Designation: D1655 – 15

AnAmerican National Standard

Standard Specification for Aviation Turbine Fuels[1]

This standard is issued under the fixed designation D1655; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the U.S. Department of Defense.

1. Scope*

1.1 This specification covers the use of purchasing agencies in formulating specifications for purchases of aviation turbine fuel under contract.

1.2 This specification defines specific types of aviation turbine fuel for civil use in the operation and certification of aircraft and describes fuels found satisfactory for the operation of aircraft and engines. Although not specifically designed for other applications, Specification D1655 fuels also may be used in other applications with or without further modification (for example, additization) when approved by the appropriate authority. The specification can be used as a standard in describing the quality of aviation turbine fuels from production to the aircraft.

1.3 This specification does not define the quality assurance testing and procedures necessary to ensure that fuel in the distribution system continues to comply with this specification after batch certification. Such procedures are defined elsewhere, for example in ICAO 9977, EI/JIG Standard 1530, JIG 1, JIG 2, API 1543, API 1595, and ATA-103.

1.4 This specification does not include all fuels satisfactory for aviation turbine engines. Certain equipment or conditions of use may permit a wider, or require a narrower, range of characteristics than is shown by this specification.

1.5 Aviation turbine fuels defined by this specification may be used in other than turbine engines that are specifically designed and certified for this fuel.

1.6 This specification no longer includes wide-cut aviation turbine fuel (Jet B). FAA has issued a Special Airworthiness Information Bulletin which now approves the use of Specification D6615 to replace Specification D1655 as the specification for Jet B and refers users to this standard for reference.

1.7 The values stated in SI units are to be regarded as standard. No other units of measurement are included in this standard.

2. Referenced Documents

2.1 *ASTM Standards:*[2]

D56 Test Method for Flash Point by Tag Closed Cup Tester
D86 Test Method for Distillation of Petroleum Products at Atmospheric Pressure
D93 Test Methods for Flash Point by Pensky-Martens Closed Cup Tester
D129 Test Method for Sulfur in Petroleum Products (General High Pressure Decomposition Device Method)
D130 Test Method for Corrosiveness to Copper from Petroleum Products by Copper Strip Test
D156 Test Method for Saybolt Color of Petroleum Products (Saybolt Chromometer Method)
D240 Test Method for Heat of Combustion of Liquid Hydrocarbon Fuels by Bomb Calorimeter
D323 Test Method for Vapor Pressure of Petroleum Products (Reid Method)
D381 Test Method for Gum Content in Fuels by Jet Evaporation
D445 Test Method for Kinematic Viscosity of Transparent and Opaque Liquids (and Calculation of Dynamic Viscosity)
D1266 Test Method for Sulfur in Petroleum Products (Lamp Method)
D1298 Test Method for Density, Relative Density, or API Gravity of Crude Petroleum and Liquid Petroleum Products by Hydrometer Method
D1319 Test Method for Hydrocarbon Types in Liquid Petroleum Products by Fluorescent Indicator Adsorption
D1322 Test Method for Smoke Point of Kerosine and Aviation Turbine Fuel
D1405 Test Method for Estimation of Net Heat of Combustion of Aviation Fuels
D1660 Method of Test for Thermal Stability of Aviation Turbine Fuels (Withdrawn 1992)[3]

[1] This specification is under the jurisdiction of ASTM Committee D02 on Petroleum Products, Liquid Fuels, and Lubricants and is the direct responsibility of Subcommittee D02.J0.01 on Jet Fuel Specifications.

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[2] For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

[3] The last approved version of this historical standard is referenced on www.astm.org.

***A Summary of Changes section appears at the end of this standard**

D1840 Test Method for Naphthalene Hydrocarbons in Aviation Turbine Fuels by Ultraviolet Spectrophotometry
D2276 Test Method for Particulate Contaminant in Aviation Fuel by Line Sampling
D2386 Test Method for Freezing Point of Aviation Fuels
D2622 Test Method for Sulfur in Petroleum Products by Wavelength Dispersive X-ray Fluorescence Spectrometry
D2624 Test Methods for Electrical Conductivity of Aviation and Distillate Fuels
D2887 Test Method for Boiling Range Distribution of Petroleum Fractions by Gas Chromatography
D2892 Test Method for Distillation of Crude Petroleum (15-Theoretical Plate Column)
D3120 Test Method for Trace Quantities of Sulfur in Light Liquid Petroleum Hydrocarbons by Oxidative Microcoulometry
D3227 Test Method for (Thiol Mercaptan) Sulfur in Gasoline, Kerosine, Aviation Turbine, and Distillate Fuels (Potentiometric Method)
D3240 Test Method for Undissolved Water In Aviation Turbine Fuels
D3241 Test Method for Thermal Oxidation Stability of Aviation Turbine Fuels
D3242 Test Method for Acidity in Aviation Turbine Fuel
D3338 Test Method for Estimation of Net Heat of Combustion of Aviation Fuels
D3343 Test Method for Estimation of Hydrogen Content of Aviation Fuels
D3701 Test Method for Hydrogen Content of Aviation Turbine Fuels by Low Resolution Nuclear Magnetic Resonance Spectrometry
D3828 Test Methods for Flash Point by Small Scale Closed Cup Tester
D3948 Test Method for Determining Water Separation Characteristics of Aviation Turbine Fuels by Portable Separometer
D4052 Test Method for Density, Relative Density, and API Gravity of Liquids by Digital Density Meter
D4057 Practice for Manual Sampling of Petroleum and Petroleum Products
D4171 Specification for Fuel System Icing Inhibitors
D4176 Test Method for Free Water and Particulate Contamination in Distillate Fuels (Visual Inspection Procedures)
D4294 Test Method for Sulfur in Petroleum and Petroleum Products by Energy Dispersive X-ray Fluorescence Spectrometry
D4306 Practice for Aviation Fuel Sample Containers for Tests Affected by Trace Contamination
D4529 Test Method for Estimation of Net Heat of Combustion of Aviation Fuels
D4809 Test Method for Heat of Combustion of Liquid Hydrocarbon Fuels by Bomb Calorimeter (Precision Method)
D4865 Guide for Generation and Dissipation of Static Electricity in Petroleum Fuel Systems
D4952 Test Method for Qualitative Analysis for Active Sulfur Species in Fuels and Solvents (Doctor Test)
D4953 Test Method for Vapor Pressure of Gasoline and Gasoline-Oxygenate Blends (Dry Method)
D5001 Test Method for Measurement of Lubricity of Aviation Turbine Fuels by the Ball-on-Cylinder Lubricity Evaluator (BOCLE)
D5006 Test Method for Measurement of Fuel System Icing Inhibitors (Ether Type) in Aviation Fuels
D5190 Test Method for Vapor Pressure of Petroleum Products (Automatic Method) (Withdrawn 2012)[3]
D5191 Test Method for Vapor Pressure of Petroleum Products (Mini Method)
D5452 Test Method for Particulate Contamination in Aviation Fuels by Laboratory Filtration
D5453 Test Method for Determination of Total Sulfur in Light Hydrocarbons, Spark Ignition Engine Fuel, Diesel Engine Fuel, and Engine Oil by Ultraviolet Fluorescence
D5972 Test Method for Freezing Point of Aviation Fuels (Automatic Phase Transition Method)
D6045 Test Method for Color of Petroleum Products by the Automatic Tristimulus Method
D6379 Test Method for Determination of Aromatic Hydrocarbon Types in Aviation Fuels and Petroleum Distillates—High Performance Liquid Chromatography Method with Refractive Index Detection
D6469 Guide for Microbial Contamination in Fuels and Fuel Systems
D6615 Specification for Jet B Wide-Cut Aviation Turbine Fuel
D6751 Specification for Biodiesel Fuel Blend Stock (B100) for Middle Distillate Fuels
D7042 Test Method for Dynamic Viscosity and Density of Liquids by Stabinger Viscometer (and the Calculation of Kinematic Viscosity)
D7153 Test Method for Freezing Point of Aviation Fuels (Automatic Laser Method)
D7154 Test Method for Freezing Point of Aviation Fuels (Automatic Fiber Optical Method)
D7345 Test Method for Distillation of Petroleum Products and Liquid Fuels at Atmospheric Pressure (Micro Distillation Method)
D7524 Test Method for Determination of Static Dissipater Additives (SDA) in Aviation Turbine Fuel and Middle Distillate Fuels—High Performance Liquid Chromatograph (HPLC) Method
D7566 Specification for Aviation Turbine Fuel Containing Synthesized Hydrocarbons
D7797 Test Method for Determination of the Fatty Acid Methyl Esters Content of Aviation Turbine Fuel Using Flow Analysis by Fourier Transform Infrared Spectroscopy – Rapid Screening Method
E29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications

2.2 *IP Standards:*[4]

EI/JIG 1530 Quality Assurance Requirements for the Manufacture, Storage and Distribution of Aviation Fuels to Airports

[4] Available from Energy Institute, 61 New Cavendish St., London, WIG 7AR, U.K., http://www.energyinst.org.uk.

IP 12 Determination of Specific Energy
IP 16 Determination of Freezing Point of Aviation Fuels—Manual Method
IP 57 Petroleum Products—Determination of the Smoke Point of Kerosine
IP 71 Section 1 Petroleum Products—Transparent and Opaque Liquids—Determination of Kinematic Viscosity and Calculation of Dynamic Viscosity
IP 123 Petroleum Products—Determination of Distillation Characteristics at Atmospheric Pressure
IP 154 Petroleum Products—Corrosiveness to Copper—Copper Strip Test
IP 156 Petroleum Products and Related Materials—Determination of Hydrocarbon Types—Fluorescent Indicator Adsorption Method
IP 160 Crude Petroleum and Liquid Petroleum Products—Laboratory Determination of Density— Hydrometer Method
IP 170 Determination of Flash Point—Abel Closed-Cup Method
IP 216 Particulate Contaminant in Aviation Fuel
IP 225 Copper Content of Aviation Turbine Fuel
IP 227 Silver Corrosion of Aviation Turbine Fuel
IP 274 Determination of Electrical Conductivity of Aviation and Distillate Fuels
IP 323 Determination of Thermal Oxidation Stability of Gas Turbine Fuels
IP 336 Petroleum Products—Determination of Sulfur Content—Energy-Dispersive X-ray Fluorescence Spectrometry
IP 342 Petroleum Products—Determination of Thiol (Mercaptan) Sulfur in Light and Middle Distillate Fuels—Potentiometric Method
IP 354 Determination of the Acid Number of Aviation Fuels-Colour-Indicator Titration Method
IP 365 Crude Petroleum and Petroleum Products—Determination of Density—Oscillating U-tube Method
IP 406 Petroleum Products—Determination of Boiling Range Distribution by Gas Chromatography
IP 423 Determination of Particulate Contamination in Aviation Turbine Fuels by Laboratory Filtration
IP 435 Determination of the Freezing Point of Aviation Turbine Fuels by the Automatic Phase Transition Method
IP 436 Determination of Aromatic Hydrocarbon Types in Aviation Fuels and Petroleum Distillates— High Performance Liquid Chromatography Method with Refractive Index Detection
IP 523 Determination of Flash Point—Rapid Equilibrium Closed Cup Method
IP 528 Determination for the Freezing Point of Aviation Turbine Fuels—Automatic Fibre Optic Method
IP 529 Determination of the Freezing Point of Aviation Turbine Fuels—Automatic Laser Method
IP 540 Determination of the Existent Gum Content of Aviation Turbine Fuel—Jet Evaporation Method
IP 583 Determination of the Fatty Acid Methyl Esters Content of Aviation Turbine Fuel Using Flow Analysis by Fourier Transform Infrared Spectroscopy—Rapid Screening Method
IP 585 Determination of Fatty Acid Methyl Esters (FAME), Derived from Bio-diesel Fuel, in Aviation Turbine Fuel—GC-MS with Selective Ion Monitoring/Scan Detection Method
IP 590 Determination of Fatty Acid Methyl Esters (FAME) in Aviation Fuel—HPLC Evaporative Light Scattering Detector Method
IP 599 Determination of Fatty Acid Methyl Esters (FAME) in Aviation Turbine Fuel by Gas Chromatography using Heart-cut and Refocusing

2.3 *API Standards:*[5]

API 1543 Documentation, Monitoring and Laboratory Testing of Aviation Fuel During Shipment from Refinery to Airport
API 1595 Design, Construction, Operation, Maintenance, and Inspection of Aviation Pre-Airfield Storage Terminals

2.4 *Joint Inspection Group Standards:*[6]

JIG 1 Aviation Fuel Quality Control & Operating Standards for Into-Plane Fuelling Services
JIG 2 Aviation Fuel Quality Control & Operating Standards for Airport Depots & Hydrants

2.5 *ANSI Standard:*[7]

ANSI 863 Report of Test Results

2.6 *Other Standards:*

Defence Standard (Def Stan) 91-91 Turbine Fuel, Aviation Kerosine Type, Jet A-1[8]
IATA Guidance Material on Microbiological Contamination in Aircraft Fuel Tanks Ref. No: 9680-02[9]
EN14214 Automotive Fuels—Fatty Acid Methyl Esters (FAME) for Diesel Engines—Requirements and Test Methods[10]
Bulletin Number 65 MSEP Protocol[11]
ATA-103 Standard for Jet Fuel Quality Control at Airports[12]
ICAO 9977 Manual on Civil Aviation Jet Fuel Supply[13]

[5] Available from American Petroleum Institute (API), 1220 L. St., NW, Washington, DC 20005-4070, http://www.api.org.
[6] Available from Joint Inspection Group (JIG), http://www.jigonline.com.
[7] Available from American National Standards Institute (ANSI), 25 W. 43rd St., 4th Floor, New York, NY 10036.
[8] Available from Procurement Executive DFS (Air), Ministry of Defence, St. Giles Court 1, St. Giles High St., London WC2H 8LD.
[9] Available from International Air Transport Association (IATA), (Head Office) 800 Place Victoria, PO Box 113, Montreal, H4Z 1M1, Quebec, Canada. www.iata-online.com.
[10] Available from European Committee for Standardization (CEN), 36 rue de Stassart, B-1050, Brussels, Belgium, http://www.cenorm.be.
[11] Available from Joint Inspection Group (JIG), http://www.jigonline.com.
[12] Available from Air Transport Association of America, Inc. (ATA) d/b/a Airlines for America, 1301 Pennsylvania Ave. NW, Suite 1100, Washington, D.C. 20004, http://www.airlines.org.
[13] Available from International Civil Aviation Organization (ICAO), 999 University St., Montreal, Quebec H3C 5H7, Canada, http://www.icao.int.

AFRL-RQ-WP-TR-2013-0271 Determination of the Minimum Use Level of Fuel System Icing Inhibitor (FSII) in JP-8 that will Provide Adequate Icing Inhibition and Biostatic Protection for Air Force Aircraft[14]

3. Terminology

3.1 *Definitions of Terms Specific to This Standard:*

3.1.1 *identified incidental materials, n*—chemicals and compositions that have defined upper content limits in an aviation fuel specification but are not approved additives.

4. General

4.1 This specification, unless otherwise provided, prescribes the required properties of aviation turbine fuel at the time and place of delivery.

5. Classification

5.1 Two types of aviation turbine fuels are provided, as follows:

5.1.1 *Jet A and Jet A-1*—Relatively high flash point distillates of the kerosine type.

5.2 Jet A and Jet A-1 represent two grades of kerosine fuel that differ in freezing point. Other grades would be suitably identified.

5.3 This specification previously cited the requirements for Jet B. Requirements for Jet B fuel now appear in Specification D6615.

6. Materials and Manufacture

6.1 Aviation turbine fuel is a complex mixture predominantly composed of hydrocarbons and varies depending on crude source and manufacturing process. Consequently, it is impossible to define the exact composition of Jet A/A-1. This specification has therefore evolved primarily as a performance specification rather than a compositional specification. It is acknowledged that this largely relies on accumulated experience; therefore the specification limits aviation turbine fuels to those made from conventional sources or by specifically approved processes.

6.1.1 Aviation turbine fuel, except as otherwise specified in this specification, shall consist predominantly of refined hydrocarbons (see Note 1) derived from conventional sources including crude oil, natural gas liquid condensates, heavy oil, shale oil, and oil sands. The use of jet fuel blends containing components from other sources is permitted only in accordance with Annex A1.

NOTE 1—Conventionally refined jet fuel contains trace levels of materials that are not hydrocarbons, including oxygenates, organosulfur, and nitrogenous compounds.

6.1.2 Fuels used in certified engines and aircraft are ultimately approved by the certifying authority subsequent to formal submission of evidence to the authority as part of the type certification program for that aircraft and engine model. Additives to be used as supplements to an approved fuel must also be similarly approved on an individual basis (see X1.2.4 and X1.15.1).

6.2 *Additives*—May be added to each type of aviation turbine fuel in the amount and of the composition specified in Table 2 or the following list of approved material:

6.2.1 Other additives are permitted under 6.1 and 8.1. These include fuel performance enhancing additives and fuel handing and maintenance additives as found under Table 2. The quantities and types must be declared by the fuel supplier and agreed to by the purchaser. Only additives approved by the aircraft certifying authority are permitted in the fuel on which an aircraft is operated.

6.2.1.1 Biocidal additives are available for controlled usage. Where such an additive is used in the fuel, the approval status of the additive and associated conditions must be checked for the specific aircraft and engines to be operated.

6.2.1.2 *Fuel System Icing Inhibitor:*

(1) Diethylene Glycol Monomethyl Ether (DiEGME), conforming to the requirements of Specification D4171, Type III, may be used in concentrations of 0.07 % to 0.15 % by volume. Some aircraft require higher levels than 0.07 % by volume.

(2) Test Method D5006 may be used to determine the concentration of DiEGME in aviation fuels.

6.3 *Identified Incidental Materials*—Table 3 lists specific materials that have an agreed limit, known as Identified Incidental Materials. Specification D1655 does not require that each batch of fuel be analyzed for identified incidental materials where there is essentially no risk of contamination exceeding Table 3 limits. Where a supplier risk assessment suggests that identified incidental materials could exceed Table 3 limits, jet fuel should be confirmed to comply with Table 3 limits prior to airport supply because airports generally are not equipped to mitigate identified incidental material content that exceeds specification limits. Further guidance concerning these materials is presented in X1.16.

6.4 Guidance material is presented in Appendix X2 concerning the need to control processing additives in jet fuel production.

7. Detailed Requirements

7.1 The aviation turbine fuel shall conform to the requirements prescribed in Table 1.

7.2 Test results shall not exceed the maximum or be less than the minimum values specified in Table 1. No allowance shall be made for the precision of the test methods. To determine conformance to the specification requirement, a test result may be rounded to the same number of significant figures as in Table 1 using Practice E29. Where multiple determinations are made, the average result, rounded in accordance with Practice E29, shall be used.

8. Workmanship, Finish, and Appearance

8.1 The aviation turbine fuel specified in this specification shall be visually free of undissolved water, sediment, and suspended matter. The odor of the fuel shall not be nauseating or irritating. If the fuel has an odor similar to that of "rotten

[14] Available from Defense Technical Information Center (DTIC), 8725 John J. Kingman Rd., Ft. Belvoir, VA 22060-6218, http://www.dtic.mil/dtic, accession number ADA595127.

TABLE 1 Detailed Requirements of Aviation Turbine Fuels[A]

Property		Jet A or Jet A-1	Test Methods[B]
COMPOSITION			
Acidity, total mg KOH/g	max	0.10	D3242/IP 354
1. Aromatics, percent by volume	max	25	D1319 or IP 156
2. Aromatics, percent by volume	max	26.5	D6379/IP 436
Sulfur, mercaptan,[C] percent by mass	max	0.003	D3227/IP 342
Sulfur, total percent by mass	max	0.30	D1266, D2622, D4294, D5453, or IP 336
VOLATILITY			
Distillation temperature, °C:			D86,[D] D2887/IP 406,[E] D7345[F], IP 123[D]
10 % recovered, temperature	max	205	
50 % recovered, temperature		report	
90 % recovered, temperature		report	
Final boiling point, temperature	max	300	
Distillation residue, %	max	1.5	
Distillation loss, %	max	1.5	
Flash point, °C	min	38[G]	D56, D93,[H] D3828,[H] IP 170[H] or IP 523[H]
Density at 15 °C, kg/m^3		775 to 840	D1298/IP 160 or D4052 or IP 365
FLUIDITY			
Freezing point, °C	max	–40 Jet A[I]	D5972/IP 435, D7153/IP 529, D7154/IP 528, or D2386/IP 16
		–47 Jet A-1[I]	
Viscosity –20 °C, mm^2/s[J]	max	8.0	D445/IP 71, Section 1 or D7042[K]
COMBUSTION			
Net heat of combustion, MJ/kg	min	42.8[L]	D4529, D3338, D4809, or IP 12
One of the following requirements shall be met:			
(*1*) Smoke point, mm, or	min	25.0	D1322/IP 57
(*2*) Smoke point, mm, and	min	18.0	D1322/IP 57
Naphthalenes, vol, %	max	3.0	D1840
CORROSION			
Copper strip, 2 h at 100 °C	max	No. 1	D130/IP 154
THERMAL STABILITY			
(2.5 h at control temperature of 260 °C min)			
Filter pressure drop, mm Hg	max	25	D3241/IP 323
Tube rating: One of the following requirements shall be met:[M]			
(*1*) Annex A1 VTR, VTR Color Code	Less than	3 No peacock or abnormal color deposits	
(*2*) Annex A2 ITR or Annex 3 ETR, nm average over area of 2.5 mm^2	max	85	
CONTAMINANTS			
Existent gum, mg/100 mL	max	7	D381, IP 540
Microseparometer,[N] Rating			D3948
Without electrical conductivity additive	min	85	
With electrical conductivity additive	min	70	
ADDITIVES		See 6.2	
Electrical conductivity, pS/m		[O]	D2624/IP 274

[A] For compliance of test results against the requirements of Table 1, see 7.2.
[B] The test methods indicated in this table are referred to in Section 11.
[C] The mercaptan sulfur determination may be waived if the fuel is considered sweet by the doctor test described in Test Method D4952.
[D] D86 and IP 123 distillation of jet fuel is run at Group 4 conditions, except Group 3 condenser temperature is used.
[E] D2887/IP 406 results shall be converted to estimated D86 or IP 123 results by application of the correlation in Appendix X4 on Correlation for Jet and Diesel Fuel in Test Method D2887 or Annex G of IP 406. Distillation residue and loss limits provide control of the distillation process during the use of Test Method D86, and they do not apply to Test Method D2887/IP 406. Distillation residue and loss shall be reported as "not applicable" (N/A) when reporting D2887 results.
[F] Results from Test Method D7345 shall be corrected for relative bias as described in Test Method D7345.
[G] A higher minimum flash point specification can be agreed upon between purchaser and supplier.
[H] Aviation turbine fuel results obtained by Test Method D93 can be up to 1 °C higher than those obtained by Test Method D56. Results obtained by Test Methods D3828, IP 170, and IP 523 can be up to 2 °C lower than those obtained by Test Method D56, which is the preferred method. In case of dispute, Test Method D56 shall apply.
[I] Other freezing points can be agreed upon between supplier and purchaser.
[J] 1 mm^2/s = 1 cSt.
[K] Test Method D7042 results shall be converted to bias-corrected kinematic viscosity results by the application of the correction described in Test Method D7042 for jet fuel at –20 °C (currently subsection 15.4.4).
[L] For all grades use either Eq 1 or Table 1 in Test Method D4529 or Eq 2 in Test Method D3338. Test Method D4809 can be used as an alternative. In case of dispute, Test Method D4809 shall be used.
[M] Tube deposit ratings shall be measured by D3241 Annex A2 ITR or Annex A3 ETR, when available. If the Annex A2 ITR device reports "N/A" for a tube's volume measurement, the test shall be a failure and the value reported as >85 nm. Visual rating of the heater tube by the method in D3241 Annex A1 is not required when Annex A2 ITR or Annex A3 ETR deposit thickness measurements are reported. In case of dispute between results from visual and metrological methods, the referee shall be considered the Annex A3 ETR method if available, otherwise Annex A2 ITR.
[N] At point of manufacture. See X1.13 for guidance concerning the application of microseparometer results in fuel distribution.
[O] If electrical conductivity additive is used, the conductivity shall not exceed 600 pS/m at the point of use of the fuel. When electrical conductivity additive is specified by the purchaser, the conductivity shall be 50 to 600 pS/m under the conditions at point of delivery.

$$1\ \text{pS/m} = 1 \times 10^{-12}\ \Omega^{-1}\ m^{-1}$$

TABLE 2 Detailed Information for Additives for Aviation Turbine Fuels

Additive	Dosage
Fuel Performance Enhancing Additives	
Antioxidants[A,B]	24.0 mg/L max[C]
One of the following:	
2,6 ditertiary-butyl phenol	
2,6 ditertiary-butyl-4-methyl phenol	
2,4 dimethyl-6-tertiary-butyl-phenol	
75 % minimum, 2,6 ditertiary-butyl phenol plus	
25 % maximum mixed tertiary and tritertiary butyl-phenols	
55 % minimum 2,4 dimethyl-6-tertiary-butyl phenol plus	
15 % minimum 2,6 ditertiary-butyl-4-methyl phenol,	
remainder as monomethyl and dimethyl tertiary-butyl phenols	
72 % minimum 2,4 dimethyl-6-tertiary-butyl phenol plus	
28 % maximum monomethyl and dimethyl-tertiary-butyl-phenols	
Metal Deactivator (MDA)[A]	
N,N-disalicylidene-1,2-propane diamine	
On initial blending	2.0 mg/L max[C,D]
After field reblending cumulative concentration	5.7 mg/L max
Fuel System Icing Inhibitor[E,F]	0.07 % by volume, min[G]
Diethylene Glycol Monomethyl Ether (see Specification D4171)	0.15 % by volume, max
Fuel Handling and Maintenance Additives	
Electrical Conductivity Improver[H]	
Stadis 450[I,J]	
On initial blending	3 mg/L max
After field reblending, cumulative concentration	5 mg/L max
If the additive concentration is unknown at time of retreatment, additional concentration is restricted to 2 mg/L max	
Leak Detection Additive Tracer A (LDTA-A)[K]	1 mg/kg max
Biocidal Additives[E,L,M]	
Biobor JF[N]	
Kathon FP1.5[O]	
Corrosion Inhibitor/Lubricity Improvers[P]	
One of the following:	
HiTEC 580	23 mg/L max
Innospec DCI-4A	23 mg/L max
Nalco 5403	23 mg/L max

[A] The active ingredient of the additive must meet the composition specified.
[B] Supporting data have been filed at ASTM International Headquarters and may be obtained by requesting Research Report RR:D02-1125.
[C] Active ingredient (not including weight of solvent).
[D] At the point of manufacture, Metal Deactivator Additive (MDA) may be added to improve thermal oxidative stability subject to the following limitations:
(*1*) No more than 5 % of the jet fuel batches produced in a 12 month period may be treated with MDA to meet Table 1 thermal oxidative stability requirements (260 °C test temperature).
(*2*) The batch of fuel shall pass Table 1 thermal oxidative stability requirements at a test temperature of 245 °C prior to any MDA addition.
(*3*) The fuel batch after MDA addition (2.0 mg/L maximum MDA) shall pass Table 1 thermal oxidative stability requirements at a test temperature of 275 °C.
(*4*) The thermal oxidative stability test result at 245 °C prior to MDA addition, the original test result at 260 °C and the test result at 275 °C (post MDA addition) and the concentration of MDA added shall be reported on the Refinery Certificate of Quality.

Initial addition of more than 2.0 mg/L MDA to jet fuel that meets Table 1 thermal oxidative stability requirements (260 °C test temperature) prior to MDA addition is permitted when fuel will be transported in supply chains where copper contamination can occur: the maximum cumulative addition in this table still applies.

MDA may be added to jet fuel in the distribution system to recover thermal oxidative stability performance lost during distribution (after refinery release). The Certificate of Quality shall show the initial thermal oxidative stability test result, the result after the addition of the MDA and the concentration of MDA added.
[E] The quantity must be declared by the fuel supplier and agreed to by the purchaser.
[F] The lower FSII concentration limit allowable in Jet Fuel is based on research by the U.S. Air Force as documented in report AFRL-RQ-WP-TR-2013-0271. Some engines and aircraft as certified require higher minimum concentrations of icing inhibitor than the lower limit in this jet fuel specification. When fueling an aircraft, the fuel should be additized to the concentration levels specified in the appropriate engine and aircraft manual.
[G] Some aircraft require higher levels than 0.07 % by volume.
[H] If electrical conductivity improver is used, the conductivity shall not exceed 600 pS/m at the point of use of the fuel. When electrical conductivity additive is specified by the purchaser, the conductivity shall be 50 pS/m to 600 pS/m under the conditions at point of delivery.

$$1\ \text{pS/m} = 1 \times 10^{-12}\ \Omega^{-1} m^{-1}$$

[I] Stadis 450 is a registered trademark marketed by Innospec Inc., Innospec Manufacturing Park, Oil Sites Road, Ellesmere Port, Cheshire, CH65 4EY, UK.
[J] Stadis 450 content can be analyzed by Test Method D7524.
[K] Tracer A (LDTA-A) is a registered trademark of Tracer Research Corp., 3755 N. Business Center Dr., Tucson, AZ 85705.
[L] Biocidal additives are available for controlled usage. Where such an additive is used in the fuel, the approval status of the additive and associated conditions must be checked for the specific aircraft and engines to be operated.
[M] Refer to the Aircraft Maintenance Manual (AMM) to determined if either biocide is approved for use and for their appropriate use and dosage.
[N] Biobor JF is a registered trademark of Hammonds Technical Services, Inc. 910 Rankin Rd., Houston, TX 77073.
[O] KATHON is a trademark of The Dow Chemical Company ("Dow") or an affiliated company of Dow, 2030 Dow Center, Midland, MI 48674.
[P] More information concerning minimum treat rates of corrosion inhibitor/lubricity improver additives is contained in X1.10.2.