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Space systems — Ground test for the separation between a launch vehicle and a spacecraft — Requirements for combined separation tests, horizontal separation tests and individual falling separation tests

Systèmes spatiaux — Essais au sol pour la séparation entre un lanceur et un engin spatial — Exigences relatives aux essais de séparation combinée, aux essais de séparation horizontale et aux essais de séparation par chute individuelle

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Foreword

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The procedures used to develop this document and those intended for its further maintenance are described in the ISO/IEC Directives, Part 1. In particular, the different approval criteria needed for the different types of ISO document should be noted. This document was drafted in accordance with the editorial rules of the ISO/IEC Directives, Part 2 (see www.iso.org/directives).

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This document was prepared by Technical Committee ISO/TC 20, Aircraft and space vehicles, Subcommittee SC 14, Space systems and operations.

Any feedback or questions on this document should be directed to the user's national standards body. A complete listing of these bodies can be found at www.iso.org/members.html.

Introduction

The separation between the launch vehicle and the spacecraft on orbit is a critical activity for a successful orbit injection. Any failure during the separation can cause serious consequences. A separation test on ground between the spacecraft and the launch vehicle is an effective way to simulate and test the characteristics of the separation, to reveal the risk of failure of the separation system between the launch vehicle and the spacecraft, and to improve the separation function during the separation on orbit. This document provides three typical separation test methods, which are the combined separation test, the horizontal separation test and the individual falling separation test, for different application scenarios.

The overview and interdependencies of test methods are illustrated in <u>Annex A</u>. For the testing of separation movements and separation dynamics between launch vehicle and spacecraft with zero-gravity effect or no acceleration conditions at the time of separation, the separation test may take the form of combined separation test or horizontal separation test as illustrated in <u>Annex B</u> and <u>Annex D</u> respectively. For the testing of separation with purposes of fit-check and shock-response-level confirmation, the test may take the form of individual falling separation test as illustrated in <u>Annex C</u>.

Apart from these three separation test methods, there are many other methods of verifying separation systems and processes, such as modelling the systems or processes, comparing similarity with existing systems or processes, which may apply to different scenarios with consistency. This document does not deny the use of these useful methods in the simulating and testing of the characteristics of the separation.

The technical requirements in this document may be tailored to meet the objectives of tests as the separation objectives can be different for different projects or for different development stages.

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Space systems — Ground test for the separation between a launch vehicle and a spacecraft — Requirements for combined separation tests, horizontal separation tests and individual falling separation tests

1 Scope

This document provides the test requirements of three typical separation test methods, which are the combined separation test, the horizontal separation test, the individual falling separation test, for the separation between the launch vehicle (LV) and the spacecraft (or between stages of a prototype LV model). It also provides the requirements for the separation test unit, test data, test timing and sequence, pre-test simulation, test environment, exception handling, test results assessment, test documentation, test facilities, test installation, preliminary adjustment of the test setup, measurement and data acquisition.

This document is applicable to test providers and interested parties to implement the separation test between the launch vehicle and the spacecraft.

2 Normative references ANDARD PREVIEW

The following documents are referred to in the text in such a way that some or all of their content constitutes requirements of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

ISO 14620-1:2018, Space systems — Safety requirements — Part 1: System safety

ISO 15864:2021, Space systems — General test methods for spacecraft, subsystems and units

ISO 22137, Space systems — Program management — Test reviews

3 Terms and definitions

For the purposes of this document, the following terms and definitions apply.

ISO and IEC maintain terminology databases for use in standardization at the following addresses:

- ISO Online browsing platform: available at https://www.iso.org/obp
- IEC Electropedia: available at https://www.electropedia.org/

3.1

combined separation test

test in which a combination of spacecraft parts and launch vehicle parts drops simultaneously from the hanging point

Note 1 to entry: The test articles finish the separation movements before being captured during the separation test.

3.2

fit check

check to verity the matching consistencies of the electrical interface and the mechanical interface between the spacecraft and the launch vehicle to some degree

Note 1 to entry: This check usually consists of an electrical interface check and a mechanical interface check between the spacecraft and the launch vehicle before the separation test.

3.3

horizontal separation test

test in which a combination of spacecraft parts and launch vehicle parts is separated horizontally

3.4

individual falling separation test

test in which a combination of spacecraft parts and launch vehicle parts is separated vertically and either spacecraft parts or launch vehicle parts fall down from the combination

3.5

launch vehicle stages separation system

assembly of functionally related subsystems and/or units for the separation between launch vehicle stages

3.6

mating and unlocking device

device that connects the spacecraft and the launch vehicle together before separation and disconnects the linkage between the spacecraft and the launch vehicle at the time of separation

3.7

separation impulse device

device that provides the energy to separate the spacecraft from the launch vehicle

3.8 ISO 5879:202

separation system between launch vehicle and spacecraft 45-10cf-411d-9ecb-321f8b1674bb/iso-assembly of functionally related subsystems and/or units for the separation between launch vehicle and spacecraft

3.9

test article

spacecraft, subsystem or unit on which a test is conducted

Note 1 to entry: For this document the definition is extended to address also launch vehicle system, subsystem or unit.

3.10

separation test unit

assembly of test articles which undergoes separation test

4 Symbols and abbreviated terms

CoG centre of gravity

LV launch vehicle

SC spacecraft

SDOF single degree of freedom

SRS shock response spectrum

 ζ damping ratio

5 General requirements

5.1 Purposes of separation testing

5.1.1 General

The separation testing is conducted for two types of purposes, which are verification purposes and diagnostic purposes. A verification purpose is to verify conformity of characteristics of a separation system between the LV and the SC (or between stages of a prototype LV model) with the specified requirements. A diagnostic purpose is testing for physical system or operational design analysis.

5.1.2 Verification purposes

Verification purposes typically include:

- a) to verify the functionality and characteristics of the separation system design and compatibility of interfaces, including mechanical and electrical interfaces (e.g. fit check), between the LV and the SC, as specified in ISO 14303:2002, Clauses 4 and 5;
- b) to measure the parameters of separation relative movement, such as separation velocity, acceleration, angular motion, separation displacement;
- c) to measure the separation clearances between the separating hardware;
- d) to measure the separation characteristic action timing during the separation process, such as time to commence, time to clear.

5.1.3 Diagnostic purposes

Diagnostic purposes typically include: ISO 5879:202

- a) to check and revise the separation analysis computational model and basic assumptions used in the separation analysis;
- b) to measure the shock levels on the points that are relevant to the launch vehicle side or to the spacecraft side, and to check if the spacecraft is able to withstand the shock environment;
- c) to test the separation function of the device (e.g. mating and unlocking device) designed for separation of structural elements;
- d) to test the distancing function provided by the separation impulse device;
- e) to test purposes demanded by the test requirements, such as flexible-body distortion loads, which are relevant to the launch vehicle side or to the spacecraft side during the separation process.

5.1.4 Tailoring guide

The purposes of separation testing in 5.1.2 and 5.1.3 may be tailored as the separation purposes can be different for different projects or for different development stages.

5.2 General separation test process

5.2.1 General

The general separation test process flow is shown in Figure 1. The test process typically includes:

- a) test documentation preparation and pre-test simulation;
- b) test implementation;

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- c) test result data collection;
- d) development and issuing of a test results assessment report.

5.2.2 Test documentation preparation and pre-test simulation

The test documents, which include documents of test plan, test specification, test procedure as defined in ISO 15864:2021, 4.9, shall be prepared.

The test requirements for test articles, test facilities, test installation, preliminary adjustment of test setup, measurement and data acquisition, as required in a separation test, shall be defined in the test specification documents.

The documents of test specification and test procedure should be reviewed by the relevant sides to make sure the documents conform to the test requirements. The test documentation review in the test process flow shall be in line with the requirements of ISO 22137.

Pre-test simulation, which is specified in 5.6, shall meet the requirements of the test specification documents and support the making of the test procedure documentation.

5.2.3 Test implementation

Test implementation typically includes the following procedures: test preparation, test execution, follow-up activities after test, as illustrated in <u>Annex E</u> for a separation test.

5.2.4 Test results data collection

The test results data collection shall meet both the test data requirements, which are specified in <u>5.4</u>, and the measurement and data acquisition requirements, which are specified in <u>Clause 9</u>.

5.2.5 Test results evaluation

The test results shall be evaluated to assess whether the test results meet the test objectives as specified in 5.9.

If the test results meet the test objectives, the test process may step into "Issue the test report documentation" and end of the test process as shown in Figure 1.

If the test results do not meet the test objectives, a selection can be made between three ways to react corresponding to the failures as shown in Figure 1.

- a) If the failure is due to systems or interfaces design problems, the test should move to the step of "redesign of the relevant systems or interfaces between the LV and SC", then move to the step of "test plan", and proceed with the whole test process.
- b) If the failure is due to test procedure problems, the test should move to the step of "test procedure documentation review" to figure out whether the test procedure documentation meet the test requirements given in the test specification documentation, then move to next step as shown in Figure 1.
- c) If the failure is due to test specification problems, the test should move to the step of "test specification review" to figure out whether the test specification documentation meet the test requirements given in the test plan documentation, then move to next step as shown in Figure 1.

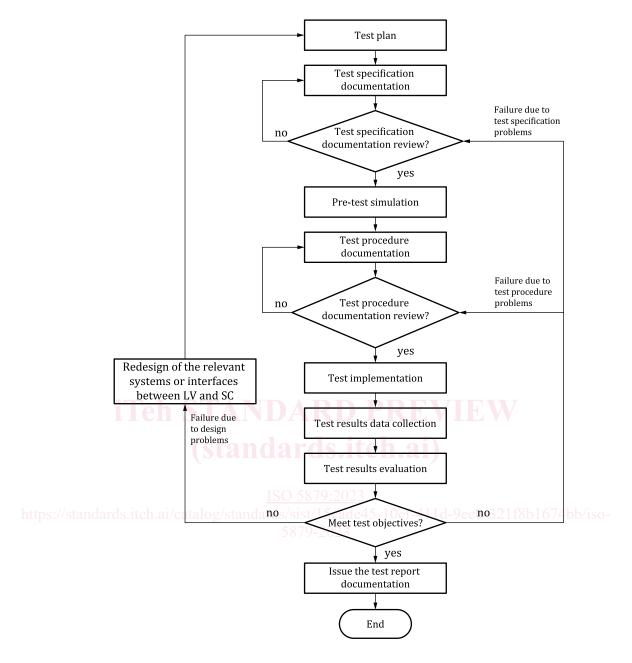


Figure 1 — General separation test process

5.3 Test articles under separation test

5.3.1 Specification of separation test unit

5.3.1.1 Acceptance test

The separation test unit for an acceptance test comprises the following test articles: a flight model or proto-flight model of a separation system between launch vehicle and spacecraft, a flight model or proto-flight model of a certain part of a spacecraft and a flight model or proto-flight model of a certain part of a launch vehicle.

5.3.1.2 Development test

In a development test between the SC and the LV, the separation test unit comprises the following test articles: a prototype model of a separation system between launch vehicle and spacecraft, a mock-up of a certain part of spacecraft and a mock-up of a certain part of a launch vehicle.

In a development test between stages of a prototype LV model, the separation test unit comprises the following test articles: the mock-ups of two connecting launch vehicle stages and a prototype model of the corresponding launch vehicle stages separation system.

5.3.2 Requirements for test articles

5.3.2.1 Technical characteristics

Technical characteristics of the test articles shall conform to the test specification documentation.

5.3.2.2 Interfaces

The mechanical and electrical interfaces at the separation surfaces of the test articles shall be identical to those of the prototype model or the flight model, depending on different test purposes as defined in 5.3.1.

5.3.2.3 Substitutes

Some parts of test articles may be substituted by the mock-up parts when these mock-up parts do not interfere with the purposes of the separation test.

5.3.2.4 Mass and stiffness

The mass and stiffness characteristics of the separation parts of test articles shall be identical to those of the prototype model or the flight model, depending on different test purposes, when measuring the separation motion parameters of the test.

5.3.2.5 Specification of difference between the SC/LV mock-up model and the flight model

The SC/LV mock-up model should be the structurally and/or physically similar items presenting a simplified reproduction of a test object of SC/LV or its part intended for test.

The SC/LV flight model or proto-flight model is dedicated to be launched and operated in orbit and should be subjected to acceptance testing.

While it is not always feasible to conduct full-scale mock-up tests, a reduced mock-up may be used in the separation test where relevant technical characteristics of the reduced mock-up model should be designed to be identical to those of the prototype model or the flight model.

5.3.2.6 Markings

There shall be obvious marks on the test articles for the high-speed video shooting.

5.3.3 Technical safety requirements for test articles

5.3.3.1 Confirmation

The test article's technical safety status shall be confirmed before test.

5.3.3.2 Handling

The operators shall follow the relevant technical safety rules under the consideration of the requirements of ISO 14620-1:2018, 8.3 in planning and executing the handling of the test articles, such as hoisting, placement, transporting, installation.

5.3.3.3 Grounding

The test articles shall be electrically grounded. IEC 62305-3 provides information on the ground resistance.

5.3.3.4 Safety protection

There shall be safety protection measures to protect the test articles from damaging both during and after the process of the test.

5.4 Data requirements

5.4.1 Test data

Depending on the purpose of the test campaign, separation movement data and/or separation shock response data, and/or the dedicated data by test requirements, shall be acquired.

5.4.2 Separation movement data

Separation movement data shall include translational acceleration, separation velocity, separation displacement, angular velocity, attitude angle of the LV test articles and the SC test articles.

Separation movement data may be measured directly or obtained by processing the measured data.

5.4.3 Separation shock response data sist/15eade45-10cf-411d-9ecb-321f8b1674bb/iso-

Separation shock response data shall include shock acceleration time history data and shock response spectrum data.

5.4.4 The dedicated data by test requirements

The dedicated data to be acquired, such as flexible-body distortion loads, shall be specified by the test requirements, as the data can be different for different types of separation test.

5.5 Test timing and sequence

Timing and sequence shall be synchronized for every system involved in the test. Synchronization accuracy shall be suitable to meet the test requirements.

5.6 Requirements for pre-test simulation

5.6.1 Purposes of pre-test simulation

Pre-test simulation shall:

- a) evaluate kinematics of the test articles before the separation test;
- b) determine the parameters of separation movement sensors, such as the measurements ranges, accuracy, and the placement of the separation movement sensors;
- c) determine the installation positions of high-speed cameras to get the most appropriate shooting angle of field of view;