



Designation: B211M – 12

Standard Specification for Aluminum and Aluminum-Alloy Rolled or Cold-Finished Bar, Rod, and Wire (Metric)¹

This standard is issued under the fixed designation B211M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense.

1. Scope*

1.1 This specification covers rolled or cold-finished bar, rod, and wire in alloys (Note 1) and tempers as shown in Table 2.

NOTE 1—Throughout this specification use of the term *alloy* in the general sense includes aluminum as well as aluminum alloy.

NOTE 2—The term *cold finished* is used to indicate the type of surface finish, sharpness of angles, and dimensional tolerances produced by drawing through a die.

NOTE 3—See Specification B221M for aluminum and aluminum-alloy extruded bars, rods, wire, shapes, and tubes; and Specification B316/B316M for aluminum and aluminum-alloy rivet and cold-heading wire and rods.

1.2 Alloy and temper designations are in accordance with ANSI H35.1M. The equivalent Unified Numbering System alloy designations are those of Table 1 preceded by A9, for example, A91100 for aluminum 1100 in accordance with Practice E527.

1.3 This specification is the metric counterpart of Specification B211.

1.4 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.

2. Referenced Documents

2.1 The following documents of the issue in effect on date of material procurement form a part of this specification to the extent referenced herein.

2.2 ASTM Standards:²

B221M Specification for Aluminum and Aluminum-Alloy Extruded Bars, Rods, Wire, Profiles, and Tubes (Metric)

B316/B316M Specification for Aluminum and Aluminum-Alloy Rivet and Cold-Heading Wire and Rods

B557M Test Methods for Tension Testing Wrought and Cast

Aluminum- and Magnesium-Alloy Products (Metric)

B594 Practice for Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Aerospace Applications

B660 Practices for Packaging/Packing of Aluminum and Magnesium Products

B666/B666M Practice for Identification Marking of Aluminum and Magnesium Products

B881 Terminology Relating to Aluminum- and Magnesium-Alloy Products

B918 Practice for Heat Treatment of Wrought Aluminum Alloys

E29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications

E34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Alloys

E290 Test Methods for Bend Testing of Material for Ductility

E527 Practice for Numbering Metals and Alloys in the Unified Numbering System (UNS)

E607 Test Method for Atomic Emission Spectrometric Analysis Aluminum Alloys by the Point to Plane Technique Nitrogen Atmosphere³

E716 Practices for Sampling and Sample Preparation of Aluminum and Aluminum Alloys for Determination of Chemical Composition by Spectrochemical Analysis

E1004 Test Method for Determining Electrical Conductivity Using the Electromagnetic (Eddy-Current) Method

E1251 Test Method for Analysis of Aluminum and Aluminum Alloys by Spark Atomic Emission Spectrometry

G47 Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

G47 Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

G47 Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

2.3 *ANSI Standards:*

H35.1M Alloy and Temper Designation Systems for Aluminum⁴

H35.2M Dimensional Tolerances for Aluminum Mill Products⁴

¹ This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum Alloy Wrought Products.

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² For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For Annual Book of ASTM Standards volume information, refer to the standard's Document Summary page on the ASTM website.

³ Withdrawn. The last approved version of this historical standard is referenced on www.astm.org.

⁴ Available from American National Standards Institute (ANSI), 25 W. 43rd St., 4th Floor, New York, NY 10036, http://www.ansi.org.

*A Summary of Changes section appears at the end of this standard.

TABLE 1 Chemical Composition Limits^{A,B,C,D}

Alloy	Si	Fe	Cu	Mn	Mg	Cr	Ni	Zn	Ti	Bi	Pb	Sn	Other Elements ^E		Aluminum
													Each	Total ^F	
1100 ^G	0.95 Si + Fe	0.05–0.20	0.05	0.10	0.05	0.15	99.00 min ^H
2011	0.40	0.7	5.0–6.0	0.30	...	0.20–0.6	0.20–0.6	...	0.05	0.15	remainder
2014	0.50–1.2	0.7	3.9–5.0	0.40–1.2	0.20–0.8	0.10	...	0.25	0.15	0.15	0.05	0.15	remainder
2017	0.20–0.8	0.7	3.5–4.5	0.40–1.0	0.40–0.8	0.10	...	0.25	0.15	0.15	0.05	0.15	remainder
2024	0.50	0.50	3.8–4.9	0.30–0.9	1.2–1.8	0.10	...	0.25	0.15	0.15	0.05	0.15	remainder
2219	0.20	0.30	5.8–6.8	0.20–0.40	0.02	0.10	0.02–0.10	0.02–0.10	0.05 ^I	0.15 ^I	remainder
3003	0.6	0.7	0.05–0.20	1.0–1.5	0.10	0.05	0.15	remainder
4032	11.0–13.5	1.0	0.50–1.3	...	0.8–1.3	0.10	0.5–1.3	0.25	0.05	0.15	remainder
5052	0.25	0.40	0.10	0.10	2.2–2.8	0.15–0.35	...	0.10	0.05	0.15	remainder
5056	0.30	0.40	0.10	0.05–0.20	4.5–5.6	0.05–0.20	...	0.10	0.05	0.15	remainder
5154 ^G	0.25	0.40	0.10	0.10	3.1–3.9	0.15–0.35	...	0.20	0.20	0.20	0.05	0.15	remainder
6013	0.6–1.0	0.50	0.6–1.1	0.20–0.8	0.8–1.2	0.10	...	0.25	0.10	0.05	0.15	remainder
6020	0.40–0.9	0.50	0.30–0.9	0.35	0.6–1.2	0.15	...	0.20	0.15	...	0.05	0.9–1.5	0.05	0.15	remainder
6061	0.40–0.8	0.7	0.15–0.40	0.15	0.8–1.2	0.04–0.35	...	0.25	0.15	0.15	0.05	0.15	remainder
6110	0.7–1.5	0.8	0.20–0.7	0.20–0.7	0.50–1.1	0.04–0.25	...	0.30	0.15	0.15	0.05	0.15	remainder
6262	0.40–0.8	0.7	0.15–0.40	0.15	0.8–1.2	0.04–0.14	...	0.25	...	0.40–0.7	0.40–0.7	0.15	0.05	0.15	remainder
7075	0.40	0.50	1.2–2.0	0.30	2.1–2.9	0.18–0.28	...	5.1–6.1	0.20	0.05	0.15	remainder

^A Limits are in mass percent maximum unless otherwise shown.

^B Analysis shall be made for the elements for which limits are shown in this table.

^C For purposes of determining conformance to these limits, an observed value or a calculated value obtained from analysis shall be rounded to the nearest unit in the last right-hand place of figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E29.

^D In case of any discrepancy in the values listed in Table 3 when compared with those listed in the “Teal Sheets” (International Alloy Designations and Chemical Composition Limits for Wrought Aluminum and Wrought Aluminum Alloys), the composition limits registered with The Aluminum Association and published in the “Teal Sheets” shall be considered the controlling composition.

^E *Others* includes listed elements for which no specific limit is shown as well as unlisted metallic elements. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic *Others* elements. Should any analysis by the producer or the purchaser establish that an *Others* element exceeds the limit of *Each* or that the aggregate of several *Others* elements exceeds the limit of *Total*, the material shall be considered non-conforming.

^F Other elements—Total shall be the sum of unspecified metallic elements 0.010 % or more each, rounded to the second decimal before determining the sum.

^G Beryllium 0.0003 maximum for welding electrode and welding rod only.

^H The aluminum content is the difference between 100.00 % and the sum of all the other metallic elements and silicon present in amounts of 0.010 % or more each, rounded to the second decimal before determining the sum.

^I Vanadium 0.05–0.15 % zirconium 0.10–0.25 %. The total for other elements does not include vanadium and zirconium.

2.4 Federal Standard:

Fed. Std. No. 123 Marking for Shipment (Civil Agencies)⁵

2.5 Military Standard:

MIL-STD-129 Marking for Shipment and Storage⁵

2.6 Aerospace Material Specification:

AMS 2772 Heat Treatment of Aluminum Alloy Raw Materials⁶

2.7 The Aluminum Association:

International Alloy Designations and Chemical Composition Limits for Wrought Aluminum and Wrought Aluminum Alloys⁷

2.8 Other Standards:

CEN EN 14242 Aluminium and Aluminium Alloys—Chemical Analysis—Inductively Coupled Plasma Optical Emission Spectral Analysis⁸

3. Terminology

3.1 Definitions:

3.1.1 Refer to Terminology **B881** for definitions of product terms used in this specification.

3.1.2 *flatten and slit wire*—flatten wire which has been slit to obtain square edges.

3.2 Definitions of Terms Specific to This Standard:

3.2.1 *capable of*—The term *capable of* as used in this specification means that the test need not be performed by the producer of the material. However, should subsequent testing by the purchaser establish that the material does not meet these requirements, the material shall be subject to rejection.

4. Ordering Information

4.1 Orders for material to this specification shall include the following information:

4.1.1 This specification number,

4.1.2 Quantity in pieces or kilograms,

4.1.3 Alloy (Section 7),

4.1.4 Temper (Section 9),

4.1.5 *Product Form*—Rolled or cold-finished bar, rolled or cold-finished rod, or wire.

4.1.6 *Geometry and Dimensions*—Diameter for rounds; distance across flats for square-cornered squares, hexagons, or octagons; width and depth for square-cornered rectangles (orders for squares, hexagons, octagons, or rectangles with rounded corners usually require a drawing),

4.1.7 Length,

4.1.8 Tensile property limits and dimensional tolerances for sizes not covered in **Table 2** and in ANSI H35.2M, respectively.

⁵ Available from Standardization Documents Order Desk, Bldg. 4 Section D, 700 Robbins Ave., Philadelphia, PA 19111-5094, Attn: NPODS.

⁶ Available from the Society of Automotive Engineers (SAE), 400 Commonwealth Drive, Warrendale, PA 15096-0001.

⁷ Available from The Aluminum Association, Inc. 1525 Wilson Boulevard, Arlington, VA 22209, www.aluminum.org.

⁸ Available from European Committee for Standardization, Central Secretariat (CEN), rue de Stassart 36, B1050 Brussels, Belgium. http://www.cen.eu/eseach

TABLE 2 Mechanical Property Limits^{A,B}

Temper	Specified Diameter or Thickness, mm		Tensile Strength, MPa		Yield Strength ^C (0.2 % offset), MPa		Elongation, ^{C,D} min, %	
	over	through	min	max	min	max	in 50 mm	in 5 × diameter (5.65 √A)
Aluminum 1100								
O	...	3.20	75	105
	3.20	...	75	105	20	...	25	22
H12	...	10.00	95
H14	...	10.00	110
H16	...	10.00	130
H18	...	10.00	150
H112	all		75	...	20
F	all		^E	...	^E
Alloy 2011								
T3	3.20	40.00	310	...	260	...	10	9
	40.00	50.00	295	...	235	10
	50.00	90.00	290	...	205	12
T4 and T451 ^F	3.20	200.00	275	...	125	...	16	14
T6 and T651	10.00	160.00	370	...	275	...	10	9
T8	3.20	80.00	370	...	275	...	10	9
Alloy 2014 ^G								
O	...	3.20	...	240
	3.20	200.00	...	240	12	10
T4, T42 ^H , and T451 ^F	...	3.20	380
	3.20	200.00 ^I	380	...	220	...	16	14
T6, T62 ^H , and T651 ^F	...	3.20	450
	3.20	200.00 ^I	450	...	380	...	8	7
Alloy 2017 ^G								
O	...	3.20	...	240
	3.20	200.00	...	240	16	14
T4, T42 ^H , and T451 ^F	...	3.20	380
	3.20	200.00 ^{I,J}	380	...	220	...	12	10
Alloy 2024 ^G								
O	...	3.20	...	240
	3.20	200.00	...	240	16	14
T36	...	3.20	475
	3.20	10.00	475	...	360	...	10	...
T4 ^K	...	3.20	425
	3.20	12.50	425	...	310 ^K	...	10	...
	12.50	120.00 ^L	425	...	290	9
	120.00	160.00 ^M	425	...	275	9
	160.00	200.00 ^M	425	...	260	9
T42 ^H	...	3.20	400
	3.20	25.00	425	...	275	...	10	9
	25.00	160.00 ^L	425	...	275	9
T351 ^F	12.50	160.00 ^L	425	...	310	9
	160.00	200.00	425	...	310	8
T6	...	3.20	425
	3.20	160.00 ^L	425	...	345	...	5	4
T62 ^H	...	3.20	415
	3.20	160.00 ^L	415	...	315	...	5	4
T851 ^F	12.50	160.00 ^L	455	...	400	4
Alloy 2219								
T851 ^F	12.50	50.00	400	...	275	3
	50.00	100.00	395	...	270	3
Alloy 3003								
O	...	3.20	95	130
	3.20	...	95	130	35	...	25	22
H12	...	10.00	115
H14	...	10.00	140
H16	...	10.00	165
H18	...	10.00	185
H112	all		95	...	35
F	all		^E	...	^E
Alloy 4032								
T86	10.00	20.00	350	...	315	...	4	3

TABLE 2 *Continued*

Temper	Specified Diameter or Thickness, mm		Tensile Strength, MPa		Yield Strength ^C (0.2 % offset), MPa		Elongation, ^{C,D} min, %	
	over	through	min	max	min	max	in 50 mm	in 5 × diameter (5.65 √A)
Alloy 5052								
O	...	3.20	170	220
	3.20	...	170	220	65	...	25	22
H32	...	3.20	215
	3.20	10.00	215	...	160
H34	...	3.20	235
	3.20	10.00	235	...	180
H36	...	3.20	255
	3.20	10.00	255	...	200
H38	...	10.00	270
F	all		^E	...	^E
Alloy 5056								
O	...	3.20	...	320
	3.20	320	20	18
H111	...	10.00	300
H12	...	10.00	315
H32	...	10.00	300
H14	...	10.00	360
H34	...	10.00	345
H18	...	10.00	400
H38	...	10.00	380
H192	...	10.00	415
H392	...	10.00	400
Alloy 5154								
O	...	3.20	205	285
	3.20	...	205	285	75	...	25	22
H32	...	10.00	250
H34	...	10.00	270
H36	...	10.00	290
H38	...	10.00	310
H112	all		205	...	75
Alloy 6013								
T651	12.50	100.00	385	...	360	6
T8	20.00	40.00	400	...	385	7
	40.00	140.00	395	...	380	6
Alloy 6020								
T8	5.00	10.00	295	...	275	...	12	...
	10.00	50.00	290	...	270	...	12	10
	50.00	80.00	270	...	250	10
Alloy 6061 ^a								
O	...	3.20	...	155
	3.20	200.00	...	155	18	16
T4 and T451 ^F	...	3.20	205
	3.20	200.00 ^J	205	...	110	...	18	16
T42 ^H	3.20	200.00 ^J	205	...	95	...	18	16
T6, T62 ^H , and T651 ^F	...	3.20	290
	3.20	200.00 ^J	290	...	240	...	10	9
T89 and T94	...	10.00	370	...	325
Alloy 6110								
T9	...	10.00	450	...	435	...	2	...
Alloy 6262								
T6 and T651 ^F	3.20	200.00 ^L	290	...	240	...	10	9
T8	20.00	50.00	310	...	295	...	12	10
T9	3.20	50.00	360	...	330	...	5	4
	50.00	80.00	345	...	315	4
Alloy 7075 ^a								
O	...	3.20	...	275
	3.20	200.00	...	275	10	9
T6, T62 ^H	...	3.20	530	...	455
	3.20	100.00 ^N	530	...	455	...	7	6
T651 ^F	...	3.20	530	...	455

TABLE 2 *Continued*

Temper	Specified Diameter or Thickness, mm		Tensile Strength, MPa		Yield Strength ^C (0.2 % offset), MPa		Elongation, ^{C,D} min, %		
	over	through	min	max	min	max	in 50 mm	in 5 × diameter (5.65 √A)	
T73 and T7351 ^F	3.20	100.00 ^N	530	...	455	...	7	...	
	100.00	160.00	515	...	440	...	7	...	
	160.00	200.00	505	...	425	...	7	...	
	...	3.20	470	
	3.20	100.00	470	...	425	...	10	9	
	100.00	120.00	455	...	380	...	8	9	
	120.00	160.00	440	...	360	7	
Temper	Specified Diameter or Thickness, mm		Bend Diameter Factor, N						
	over	through							
Alloy 2017									
T4, T42, and T451	...	3.20							3 ^O
	3.20	200.00 ^J							6 ^O
Alloy 2024									
0	...	3.20							1
T351, T4, T42	...	3.20							3
	3.20	160.00							6
Alloy 3003									
0	all								0
H12	...	10.00							2
H14	...	10.00							2
H16	...	10.00							8

^A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest 1 MPa and each value for elongation to the nearest 0.5 %, both in accordance with the rounding-off method of Practice E29.

^B The basis for establishment of tensile property limits is shown in Annex A1.

^C The measurement of yield strength and elongation is not required for wire up through 3.20 mm in thickness or diameter.

^D Elongations in 50 mm apply to rectangular bar up through 12.5 mm thickness from which a standard rectangular tension test specimen is machined. The 5× diameter (5.65 √A) requirements, where D and A are diameter and cross-sectional area of the specimen, respectively, apply to round specimens tested in full-section or to standard or proportional, round-machined, tension test specimens.

^E There are no tensile requirements for material in the F temper but it usually can be expected that material 40 mm or less in thickness or diameter (except sections over 100 mm in width) will have a strength about equivalent to the H14 or H34 temper. As size increases the strength decreases to nearly that of the O temper.

^F For stress-relieved tempers, characteristics and properties other than those specified may differ somewhat from the corresponding characteristics and properties of material in the basic tempers.

^G Also available in the F temper for which no properties are specified and no tension tests are performed but for which tests are performed for confirmation of heat-treat response as required by Section 10.

^H Material in the T42 or T62 tempers is not available from the materials producers. These properties can usually be obtained by the user when material is properly solution heat treated or solution and precipitation heat treated from O or F temper. These properties also apply to samples of material in the O or F temper that are solution heat treated or solution and precipitation heat treated by the producer to determine that the material will respond to proper heat treatment. Properties attained by the user, however, may be lower than those listed if the material has been formed or otherwise cold or hot worked, particularly in the O temper, prior to solution heat treatment.

^I For rounds, maximum diameter is 200 mm; for square, rectangular, hexagonal, or octagonal bar, maximum thickness is 100 mm and maximum cross-sectional area is 23 000 mm².

^J For bar, maximum cross-sectional area is 32 000 mm².

^K Minimum yield strength for 2024-T4 wire and rod over 3.20 mm in thickness or diameter, produced in coil form for both straight length and coiled products, is 275 MPa.

^L Properties listed for this size increment are applicable to rod with a maximum diameter of 160 mm and to square, rectangular, hexagonal, or octagonal bar having a maximum thickness of 100 mm and maximum cross-sectional area of 23 000 mm².

^M Properties listed for this size increment are listed for rod only.

^N For rounds, maximum diameter is 100 mm; for square, hexagonal, or octagonal bar, maximum thickness is 90 mm; for rectangular bar, maximum thickness is 80 mm, with corresponding maximum width of 150 mm; for rectangular bar less than 80 mm in thickness, maximum width is 250 mm.

^O Bend diameter factor values stated for this full size increment apply to T4 product only. Values listed also apply to T451 produce in the 12.2-200 mm size range.

4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:

4.2.1 Whether heat treatment in accordance with Practice B918 is required (8.2),

4.2.2 Whether 7075-O material is required to develop requirements for T73 temper (see 10.1.2),

4.2.3 Whether bend testing is required for 2017, 2024, or 3003 (Section 12),

4.2.4 When specified finish of bar and rod is not required (Section 15),

4.2.5 Whether marking for identification is required (Section 16),

4.2.6 Whether ultrasonic inspection is required (Section 17, Table 3),

4.2.7 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (Section 19),

4.2.8 Whether certification is required (Section 21), and

4.2.9 Whether Practices B660 applies, and if so, the levels of preservation, packaging, and packing required (Section 22).

TABLE 3 Ultrasonic Discontinuity Limits for Rolled or Cold-Finished Bar^A

Alloys	Size		Maximum Mass per Piece, kg	Discontinuity Class ^B
	Thickness, mm			
	over	through		
2014, 9221 } 2024, 7075 }	12.50	35.00	300	B
	35.00	80.00	300	A
	80.00	155.00	500	B

^A Discontinuities in excess of those listed in this table shall be allowed if it is established that they will be removed by machining or that they are in noncritical areas.

^B The discontinuity class limits are defined in Section 11 of Practice B594.

5. Manufacture

5.1 The products covered by this specification shall be produced either by hot extruding and cold finishing or by hot rolling with or without cold finishing, at the option of the producer.

6. Quality Assurance

6.1 *Responsibility for Inspection and Tests*—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use their own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set forth in this specification where such inspections are deemed necessary to assure that material conforms to prescribed requirements.

6.2 *Lot Definition*—An inspection lot shall be defined as follows:

6.2.1 For heat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions traceable to a heat-treat lot or lots, and subjected to inspection at one time.

6.2.2 For nonheat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions subjected to inspection at one time.

7. Chemical Composition

7.1 *Limits*—The bars, rods, and wire shall conform to the chemical composition limits specified in Table 1. Conformance shall be determined by the producer by taking samples in accordance with E716 when the ingots are poured and analyzing those samples in accordance with E607, E1251, E34 or EN 14242. At least one sample shall be taken for each group of ingots poured simultaneously from the same source of molten metal. If the producer has determined the chemical composition of the material during pouring of the ingots, they shall not be required to sample and analyze the product.

NOTE 4—It is standard practice in the United States aluminum industry to determine conformance to the chemical composition limits prior to further processing of ingots into wrought products. Due to the continuous nature of the process, it is not practical to keep a specific ingot analysis identified with a specific quantity of finished material.

7.2 If it becomes necessary to analyze bars, rod or wire for conformance to chemical composition limits, the method used to sample for the determination of chemical composition shall be by agreement between the producer and the purchaser. Analysis shall be performed in accordance with E716, E607, E1251, E34 or EN 14242 (ICP method). The number of samples taken for determination of chemical composition shall be as follows:

7.2.1 When samples are taken from the finished or semifinished product, a sample shall be taken to represent each 4000 lb, or fraction thereof, in the lot, except that no more than one sample shall be required per piece.

7.3 Other methods of analysis or in the case of dispute may be by agreement between the producer and the purchaser.

8. Heat Treatment

8.1 Unless otherwise specified in 8.2, producer or supplier heat treatment for the applicable tempers designated in Table 2 shall be in accordance with AMS 2772.

8.2 When specified, heat treatment of applicable tempers in Table 2 shall be in accordance with Practice B918.

9. Tensile Properties of Material as Supplied

9.1 *Limits*—The bar, rod, and wire shall conform to the tensile requirements in Table 2.

9.2 *Number of Specimens*:

9.2.1 For material having a nominal mass up through 1.7 kg/linear m, one tension test specimen shall be taken for each 500 kg or fraction thereof in the lot. Only one specimen shall be taken from any one piece when more than one piece is available.

9.2.2 For material having a nominal mass over 1.7 kg/linear m, one tension test specimen shall be taken for each 300 m or fraction thereof in the lot. Only one specimen shall be taken from any one piece when more than one piece is available.

9.3 *Test Specimens*—Geometry of test specimens and the location in the product from which they are taken shall be as specified in Test Methods B557M.

9.4 *Test Methods*—The tension tests shall be made in accordance with Test Method B557M.

10. Producer Confirmation of Heat-Treat Response

10.1 In addition to the requirements of 9.1, material in Alloys 2014, 2017, 2024, and 6061 produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for T42 temper material. The heat-treated samples may be tested prior to 4 days natural aging but if they fail to conform to the T42 temper properties, the tests may be repeated after completion of 4 days natural aging without prejudice.

10.1.1 Alloy 7075 material produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and precipitation heat treatment, conform to the properties specified in Table 2 for T62 temper material.

10.1.2 When specified, 7075-O material (within the size limits specified in Table 2) shall, after proper solution and