FINAL DRAFT INTERNATIONAL STANDARD

ISO/FDIS 17546

ISO/FDIS 17546:2023

ISO-TC 20/SC 14/WG 1

Date: 2023-09-09

Secretariat: ANSI/AIAA

Date: 2023-11-01

Space systems — Lithium -ion battery for space vehicles — Design and verification requirements

<u>Systèmes spatiaux — Batteries à ions lithium pour véhicules spatiaux — Exigences de vérification et de conception</u>

(https://standards.iteh.ai)
Document Preview

ISO/FDIS 17546

https://standards.iteh.ai/catalog/standards/sist/7a8f6d0f-5341-4acc-aada-60e14409f32c/iso-fdis-17546

FDIS stage

© ISO 20222023

All rights reserved. Unless otherwise specified, or required in the context of its implementation, no part of this publication may be reproduced or utilized otherwise in any form or by any means, electronic or mechanical, including photocopying, or posting on the internet or an intranet, without prior written permission. Permission can be requested from either ISO at the address below or ISO's member body in the country of the requester.

ISO copyright office CP 401 • Ch. de Blandonnet 8 CH-1214 Vernier, Geneva Phone: + 41 22 749 01 11 EmailE-mail: copyright@iso.org

Website: www.iso.org

Published in Switzerland

iTeh Standards (https://standards.iteh.ai) Document Preview

ISO/FDIS 17546

https://standards.iteh.ai/catalog/standards/sist/7a8f6d0f-5341-4acc-aada-60e14409f32c/iso-fdis-17546

Contents

Forew	ord	viii
Introd	uction	ix
1	Scope	1
2	Normative references	1
3	Terms and definitions	1
4	Symbols and abbreviated terms	6
5	Cell	7
5.1	Performance	7
5.1.1	General	7
5.1.2	Test requirements	7
5.1.3	Test data trending [4]	8
5.1.4	Cell qualification test	9
5.1.5	Leakage (hermetic) test	9
5.1.6	Safety tests	9
5.1.7	Thermal/thermal vacuum test [5]	
5.1.8	Mechanical environmental test	9
5.1.9	Radiation test	
5.1.10	Life cycle testLife cycle test	10
5.1.11	Models for analysis	11
5.2	Safety	
5.2.1	General ISO/FDIS 17546	11
5.2.2	Hazard description Standards/sist/7a8f6d0f-5341-4acc-aada-60e14409f32e/iso-fdis	-17546 111
5.2.3	Protective devices as a hazard control	12
5.2.4	Safety testing	13
5.2.5	Important test considerations	13
5.2.6	Optional test	14
5.3	Logistics	15
5.3.1	General	15
5.3.2	Cell manufacturing, storage and testing	15
5.3.3	Safety measure for handling	16
5.3.4	Cell transportation	17
5.4	COTS cells for space use	17
5.4.1	General	17
5.4.2	Safety requirements	17

5.4.3	Lot integrity assessment	17
5.4.4	Mission conformance test	17
5.4.5	Charge and discharge test	18
5.4.6	Thermal vacuum test	18
5.4.7	Mechanical environment test	18
5.4.8	Cycle life test	18
5.4.9	Storage life test	18
6	Battery	18
6.1	Performance	18
6.1.1	General	18
6.1.2	C/n charge or discharge current (c-rate) ^{[4][5][6]}	19
6.1.3	Cut-off voltage	19
6.1.4	Cycle	19
6.1.5	Depth of discharge (DOD)[4][5][6][13]	19
6.1.6	End of charge voltage	19
6.1.7	Energy	19
6.1.8	Energy density	20
6.1.9	Energy reserve [4][5]Energy reserve [4][5]	20
6.1.10	Fully charged [9][13]	20
6.1.11	Nameplate capacity	20
6.1.12	Nominal capacity Document Preview	20
6.1.13	Nameplate energy [5][12]	20
6.1.14	Nominal voltage [9]	20
6.1.15	State of charge	
6.1.16	Standard method for capacity measurement	21
6.1.17	Battery internal resistance (ohmic)	21
6.1.18	Battery impedance	22
6.1.19	Life test demonstration	22
6.1.20	For GEO simulated	22
6.1.21	For LEO simulated	23
6.1.22	For launch vehicle: simulate ground storage and usage at launch phase	23
6.1.23	Battery general requirements [5][11]	23
6.1.24	Electrical design	23
6.1.25	Thermal design	23
6.1.26	Mechanical design	24
6.1.27	Cell-to-cell balancing mechanisms	24

iv

6.1.28	Marking	24
6.1.29	Cell matching	24
6.1.30	Polarization testing (optional)	24
6.1.31	Self-discharge rate test	24
6.1.32	Tailoring screening tests	25
6.1.33	Cell matching criteria	25
6.1.34	Contamination control	25
6.1.35	Test data trending [4]	26
6.1.36	Flight verification acceptance testing [5]	26
6.1.37	Assurance of the life estimation [8]	26
6.1.38	Parameter measurement tolerances	26
6.1.39	Battery testing [5][15]	26
6.1.40	Development testing [4][15]	27
6.1.41	Charge control testing	27
6.1.42	Thermal control testing	27
6.1.43	Mechanical test	28
6.1.44	Qualification test [4]	28
6.1.45	Qualification test levels and duration	28
6.1.46	In-process inspections and tests	28
6.1.47	Data collection and acquisition rates	28
6.2 S	afety	29
6.2.1	General	29
6.2.2	Definitions of dangerous phenomenon	30
6.2.3	Technical requirement	
6.2.4	Fault tolerance	31
6.2.5	Hazard controls [5]	31
6.2.6	Over-current prevention	31
6.2.7	Over-voltage protection	31
6.2.8	Temperature/current management	32
6.2.9	Insulation and wiring	32
6.2.10	Positive protection against accidental shorting	32
6.2.11	Venting	32
6.2.12	Crew touch temperature requirements	33
6.2.13	Terminal contacts	33
6.2.14	Safety testing	33
6.2.15	Important test considerations	

© ISO 2022 – All rights reserved

© ISO 2023 – All rights reserved

6.2.16	Thermal runaway propagation	35
6.2.17	Special provision	35
6.2.18	B Description for necessary information for system safety review	35
6.3	Logistics	35
6.3.1	General	35
6.3.2	Manufacture/assembly storage and testing	36
6.3.3	Safety measure for handling	37
6.3.4	Transportation	37
7	Battery onboard space vehicle	38
7.1	Performance	38
7.1.1	General	38
7.1.2	Basic design	38
7.1.3	Electrical ground bonding	39
7.1.4	Temperature reference point of battery module	39
7.1.5	Preparation for handling, transportation	39
7.2	Safety	39
7.2.1	General	
7.2.2	Definitions of dangerous phenomenon	39
7.2.3	Technical requirement	40
7.3	Logistics	40
7.3.1	General Document Preview	40
7.3.2	Safety measure for handling	40
7.3.3	Integration to the space vehicle	40
7.3.4	Battery maintenance on the space vehicle	41
7.3.5	Battery transportation equipped in space vehicle	41
8	Launch site	41
8.1	General	41
8.2	Performance	42
8.3	Safety	42
8.4	Logistics	42
8.4.1	General	42
8.4.2	Safety measure for handling	43
8.4.3	Preparation for transportation	43
8.4.4	SOC level for transportation	43
8.4.5	Container for transportation	43
8.4.6	Battery testing (health checking after transportation)	43

8.4.7	Inspection	
8.4.8	State of health verification	44
8.4.9	Battery storage at launch site	44
8.4.10	Self-discharge rate	44
8.4.11	Protection under integration	45
8.4.12	Handling plate	45
8.4.13	Electrical connection	45
8.4.14	Electrical checkout	45
8.4.15	Battery monitoring	
8.4.16	Battery monitoring preceding launch	45
9	Mission in orbit and end of life	45
Annex A	A (informative) Parameter measurement tolerances	47
Annex l	B (informative) Example of cell qualification test	48
Annex (C (informative) Hazard identification method	51
Annex l	D (normative) Safety measure for handling	54
	E (informative) Transportation	
Annex l	F (informative) Lot assessment of COTS cells	62
Bibliog	raphystanualus	65
50		

ISO/FDIS 17546

https://standards.iteh.ai/catalog/standards/sist/7a8f6d0f-5341-4acc-aada-60e14409f32c/iso-fdis-17546

Foreword

ISO (the International Organization for Standardization) is a worldwide federation of national standards bodies (ISO member bodies). The work of preparing International Standards is normally carried out through ISO technical committees. Each member body interested in a subject for which a technical committee has been established has the right to be represented on that committee. International organizations, governmental and non-governmental, in liaison with ISO, also take part in the work. ISO collaborates closely with the International Electro-technical Electrotechnical Commission (IEC) on all matters of electro-technicalelectrotechnical standardization.

The procedures used to develop this document and those intended for its further maintenance are described in the ISO/IEC Directives, Part 1. In particular, the different approval criteria needed for the different types of ISO documents should be noted. This document was drafted in accordance with the editorial rules of the ISO/IEC Directives, Part 2 (see www.iso.org/directives).

Attention is drawn ISO draws attention to the possibility that some of the elements implementation of this document may be involve the subjectuse of (a) patent(s). ISO takes no position concerning the evidence, validity or applicability of any claimed patent rights in respect thereof. As of the date of publication of this document, ISO had not received notice of (a) patent(s) which may be required to implement this document. However, implementers are cautioned that this may not represent the latest information, which may be obtained from the patent database available at www.iso.org/patents... ISO shall not be held responsible for identifying any or all such patent rights. Details of any patent rights identified during the development of the document will be in the Introduction and/or on the ISO list of patent declarations received (see).

Any trade name used in this document is information given for the convenience of users and does not constitute an endorsement.

For an explanation of the voluntary nature of standards, the meaning of ISO specific terms and expressions related to conformity assessment, as well as information about ISO's adherence to the World Trade Organization (WTO) principles in the Technical Barriers to Trade (TBT), see www.iso.org/iso/foreword.html.

This document was prepared by Technical Committee ISO/TC 20, *Aircraft and space vehicles*, Subcommittee SC 14, *Space systems and operations*.

This second edition cancels and replaces the first edition (ISO 17546:2016), which has been technically revised.

The main changes are as follows:

— updated <u>5.4clause 5.4</u> and <u>Annex FAnnex F.</u>

Any feedback or questions on this document should be directed to the user's national standards body. A complete listing of these bodies can be found at www.iso.org/members.html.

viii

Viii

Introduction

This document has been developed for the purpose of establishing the standard to obtain sustainable development and to prevent incidents of lithium-ion battery for space vehicles.

Lithium-ion batteries belong to the category of rechargeable batteries which are based on electrochemical systems. The batteries generally consist of lithium metal oxide for positive electrodes and carbon for negative electrodes. Lithium element exists in an ionic or quasi-atomic form within a lattice structure of each electrode material.

For battery developers and spacecraft system architects, this document leads the way to assessing the whole life cycle from electrolyte filling to the end of the mission in space and to clarify what is considered in the battery design phase and the processes to reach the appropriate verification.

It is important to prevent lithium-ion batteries (LIB) for space vehicles from having performance defects in orbit and incidents through the life cycle. The total life cycle of lithium-ion batteries consists of material manufacturing to deorbit after mission completion as shown in Figure 1-Figure 1.

Since lithium-ion batteries start to deteriorate just after cell activation during cell manufacturing of stage 1, the service life starts at that point. And it continues through cell testing, cell transportation of stage 2, battery manufacturing/testing of stage 3, battery transportation of stage 4. Eventually, the service life is regarded defined as the duration until deorbit which corresponds to the end of life of stage 9.

Clauses in this document address "performance", "safety"," and "logistics" according to each stage of the life cycle, and each requirement belongs to the shelf life of lithium-ion batteries. The shelf life means from cell activation to launch and does not exceed the shelf life limit. A battery whose shelf life exceeds the shelf life limit is judged to be non-conforming even if it is not used because the battery capacity is insufficient for the mission completion. In other words, the shell life is the period until launch while maintaining the battery capacity to complete required missions.

- Performance Document Preview

Since LIB starts to degrade from activation, it is necessary to consider meeting the power requirement through the mission life; that is, to be unaffected from handling conditions (temperature) and usage conditions in orbit (temperature, cycle, current or power and depth of discharge). Also, the risk in orbit can be mitigated based on the life estimation; and with care unexpected degradation can be avoided throughout the whole life cycle.

— Safety

A complex risk assessment process that is easy to understand is established. The method was agreed internationally at ISO/IEC and is a traditional method for space use. LIB keeps some amount of the SOC to avoid significant capacity degradation, so that the specific consideration and care for handling are required because of potential hazard source. It is well known that LIB has specific risks with higher voltage when compared to other power sources and no saturation characteristic for over-charge. The important thing is that the process, which can result in a hazardous situation, does not always immediately result in an incident. Because of these risks, LIB is considered hazardous at all times. The risk assessment becomes very important to cover a variety of environments during the handling or use and history of stress.

— Logistics

The most important aspect of assuring battery safety and space quality in transportation is to perform a life cycle assessment of performance and safety from a broad perspective.

© ISO 2023 – All rights reserved

Critical damage includes, for example, the temperature history (especially high uncontrolled temperature outdoors), shock/vibration, and electrical shorts. Also, to reflect the results of handling or usage, the measurement should be done. All the personnel with responsibilities for the development, design, and handling should survey and estimate the influence of their assessment spontaneously to improve the sustainable development of the space component.

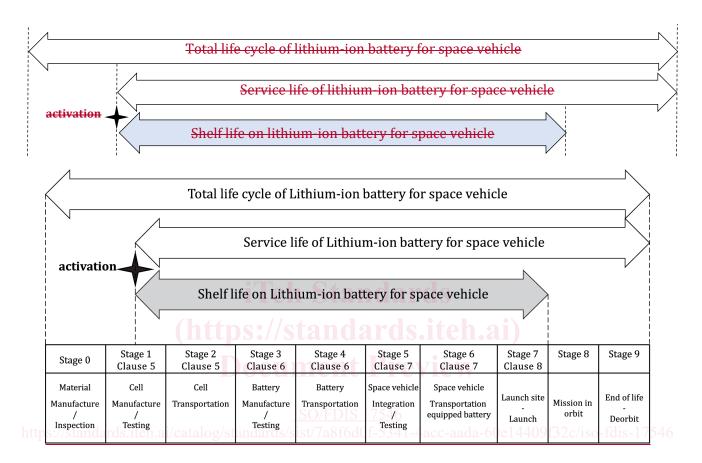


Figure 1 — Definition of Lifelife cycle stages of Lithiumlithium-ion battery for space vehicle

Space systems — Lithium -ion battery for space vehicle vehicles — Design and verification requirements

1 Scope

This document specifies design and minimum verification requirements for lithium-ion batteries from the perspectives of performance, safety and logistics.

This document is applicable to battery assemblies for space vehicles and component cells of batteries, which are critical devices to be harmonized with standards and regulations for other industries. In addition, this document is applicable to component cells which are not designed for space vehicles but can be used in space.

2 Normative references

The following documents are referred to in the text in such a way that some or all of their content constitutes requirements of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

ISO 24113:2019, Space systems — Space debris mitigation requirements

IEC 61960, Secondary cells and batteries containing alkaline or other non-acid electrolytes — Secondary Lithium cells and batteries for portable applications

MIL-STD-1686.9, ELECTROSTATIC DISCHARGE CONTROL PROGRAM FOR PROTECTION OF ELECTRICAL AND ELECTRONIC PARTS, ASSEMBLIES AND EQUIPMENT (EXCLUDING ELECTRICALLY INITIATED EXPLOSIVE DEVICES)

3 p 3-Terms and definitions ndards/sist/7a8f6d0f-5341-4acc-aada-60e14409f32c/iso-fdis-17546

For the purposes of this document, the following terms and definitions apply.

ISO and IEC maintain terminology databases for use in standardization at the following addresses:

- IEC Electropedia: available at https://www.electropedia.org/
- ISO Online browsing platform: available at https://www.iso.org/obp

3.1

activation

process of making an assembled *cell* (3.4) functional, by introducing an electrolyte at the manufacturing facility during cell production, which is used to define the start of *battery* (3.3) shelf *life* (3.19)

Note 1 to entry: See References [4], [5], [6-[4], [5], [6]] and [11-[11].].

3.2

aging

permanent loss of capacity due to repeated cycling or passage of time from *activation* (3.1(3.2))

Note 1 to entry: See Reference [6References [6].].

3.3

battery

two or more *cells* (3.4) which are electrically connected together, fitted with devices necessary for use, for example, case, terminals, marking and *protective devices* (3.26)

Note 1 to entry: A single cell battery is considered a "cell" [9" [9].].

Note 2 to entry: A battery may also include some or more attachments, such as electrical bypass devices, charge control electronics, heaters, temperature sensors, thermal switches, and thermal control elements [4][5][4][5].

Note 3 to entry: Units that are commonly referred to as "battery packs", "modules", or "battery assemblies" having the primary function of providing a source of power to another piece of equipment are, for the purposes of this document, treated as batteries [9].

3.4

cell

single encased electrochemical unit (one positive and one negative electrode) which exhibits a voltage differential across its two terminals

Note 1 to entry: See Reference [9References [9].].

3.5

COTS cell

cell (3.4Cell) mass-produced for terrestrial use by third parties such as distributors

Note 1 to entry: COTS cells have various kinds of types, 26650, 21700,18650, 17670, 18500, 18350, 17500, 16340, 14500, 10440 of cylindrical shape identified by external dimensions, and rectangular shape in metallic container, and thin pouch type.

3.6

dangerous phenomenon

phenomenon where a *lithium-ion battery* (3.20) is damaged

EXAMPLE fire (3.10(3.10),) bust/explosion (3.8(3.8),) leakage (3.18(3.18)) of cell (3.4) electrolyte, venting (3.34(3.34),) burns from excessively high external temperatures, rupture (3.28(3.28)) of battery case with exposure of internal components, and smokes

3.7

disassembly

vent (3.34(3.39)) or *rupture* (3.28) where solid matter from any part of a *cell* (3.4) or *battery* (3.3) penetrates a wire mesh screen (annealed aluminium wire with a diameter of 0,25 mm and grid density of 6 wires per centimetre to 7 wires per centimetre) placed 25 cm away from the cell or battery

Note 1 to entry: See Reference [9References [9].].

3.8

explosion

condition that occurs when a *cell* (3.4) container or *battery* (3.3) case violently opens and major components are forcibly expelled and the cell or battery casing is torn or split

Note 1 to entry: See References [12 [12]] and [14[14].].

2

2

3.9

external short circuit

direct connection between positive and negative terminals of a *cell* (3.4) or *battery* (3.3) that provides less than 0,1 Ω resistance path for current flow

Note 1 to entry: An external short circuit occurs when a direct connection between the positive and negative terminals is made where the connection resistance is sufficiently low enough to higher than rated current flow through the cell.

Note 2 to entry: See Reference [9References [9].].

3.10

fire

flames emitted from the test cell (3.4) or battery (3.3)

Note 1 to entry: See References [9[9]] and [12[12]].

3.11

out-gassing

evolution of gas from one or more of the electrodes in a *cell* (3.4)

Note 1 to entry: See Reference [6References [6].].

3.12

harm

physical injury or damage to the health of people or damage to property or the environment

3.13

hazard

potential source of harm (3.12(3.16))

Note 1 to entry: The term hazard is qualified in order to define its origin or the nature of the expected *harm* (3.12(3.16)) (e.g. electric shock hazard, crushing hazard, cutting hazard, toxic hazard, *fire* (3.10(3.14)) hazard, drowning hazard).

3.14

hermetic seal

permanent air-tight seal

Note 1 to entry: See Reference [10 References [10].].

3.15

intercalation

process where lithium-ions are reversibly removed or inserted into a host material without causing significant structural change to that host

Note 1 to entry: See Reference [11 References [11].].

© ISO 2023 - All rights reserved

3.16

intended use

use of a product, process or service in accordance with specifications, instructions and information provided by the supplier

Note 1 to entry: See Reference [12References [12].].

3.17

internal resistance

opposition to the flow of current within a *cell* (3.4) or a *battery* (3.3) that is, sum of electronic resistance and ionic resistance with the contribution to total effective resistance including inductive/capacitive properties

3.18

leakage

visible escape of electrolyte or other material from a *cell* (3.4) or *battery* (3.3) or the loss of material (except battery casing, handling devices or labels) from a cell or battery such that the loss of mass exceeds specific values

Note 1 to entry: Mass loss means a loss of mass that exceeds the values in <u>Table 1.</u>

Table 1 — Mass loss limit

mass M of cell	Toh Standards mass loss limit
<i>M</i> < 1 g	0,5 %
$1 g \le M \le 75 g$	os://standards.iteh. 0,2%
<i>M</i> > 75 g	0,1 %

Note 2 to entry: In order to quantify the mass loss, the following procedure is provided:

$$L_{mass} = \frac{M_{+} - M_{2}}{M_{+}} = \frac{M_{1} - M_{2}}{M_{+}} \times 100 \frac{|SO/FDIS|}{M_{1}} = \frac{M_{1} - M_{2}}{M_{1}} \times 100 \frac{|SO/FDIS|}{M_{1}} = \frac{M_{1} - M_{2}}{M_{1}} \times 100 \frac{|SO/FDIS|}{M_{2}} = \frac{M_{1} - M_{2}}{M_{1}} \times 100 \frac{|SO/FDIS|}{M_{2}} = \frac{M_{1} - M_{2}}{M_{1}} \times 100 \frac{|SO/FDIS|}{M_{2}} = \frac{M_{1} - M_{2}}{M_{2}} = \frac{M_{1} - M_{2}}{M_{2}} \times 100 \frac{|SO/FDIS|}{M_{2}} = \frac{M_{$$

where M_1 is the mass before test-and; M_2 is the mass after test; and L_{mass} is lost mass (%).

When mass loss does not exceed the values in Table 1 Table 1, it is considered as "no mass loss" [9" [9]].

3.19

life

duration of maintaining a required performance [e.g. 50 % of beginning of life (BOL) capacity], estimated in years (calendar life) or in the number of charge/discharge cycle

Note 1 to entry: See Reference [6References [6].].

3.20

lithium-ion battery

rechargeable electrochemical *cell* (3.4) or *battery* (3.3) in which the positive and negative electrodes are both *intercalation* (3.15(3.19)) compounds (intercalated lithium exists in an ionic or quasi-atomic form with the lattice of the electrode material) constructed with no metallic lithium in either electrode

Note 1 to entry: See Reference [9References [9].].

4

4