



Standard Specification for Aviation Certification Turbine Fuel¹

This standard is issued under the fixed designation D7223; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

1. Scope

1.1 This specification covers the use of purchasing agencies in formulating specifications for purchases of aviation turbine fuel under contract.

1.2 This specification defines one specific type of aviation turbine fuel for civil use in the certification of aircraft. The specification can be used as a standard in describing the quality of this aviation fuel from the refinery to the aircraft.

1.3 This specification does not include the fuels that are commonly used in aviation turbine engines. Those are listed in Specification [D1655](#).

1.4 The aviation turbine fuel defined by this specification may be used in other than turbine engines that are specifically designed and certified for this fuel.

1.5 The values stated in SI units are to be regarded as standard. No other units of measurement are included in this standard.

1.5.1 *Exception*—Units of pressure are also given in psi.

2. Referenced Documents

2.1 *ASTM Standards*:²

- [D56](#) Test Method for Flash Point by Tag Closed Cup Tester
- [D86](#) Test Method for Distillation of Petroleum Products at Atmospheric Pressure
- [D130](#) Test Method for Corrosiveness to Copper from Petroleum Products by Copper Strip Test
- [D381](#) Test Method for Gum Content in Fuels by Jet Evaporation
- [D445](#) Test Method for Kinematic Viscosity of Transparent and Opaque Liquids (and Calculation of Dynamic Viscosity)

¹ This specification is under the jurisdiction of ASTM Committee D02 on Petroleum Products, Liquid Fuels, and Lubricants and is the direct responsibility of Subcommittee D02.J0 on Aviation Fuels.

Current edition approved April 1, 2015. Published May 2015. Originally approved in 2005. Last previous edition approved in 2011 as D7223 – 11. DOI: 10.1520/D7223-11R15.

² For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

- [D1266](#) Test Method for Sulfur in Petroleum Products (Lamp Method)
- [D1298](#) Test Method for Density, Relative Density, or API Gravity of Crude Petroleum and Liquid Petroleum Products by Hydrometer Method
- [D1319](#) Test Method for Hydrocarbon Types in Liquid Petroleum Products by Fluorescent Indicator Adsorption
- [D1322](#) Test Method for Smoke Point of Kerosine and Aviation Turbine Fuel
- [D1655](#) Specification for Aviation Turbine Fuels
- [D1840](#) Test Method for Naphthalene Hydrocarbons in Aviation Turbine Fuels by Ultraviolet Spectrophotometry
- [D2386](#) Test Method for Freezing Point of Aviation Fuels
- [D2622](#) Test Method for Sulfur in Petroleum Products by Wavelength Dispersive X-ray Fluorescence Spectrometry
- [D2624](#) Test Methods for Electrical Conductivity of Aviation and Distillate Fuels
- [D2887](#) Test Method for Boiling Range Distribution of Petroleum Fractions by Gas Chromatography
- [D3227](#) Test Method for (Thiol Mercaptan) Sulfur in Gasoline, Kerosine, Aviation Turbine, and Distillate Fuels (Potentiometric Method)
- [D3241](#) Test Method for Thermal Oxidation Stability of Aviation Turbine Fuels
- [D3242](#) Test Method for Acidity in Aviation Turbine Fuel
- [D3338](#) Test Method for Estimation of Net Heat of Combustion of Aviation Fuels
- [D3828](#) Test Methods for Flash Point by Small Scale Closed Cup Tester
- [D3948](#) Test Method for Determining Water Separation Characteristics of Aviation Turbine Fuels by Portable Separator
- [D4052](#) Test Method for Density, Relative Density, and API Gravity of Liquids by Digital Density Meter
- [D4057](#) Practice for Manual Sampling of Petroleum and Petroleum Products
- [D4171](#) Specification for Fuel System Icing Inhibitors
- [D4294](#) Test Method for Sulfur in Petroleum and Petroleum Products by Energy Dispersive X-ray Fluorescence Spectrometry
- [D4306](#) Practice for Aviation Fuel Sample Containers for Tests Affected by Trace Contamination

- [D4529 Test Method for Estimation of Net Heat of Combustion of Aviation Fuels](#)
- [D4809 Test Method for Heat of Combustion of Liquid Hydrocarbon Fuels by Bomb Calorimeter \(Precision Method\)](#)
- [D4952 Test Method for Qualitative Analysis for Active Sulfur Species in Fuels and Solvents \(Doctor Test\)](#)
- [D5001 Test Method for Measurement of Lubricity of Aviation Turbine Fuels by the Ball-on-Cylinder Lubricity Evaluator \(BOCLE\)](#)
- [D5006 Test Method for Measurement of Fuel System Icing Inhibitors \(Ether Type\) in Aviation Fuels](#)
- [D5453 Test Method for Determination of Total Sulfur in Light Hydrocarbons, Spark Ignition Engine Fuel, Diesel Engine Fuel, and Engine Oil by Ultraviolet Fluorescence](#)
- [D5972 Test Method for Freezing Point of Aviation Fuels \(Automatic Phase Transition Method\)](#)
- [D6378 Test Method for Determination of Vapor Pressure \(VP_x\) of Petroleum Products, Hydrocarbons, and Hydrocarbon-Oxygenate Mixtures \(Triple Expansion Method\)](#)
- [D6469 Guide for Microbial Contamination in Fuels and Fuel Systems](#)
- [E29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications](#)

3. General

3.1 This specification, unless otherwise provided, prescribes the required properties of aviation certification turbine fuel at the time and place of delivery.

4. Classification

- 4.1 One type of aviation turbine fuel is provided, as follows:
 - 4.1.1 *Jet C-1*—A relatively wide boiling range volatile distillate.

5. Materials and Manufacture

5.1 Aviation turbine fuel, except as otherwise specified in this specification, shall consist of blends of refined hydrocarbons (see [Note 1](#)) derived from conventional sources including crude oil, natural gas liquid condensates, heavy oil, shale oil, and oil sands. The use of jet fuel blends, containing components from other sources, is permitted only on a specific, individual basis (see [Annex A1](#) on fuels from non-conventional sources in [Specification D1655](#)).

NOTE 1—Conventionally refined jet fuel contains trace levels of materials which are not hydrocarbons, including oxygenates, organosulfur, and nitrogenous compounds.

5.1.1 Fuels used in engines and aircraft are ultimately approved by the certifying authority subsequent to formal submission of evidence to the authority as part of the type certification program for that aircraft and engine model. Additives to be used as supplements to an approved fuel must also be similarly approved on an individual basis (see [Specification D1655](#)).

5.2 *Additives*—May be added to this aviation turbine fuel in the amount and of the composition specified in the following list of approved material:³

5.2.1 *Antioxidants*—In amounts not to exceed 24.0 mg/L active ingredients (not including mass of solvent):

- 5.2.1.1 2,6-ditertiary-butyl phenol.
- 5.2.1.2 2,6-ditertiary-butyl-4-methyl phenol.
- 5.2.1.3 2,4-dimethyl-6-tertiary-butyl phenol.

5.2.1.4 75 % minimum 2,6-ditertiary-butyl phenol, plus 25 % maximum mixed tertiary and tritertiary-butyl phenols.

5.2.1.5 55 % minimum 2,4-dimethyl-6-tertiary-butyl phenol, plus 15 % minimum 2,6-ditertiary-butyl-4-methyl phenol, remainder as monomethyl and dimethyl tertiary-butyl phenols.

5.2.1.6 72 % minimum 2,4-dimethyl-6-tertiary-butyl phenol, 28 % maximum monomethyl and dimethyl-tertiary-butyl phenols.

5.2.2 *Metal Deactivator Additive (MDA)*, in amount not to exceed 2.0 mg/L (not including mass of solvent) on initial fuel manufacture at the refinery. Higher initial concentrations are permitted in circumstances where copper contamination is suspected to occur during distribution. Cumulative concentration of MDA when retreating the fuel shall not exceed 5.7 mg/L:

5.2.2.1 *N,N*-disalicylidene-1,2-propane diamine.

5.2.3 *Electrical Conductivity Additive*—Stadis 450⁴ not to exceed 3 mg/L.

5.2.3.1 When loss of fuel conductivity necessitates retreatment with electrical conductivity additive, the following concentration limits apply:

At Manufacture:		
Stadis 450		3 mg/L, max
Retreatment:		
Stadis 450		cumulative total 5 mg/L, max

5.2.4 *Leak Detection Additive*—Tracer A (LDTA-A)⁵ may be added to the fuel in amounts not to exceed 1 mg/kg.

5.2.5 Other additives are permitted. These include fuel system icing inhibitor and special purpose additives such as biocides. The quantities and types must be declared by the fuel supplier and agreed to by the purchaser. Only additives approved by the aircraft certifying authority are permitted in the fuel on which an aircraft is operated.

5.2.5.1 Biocidal additives are available for controlled usage. Where such an additive is used in the fuel, the approval status of the additive and associated conditions must be checked for the specific aircraft and engines to be operated.

5.2.5.2 *Fuel System Icing Inhibitor*:

(1) *Diethylene Glycol Monomethyl Ether (DIEGME)*, conforming to the requirements of [Specification D4171](#), Type III, may be used in concentrations of 0.10 % to 0.15 % by volume.

(2) [Test Method D5006](#) may be used to determine the concentration of DIEGME in aviation fuels.

³ Supporting data (Guidelines for Approval or Disapproval of Additives) have been filed at ASTM International Headquarters and may be obtained by requesting Research Report RR:D02-1125.

⁴ Stadis 450 is a registered trademark marketed by Innospec Inc., Innospec Manufacturing Park, Oil Sites Road, Ellesmere Port, Cheshire, CH65 4EY, UK.

⁵ Tracer A (LDTA-A) is a registered trademark of Tracer Research Corp., 3755 N. Business Center Dr., Tucson, AZ 85705.

5.3 Guidance material is presented in Appendix X3 of Specification **D1655** concerning the need to control processing additives in jet fuel production.

6. Detailed Requirements

6.1 The aviation turbine fuel shall conform to the requirements prescribed in **Table 1**.

6.2 Test results shall not exceed the maximum or be less than the minimum values specified in **Table 1**. No allowance shall be made for the precision of the test methods. To determine conformance to the specification requirement, a test

result may be rounded to the same number of significant figures as in **Table 1** using Practice **E29**. Where multiple determinations are made, the average result, rounded according to Practice **E29**, shall be used.

6.3 If any additives are used, the aviation turbine fuel shall conform to the **Table 2** listed requirements.

7. Workmanship, Finish, and Appearance

7.1 The aviation turbine fuel herein specified shall be visually free of undissolved water, sediment, and suspended matter. The odor of the fuel shall not be nauseating or irritating.

TABLE 1 Detailed Requirements of Aviation Certification Turbine Fuel^A

Property		Jet C-1	ASTM Test Method ^B
Acidity, total mg KOH/g	max	0.10	D3242
Vol fraction of aromatics, cL/L	min/max	8 to 25	D1319
Mass fraction of mercaptan sulfur, ^C cg/g	max	0.003	D3227
Mass fraction of total sulfur, cg/g	max	0.30	D1266, D2622, D4294 or D5453
Distillation temperature, °C (°F):			
Initial boiling point, temperature	min/max	70/100 (158/212)	D2887, D86^D
5 % recovered, temperature	min/max	80/110 (176/230)	D2887, D86^D
10 % recovered, temperature	min/max	90/120 (194/248)	D2887, D86^D
20 % recovered, temperature	min/max	105/140 (221/284)	D2887, D86^D
50 % recovered, temperature	min/max	150/195 (302/383)	D2887, D86^D
90 % recovered, temperature	min/max	215/255 (419/491)	D2887, D86^D
Final boiling point, temperature	min/max	240/290 (464/554)	D2887, D86^D
Flash Point, °C (°F)	report		D56, D3828
Density at 15 °C, kg/m ³		750 to 840	D1298, D4052
Vapor pressure ^E			
at 25 °C, kPa (psi)	report	3.0 (0.44) to 5.5 (0.80)	D6378^F
at 38 °C, kPa (psi)	report	5.6 (0.8) to 8.2 (1.2)	D6378^F
at 50 °C, kPa (psi)	min/max	10.0 (1.45) to 12.5 (1.82)	D6378^F
at 100 °C, kPa (psi)	report	56 (8.1) to 60 (8.7)	D6378^F
Freezing point, °C	max	-35	D2386, D5972^G
Viscosity at -20 °C, mm ² /s ^H	max	8.0	D445
Net heat of combustion, MJ/kg	min	42.8 ^I	D4529, D3338, or D4809
One of the following requirements shall be met:			
(1) Smoke point, mm, or	min	25	D1322
(2) Smoke point, mm, and	min	18	D1322
Naphthalenes, vol, %	max	3.0	D1840
Copper strip, 2 h at 100 °C	max	No. 1	D1300
Thermal stability:			
(2.5 h at control temperature of 260 °C min):			
Filter pressure drop, mm Hg	max	25	D3241^J
Tube deposit less than		3	D3241
		No <i>Peacock</i> or <i>Abnormal</i> Color Deposits	
Existent gum, mg/100 mL	max	7	D381
Microseparator, ^K Rating	max	1b	D3948
Without electrical conductivity additive	min	85	
With electrical conductivity additive	min	70	
Lubricity ^L – BOCLE WSD, mm	max	0.85	D5001^L
Additives:			see 5.2
Electrical conductivity, pS/m	required	50 to 600	D2624
Other	optional		

^A For compliance of test results against the requirements of **Table 1**, see **6.2**.

^B The test methods indicated in this table are referred to in Section **10**.

^C The mercaptan sulfur determination may be waived if the fuel is considered sweet by the doctor test described in Test Method **D4952**.

^D If Test Method **D2887** is used, use correlation procedure (Appendix X5) in Test Method **D2887** to convert **D2887** temperatures to **D86** equivalent temperatures. Both minimums and maximums shall be met.

^E Absolute vapor pressure (VPx) is the primary property to be controlled; 2,2 dimethylbutane and toluene, as cited in Section 11 and Note 14 of Test Method **D6378** – 08, shall be used as verification fluids. 1.0 kPa = 0.145 psi.

^F Latest version. Record absolute vapor pressure (VPx).

^G Test Method **D5972** may produce a higher (warmer) result than that from Test Method **D2386**. In case of dispute, Test Method **D2386** shall be the referee method.

^H 1 mm²/s = 1 cSt.

^I Use either Eq 1 or Table 1 in Test Method **D4529** or Eq 2 in Test Method **D3338**. Test Method **D4809** may be used as an alternative. In case of dispute, Test Method **D4809** shall be used.

^J Tube deposits shall always be reported by the Visual Method.

^K At point of manufacture.

^L Lubricity test can be waived with purchaser's agreement.