 0.500 to 1.500 | 44.0 | 56.0 | 31.0 | | 12 | |
| | 1.501 to 3.000 | 41.0 | 54.0 | 29.0 | | 12 | |

^A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest 0.1 ksi and each value for elongation to the nearest 0.5 %, both in accordance with the rounding method of Practice E29.

 $^{^{\}it B}$ Analysis shall be made for the elements for which limits are shown in this table.

^C For purposes of determining conformance to these limits, an observed value or a calculated value attained from analysis shall be rounded to the nearest unit in the last right-hand place of figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E29.

^D Others include listed elements for which no specific limit is shown, as well as unlisted metallic elements, but doesn't include elements shown with composition limits in

^D Others include listed elements for which no specific limit is shown, as well as unlisted metallic elements, but doesn't include elements shown with composition limits in the footnotes. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic Others elements. Should any analysis by the producer or the purchaser establish that an Others element exceeds the limit of Each or that the aggregate of several Others elements exceeds the limit of Total, the material shall be considered nonconforming.

E Other Elements—Total shall be the sum of unspecified metallic elements 0.010 % or more, rounded to the second decimal before determining the sum.

F 0.05 to 0.25 Zr.

^G 0.20 Zr max.

^B The basis for establishment of mechanical property limits is shown in Annex A1.

^C Tentative — properties subject to revision.

TABLE 3 Longitudinal Mechanical Property Limits [SI Units] A,B

| Tompor | Specified T | nickness, mm | Tensil | Tensile Strength, MPa | | | Elongation, min, % ^C | |
|-------------------|-------------|--------------|---------------------|-----------------------|---------|-----|---------------------------------|---------------------------------|
| Temper | over | through | min | max | min | max | in 50 mm | in 5× Diamete $(5.65 \sqrt{A})$ |
| | | | | Alloy 5059 | | | | |
| H116 | 1.99 | 6.30 | 370 | | 270 | | 10 | |
| | 6.30 | 12.50 | 370 | | 270 | | 10 | |
| | 12.50 | 20.00 | 370 | | 270 | | | 10 |
| | 20.00 | 40.00 | 360 | | 260 | | | 10 |
| H321 | 1.99 | 6.30 | 370 | | 270 | | 10 | |
| | 6.30 | 12.50 | 370 | | 270 | | 10 | |
| | 12.50 | 20.00 | 370 | | 270 | | | 10 |
| | 20.00 | 40.00 | 360 | | 260 | | | 10 |
| | | | | Alloy 5083 | | | | |
| H116 | 1.60 | 12.50 | 305 | | 215 | | 10 | |
| | 12.50 | 30.00 | 305 | | 215 | | | 10 |
| | 30.00 | 40.00 | 305 | | 215 | | | 10 |
| | 40.00 | 80.00 | 285 | | 200 | | | 10 |
| H321 | 3.20 | 5.00 | 305 | 385 | 215 | | 10 | |
| | 5.00 | 12.50 | 305 | 385 | 215 | | 12 | |
| | 12.50 | 40.00 | 305 | 385 | 215 | | | 10 |
| | 40.00 | 80.00 | 285 | 385 | 200 | | | 10 |
| | | | | Alloy 5086 | | | | |
| H116 | 1.60 | 6.30 | 275 | | 195 | | 8 | |
| | 6.30 | 12.50 | 275 | | 195 | | 10 | |
| | 12.50 | 30.00 | 275 | | 195 | | | 9 |
| | 30.00 | 50.00 | 275 | | 195 | | | 9 |
| H321 ^D | 1.60 | 6.30 | 275 | 355 | 195 | | 8 | |
| | 6.30 | 8.00 | 275 | 355 | 195 | | 9 | |
| 11446 | | 10.50 | | Alloy 5383 | | | | |
| H116 | 3.00 | 12.50 | 330 | | 230 | | 10 | |
| 11001 | 12.50 | 50.00 | 330 | andär | 230 | | | 10 |
| H321 | 3.00 | 12.50 | 330 | amuan | 230 | | 10 | |
| | 12.50 | 50.00 | 330 | | 230 | | | 10 |
| 11440 | 1.00 | 40.50 | | Alloy 5456 | 4 0 000 | | | |
| H116 | 1.60 | 12.50 | DS://ST2 315 315 | 0121170:55 | 230 | | 10 | |
| | 12.50 | 30.00 | | | 230 | | | 10 |
| | 30.00 | 40.00 | 305 | | 215 | | | 10 |
| | 40.00 | 80.00 | 285 275 | it Prev | 200 | | | 10 |
| 11004 | 80.00 | 110.00 | | 405 | 170 | | | 10 |
| H321 | 2.50 | 4.00 | 330 | 405 | 235 | | 10 | • • • |
| | 4.00 | 12.50 | 315 | 405 | 230 | | 12 | |
| | 12.50 | 40.00 | AST 305 | 8/R928 385 | 215 | | | 10 |
| | 40.00 | 80.00 | AS IIVI 1285 | 8/B928 370 U9 | 200 | | | 10 |

A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest 1 MPa and each value for elongation to the nearest 0.5 %, both in accordance with the rounding method of Practice E29. $^{\mathcal{B}}$ The basis for establishment of mechanical property limits is shown in Annex A1.

TABLE 4 Long Transverse Mechanical Property Limits, Inch-Pound Units^{A,B}

| Temper | Specified Thickness, in. | Tensile Strength, ksi | | Yield Strength (0.2 % offset), ksi | | Elongation in 2 in. or ×4 Diameter, | |
|-------------------|--------------------------|-----------------------|------|------------------------------------|-----|--------------------------------------|--|
| remper | | min | max | min | max | min, % | |
| | | Alloy | 5083 | | | | |
| H116 | 0.118 to 0.249 | 44.0 | | 31.0 | | 10 | |
| | 0.250 to 0.499 | 44.0 | | 31.0 | | 10 | |
| H321 | 0.118 to 0.236 | 44.0 | 55.0 | 31.0 | | 10 | |
| | | Alloy | 5086 | | | | |
| H321 ^C | 0.250 to 0.320 | 40.0 | 52.0 | 28.0 | | 10 | |

A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest 0.1 ksi and each value for elongation to the nearest 0.5 %, both in accordance with the rounding method of Practice E29. B The basis for establishment of mechanical property limits is shown in Annex A1.

^C Elongations in 50 mm apply for thicknesses up through 12.50 mm and in $5 \times$ diameter $(5.65 \sqrt{A})$ for thicknesses over 12.50 mm where A is the cross-sectional area of the specimen.

^D Tentative — properties subject to revision.

^C Tentative — properties subject to revision.

TABLE 5 Long Transverse Mechanical Property Limits [SI Units]^{A,B}

| Temper | Specified Thickness, mm | | Tensile Strength, MPa | | Yield Strength (0.2 % offset), MPa | | Elongation, min, % ^C | |
|-------------------|-------------------------|---------|-----------------------|-------|------------------------------------|-----|---------------------------------|----------------------------------|
| | over | through | min | max | min | max | in 50 mm | in 5x Diameter $(5.65 \sqrt{A})$ |
| | | | | Alloy | 5083 | | | |
| H116 | 3.00 | 6.00 | 305 | | 215 | | 10 | |
| | 6.00 | 12.50 | 305 | | 215 | | 10 | |
| H321 | 3.00 | 6.00 | 305 | 380 | 215 | | 10 | |
| | | | | Alloy | 5086 | | | |
| H321 ^D | 6.00 | 8.00 | 275 | 355 | 195 | | 10 | |
| | | | | | | | | |

^ATo determine conformance to this specification, each value for tensile strength shall be rounded to the nearest 1 MPa and each value for elongation to the nearest 0.5 %, both in accordance with the rounding method of Practice E29.

2.3 ANSI Standards:4

H35.1/H35.1(M) Alloy and Temper Designation Systems for Aluminum

H35.2 Dimensional Tolerances for Aluminum Mill ProductsH35.2(M) Dimensional Tolerances for Aluminum Mill Products

2.4 Other Standards

CEN EN 14242 Aluminum and aluminum alloys. Chemical analysis. Inductively coupled plasma optical emission spectral analysis⁵

3. Terminology

- 3.1 *Definitions*—Refer to Terminology B881 for definitions of product terms used in this specification.
 - 3.2 Definitions of Terms Specific to This Standard:
- 3.2.1 *exfoliation*—corrosion that proceeds laterally from the sites of initiation along planes parallel to the original rolling surface, generally at grain boundaries, forming corrosion products that force metal away from the body of the material, giving rise to a layered appearance.
- 3.2.2 *intergranular corrosion*—corrosion that preferentially occurs at, or adjacent to, the grain boundaries of a metal or alloy.
- 3.2.3 *sensitization*—the development of a continuous or nearly continuous grain boundary precipitate in 5xxx alloy-temper material, that causes the material to be susceptible to intergranular forms of corrosion.
- 3.2.4 stress-corrosion cracking—a cracking process that requires the simultaneous action of a corrodent, and sustained tensile stress. (This excludes corrosion-reduced sections, which fail by fast fracture. It also excludes intercrystalline or transcrystalline corrosion which can disintegrate an alloy without either applied or residual stress.)

4. Ordering Information

4.1 Orders for material to this specification shall include the following information:

- 4.1.1 This specification designation (which includes the number, the year, and the revision letter, if applicable),
 - 4.1.2 Quantity in pieces or pounds [kilograms],
 - 4.1.3 Alloy (see 7.1 and Table 1),
- 4.1.4 Temper (see 8.1 and Table 2 and Table 4 [Table 3 and Table 5]),
 - 4.1.5 For sheet, whether flat or coiled, and
 - 4.1.6 Dimensions (thickness, width, and length or coil size).
- 4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:
- 4.2.1 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (see 11.1),
- 4.2.2 Whether Practices B660 applies and, if so, the levels of preservation, packaging, and packing required (see 15.3),
 - 4.2.3 Whether certification is required (see Section 13), and.
- 4.2.4 Whether tensile testing should be in the longitudinal or long transverse direction (see 8.5).

5. Responsibility for Quality Assurance

- 5.1 Responsibility for Inspection and Tests—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use his own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order or at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set forth in this specification where such inspections are deemed necessary to ensure that material conforms to prescribed requirements.
- 5.2 Lot Definition—An inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, cast or melt lot, and thickness, subjected to inspection at one time.

6. General Quality

6.1 Unless otherwise specified, the material shall be supplied in the mill finish, shall be uniform as defined by the requirements of this specification and shall be commercially sound. Any requirement not so covered is subject to negotiation between producer and purchaser.

^BThe basis for establishment of mechanical property limits is shown in Annex A1.

^CElongations in 50 mm apply for thicknesses up through 12.50 mm and in 5x diameter (5.65 \sqrt{A}) for thicknesses over 12.50 mm where A is the cross-sectional area of the specimen.

D Tentative — properties subject to revision.

⁴ Available from Aluminum Association, Inc., 1525 Wilson Blvd., Suite 600, Arlington, VA 22209, http://www.aluminum.org.

⁵ Available from European Committee for Standardization (CEN), Avenue Marnix 17, B-1000 Brussels, http://www.cenorm.be.

6.2 Each coil, sheet and plate shall be examined to determine conformance to this specification with respect to general quality and identification marking. On approval of the purchaser, however, the producer may use a system of statistical quality control for such examinations.

7. Chemical Composition

- 7.1 *Limits*—The sheet and plate shall conform to the chemical composition limits specified in Table 1. Conformance shall be determined by the producer, by taking samples in accordance with E716 when the ingots are poured and analyzing those samples in accordance with E607, E1251, E34 or EN 14242. At least one sample shall be taken for each group of ingots poured simultaneously from the same source of molten metal. If the producer has determined the chemical composition during pouring of the ingots, they shall not be required to sample and analyze the finished product.
- 7.2 If it becomes necessary to analyze the finished or semifinished product for conformance to chemical composition limits, the method used to sample the finished or semifinished product for the determination of chemical composition shall be by agreement between the producer and the purchaser. Analysis shall be performed in accordance with E716, E607, E1251, E34 or EN 14242 (ICP method). The number of samples taken for determination of chemical composition shall be as follows:
- 7.2.1 When samples are taken from the finished or semifinished product, a sample shall be taken to represent each 4000 lb [2000 kg] or fraction thereof, of material in the lot, except that not more than one sample shall be required per piece.
- 7.3 Other methods of analysis or in the case of dispute may be by agreement between the producer and the purchaser.

8. Tensile Properties of Material as Supplied

- 8.1 *Limits*—The sheet and plate shall conform to the requirements for tensile properties as specified in Table 2 [Table 3] or Table 4 [Table 5]. Table 2 [Table 3] includes specification limits for tensile properties in the longitudinal direction. Table 4 [Table 5] includes specification limits for tensile properties in the long transverse direction.
- 8.1.1 Tensile property limits for sizes not covered in Table 2 or Table 4 [Table 3 or Table 5] shall be as agreed upon between the producer and purchaser and shall be so specified in the contract or purchase order.
- 8.2 *Number of Samples*—One sample shall be taken from each end of each parent coil, or parent plate, but no more than one sample per 2000 lb [1000 kg] of sheet or 4000 lb [2000 kg] of plate, or part thereof, in a lot shall be required. Other procedures for selecting samples may be employed if agreed upon between the producer and purchaser.
- 8.3 *Test Specimens*—Geometry of test specimens and the location in the product from which they are taken shall be as specified in Test Methods B557 or B557M, with the exception that the test direction will be as specified in 8.5.
- 8.4 *Test Methods*—The tension test shall be made in accordance with Test Methods B557 or B557M.
- 8.5 Testing Direction—Tensile testing shall be in the longitudinal direction unless the long transverse direction is speci-

fied in the contract or purchase order. Tensile testing direction shall be noted on all documentation.

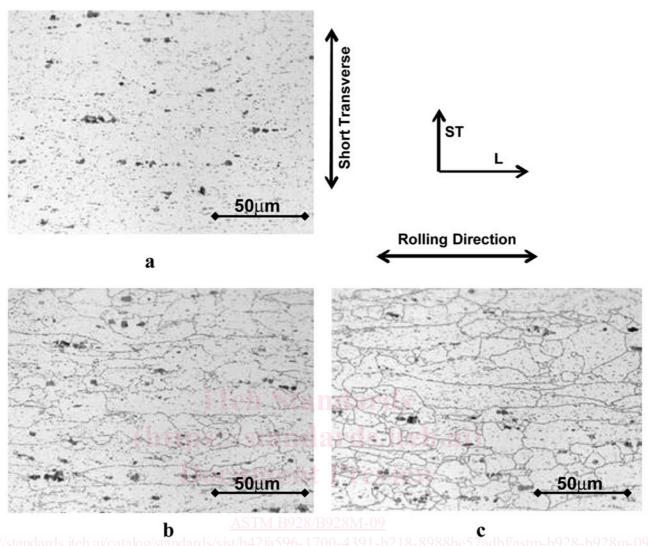
9. Exfoliation and Intergranular Corrosion Resistance

9.1 Only the Alloy-Tempers shown in Table 2 and Table 4 [Table 3 and Table 5] are manufactured and corrosion tested for intended use in marine hull construction or in marine applications where frequent or constant direct contact with seawater is expected. See Notes 3 and 4. (Warning—It is possible to meet the requirements of Test Method G66 (ASSET) and fail the requirements of Test Method G67 (NAMLT). Therefore both tests shall be performed for process qualification (see 9.4), for lot release, that is, in developing producer-established reference photomicrographs (see 9.5), and for surveillance (see 9.8).)

Note 3—Background Information—Aluminum-magnesium-alloy products that have a continuous or nearly continuous grain boundary precipitate are susceptible to intergranular forms of corrosion, (that is, IGC, SCC, or exfoliation corrosion). Examples of varying degrees of grain boundary precipitate continuity are shown in Figs. 1 and 2. The term "sensitization" is used to describe the development of this susceptible microstructure. The type of corrosion that occurs in a sensitized 5xxx alloy will depend primarily on the morphology of the grain structure and on the residual and applied stresses that are present. The extent of corrosion that will occur depends on the degree of continuity of the grain boundary precipitation and the corrosiveness of the environment. Both recrystallized and unrecrystallized 5xxx alloys that have been sensitized, are susceptible to intergranular corrosion, and when subjected to sustained tensile stress, may exhibit intergranular stress corrosion cracking. Unrecrystallized 5xxx alloys that have been sensitized are also susceptible to exfoliation corrosion.

Note 4—Alloys 5059, 5083, 5086, 5383, and 5456 should not be used for service, which provides prolonged exposure to temperatures exceeding 150°F [65°C] (whether continuous exposure or discontinuous exposure) because of the risk of sensitization and the resulting susceptibility to intergranular corrosion and stress corrosion cracking. Cold forming can also increases susceptibility to intergranular corrosion and stress corrosion cracking.

- 49.2 Exfoliation-Corrosion Resistance—The alloy-tempers listed in Table 2 and Table 4 [Table 3 and Table 5] shall be capable of exhibiting no evidence of exfoliation corrosion and a pitting rating of PB or better when subjected to the test described in Test Method G66 (ASSET).
- 9.3 Intergranular-Corrosion Resistance—The alloy-tempers listed in Table 2 and Table 4 [Table 3 and Table 5] shall be capable of exhibiting resistance to intergranular corrosion as indicated by an acceptable mass-loss when tested in accordance with Test Method G67 (NAMLT). Test Method G67 mass loss results shall be interpreted as defined in 9.3.1 through 9.3.4.
- 9.3.1 *Pass*—Samples with mass loss no greater than 100 mg/in.² [15 mg/cm²], shall be accepted.
- 9.3.2 *Fail*—Samples with mass loss greater than 160 mg/in.² [25 mg/cm²] and the lots they represent, shall be rejected.
- 9.3.3 *Questionable*—Samples with mass loss greater than 100 mg/in.² [15 mg/cm²] but less than 160 mg/in.² [25 mg/cm²] shall be deemed questionable and shall be subjected to metallographic examination (See 9.3.4).
- 9.3.4 Examination of Samples Deemed Questionable—A longitudinal face perpendicular to the rolled surface of Test Method G67 corroded test coupons testing "questionable," shall be prepared (see Fig. 3). The exposed "corroded" surface



Specimens prepared as per 9.6.1 (Phosphoric Acid etched). (This is as-produced material, not subjected to Test Method G67 testing.) Metallographic examination is to be conducted ×500 magnification.

Figure 1a has discontinuous grain boundary precipitation, typical of a mass-loss of less than 15 mg/cm² in Test Method G67.

Figure 1b has semi-continuous grain boundary precipitation and would likely fall in the mid-range, 15-25 mg/cm2 in Test Method G67.

Figure 1c has a continous network of grain boundary precipitation, typical of a mass loss greater than 25 mg/cm² in Test Method G67.

(Warning—These photomicrographs are examples of typical microstructures and due to variations in alloy, temper and process, they may or may not be similar to the microstructure of production sheet or plate. These photographs shall not be used in lieu of producer-established reference photographs for comparison with production material in surveillance or in determining process qualification or lot release.)

FIG. 1 Examples of Microstructures with Varied Degrees of Grain Boundary Beta-phase Continuity, for a Partially Recrystallized Grain Structure.

of this sample shall be examined metallographically in the as-polished condition to determine if the loss of mass was a result of intergranular attack or general corrosion and pitting attack (see examples shown in Fig. 4). When preparing the polished metallographic sample, a rough-grinding step that removes at least 0.02 in. [0.5 mm] of metal should precede the final polishing step. A magnification of ×250 is recommended.

9.3.4.1 *Pass*—Samples exhibiting general or pitting attack with no intergranular attack shall be accepted.

9.3.4.2 *Fail*—Samples exhibiting intergranular attack and the lots they represent, shall be rejected.

9.4 *Process Qualification* (see 9.1)—For material produced to this specification, the producer's production process shall be qualified prior to production to this specification, by sampling

and testing material to establish the relationship between microstructure and resistance to corrosion.

9.4.1 A reference photomicrograph, taken at $\times 500$ after 3 minutes etching in a phosphoric acid etch that is 40 parts by volume of reagent grade (85% concentration) phosphoric acid and 60 parts by volume distilled water at 95°F [35°C] (the etchant may be referred to as $\rm H_3PO_4$ (40+60) as defined by Practice E50), shall be established for each of the alloytempers and thickness ranges shown in Table 2 and Table 4 [Table 3 and Table 5], and shall be taken from a sample within that thickness range.

9.4.1.1 The reference photomicrographs shall be taken from samples (see 9.5 and 9.6 for sample location and preparation) which exhibit no evidence of exfoliation corrosion and a pitting