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INTERNATIONAL

Designation: B 211 - 00

Standard Specification for Aluminum and Aluminum-Alloy Bar, Rod, and Wire¹

This standard is issued under the fixed designation B 211; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense.

1. Scope*

1.1 This specification² covers rolled or cold-finished bar, rod, and wire in alloys (Note 1) and tempers as shown in Table 2.

Note 1—Throughout this specification use of the term *alloy* in the general sense includes aluminum as well as aluminum alloy.

Note 2—The term *cold finished* is used to indicate the type of surface finish, sharpness of angles, and dimensional tolerances produced by drawing through a die.

Note 3—See Specification B 221 for aluminum and aluminum-alloy extruded bars, rods, wire, shapes, and tubes; and Specification B 316 for aluminum and aluminum-alloy rivet and cold-heading wire and rods.

- 1.2 Alloy and temper designations are in accordance with ANSI H35.1. The equivalent UNS alloy designations are those of Table 1 preceded by A9, for example, A91100 for aluminum 1100 in accordance with Practice E 527E 527.
- 1.3 A complete metric companion to Specification B 211 has been developed—B 211M; therefore, no metric equivalents are presented in this specification.
- 1.4 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.

2. Referenced Documents

- 2.1 The following documents of the issue in effect on date of material purchase form a part of this specification to the extent referenced herein:
 - 2.2 ASTM Standards:
 - B 557 Test Methods of Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products³
 - B 594 Practice for Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Aerospace Applications³
 - B 597 Practice for Heat Treatment of Aluminum Alloys³
 - B 660 Practices for Packaging/Packing of Aluminum and Magnesium Products³
- ¹ This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum-Alloy Wrought Products.
- Current edition approved May 10, 2000. Published August 2000. Originally published as B 211–46T. Last previous edition B 211–99.
- ² For ASME Boiler and Pressure Vessel Code applications see related Specification SB-211 in Section II of that Code.
 - ³ Annual Book of ASTM Standards, Vol 02.02.

- B 666/B 666M Practice for Identification Marking of Aluminum Products³
- E 29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications⁴
- E 34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Alloys⁵
- E 55 Practice for Sampling Wrought Nonferrous Metals and Alloys for Determination of Chemical Composition⁵
- E 227 Test Method for Optical Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Point-to-Plane Technique⁵
- E 290 Test Methods for Bend Testing of Material for Ductility⁶
- E 527 Practice for Numbering Metals and Alloys (UNS)⁷
- E 607 Test Method for Optical Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Point-to-Plane Technique, Nitrogen Atmosphere⁸
- E 716 Practices for Sampling Aluminum and Aluminum Alloys for Spectrochemical Analysis⁸
- E 1004 Test Method for Electromagnetic (Eddy-Current)
 Measurements of Electrical Conductivity⁹
- E 1251 Test Method for Optical Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Argon Atmosphere, Point-to-Plane, Unipolar Self Initiating Capacitor Discharge⁸
- G 47 Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum-Alloy Products¹⁰
- 2.3 ANSI Standards:
- H35.1 Alloy and Temper Designation Systems for Aluminum³
- H35.2 Dimensional Tolerances for Aluminum Mill Products³
- 2.4 Federal Standard:

⁴ Annual Book of ASTM Standards, Vol 14.02.

⁵ Annual Book of ASTM Standards, Vol 03.05.

Annual Book of ASTM Standards, Vol 03.01.
 Annual Book of ASTM Standards, Vol 01.01.

⁸ Annual Book of ASTM Standards, Vol 03.06.

⁹ Annual Book of ASTM Standards, Vol 03.03.

¹⁰ Annual Book of ASTM Standards, Vol 03.02.

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Fed. Std. No. 123 Marking for Shipment (Civil Agencies)¹¹ 2.5 *Military Standard*:

MIL-STD-129 Marking for Shipment and Storage¹¹

3.1.4 *cold-finished rod*—rod brought to final dimensions by cold working to obtain improved surface finish and dimensional tolerances.

TABLE 1 Chemical Composition Limits^{A,B,C}

Alloy	Silicon	Iron	Copper	Manganese	Magnesium	Chromium	Zinc	Bismuth	Lead	Titanium		Other ments ^D	Aluminum
				Ü	· ·						Each	Total ^E	•
1060	0.25	0.35	0.05	0.03	0.03		0.05			0.03	0.03 ^F		99.60 min ^G
1100	0.95 Si + Fe		0.05-0.20	0.05			0.10				0.05	0.15	99.00 min ^G
2011	0.40	0.7	5.0-6.0				0.30	0.20-0.6	0.20-0.6		0.05	0.15	remainder
2014	0.50-1.2	0.7	3.9-5.0	0.40-1.2	0.20-0.8	0.10	0.25			0.15	0.05	0.15	remainder
2017	0.20-0.8	0.7	3.5-4.5	0.40-1.0	0.40-0.8	0.10	0.25			0.15	0.05	0.15	remainder
2024	0.50	0.50	3.8-4.9	0.30-0.9	1.2-1.8	0.10	0.25			0.15	0.05	0.15	remainder
2219	0.20	0.30	5.8-6.8	0.20-0.40	0.02		0.10			0.02-0.10	0.05^{H}	0.15 ^H	remainder
3003	0.6	0.7	0.05-0.20	1.0-1.5			0.10				0.05	0.15	remainder
5052	0.25	0.40	0.10	0.10	2.2-2.8	0.15-0.35	0.10				0.05	0.15	remainder
5056	0.30	0.40	0.10	0.05-0.20	4.5-5.6	0.05-0.20	0.10				0.05	0.15	remainder
Alclad 5056	5056 allo	y clad wit	h 6253 alloy										
5154	0.25	0.40	0.10	0.10	3.1-3.9	0.15-0.35	0.20			0.20	0.05	0.15	remainder
6061	0.40-0.8	0.7	0.15-0.40	0.15	0.8-1.2	0.04-0.35	0.25			0.15	0.05	0.15	remainder
6110	0.7-1.5	0.8	0.20-0.7	0.20-0.7	0.50-1.1	0.04-0.25	0.30			0.15	0.05	0.15	remainder
6253 ¹	J	0.50	0.10		1.0-1.5	0.04-0.35	1.6-2.4				0.05	0.15	remainder
6262	0.40-0.8	0.7	0.15-0.40	0.15	0.8-1.2	0.04-0.14	0.25	0.40-0.7	0.40-0.7	0.15	0.05	0.15	remainder
7075	0.40	0.50	1.2-2.0	0.30	2.1-2.9	0.18-0.28	5.1-6.1			0.20	0.05	0.15	remainder

^A Limits are in mass percent maximum unless otherwise shown.

2.6 Aerospace Material Specification:

AMS-H-6088 Heat Treatment of Aluminum Alloys¹²

3. Terminology

- 3.1 Definitions: Definitions:
- 3.1.1 *alclad wire*—wire having on its surface a metallurgically bonded aluminum or aluminum-alloy coating that is anodic to the core alloy to which it is bonded, thus electrolytically protecting the core alloy against corrosion.
- 3.1.2 *bar*—a solid product that is long in relation to cross section which is square or rectangular (excluding plate and flattened wire) with sharp or rounded corners or edges, or is a regular hexagon or octagon, and in which at least one perpendicular distance between parallel faces is 0.375 in. or greater.
- 3.1.3 *cold-finished bar*—bar brought to final dimensions by cold working to obtain improved surface finish and dimensional tolerances.

- 3.1.5 *drawn wire*—wire brought to final dimensions by drawing through a die. 1995/511/astm-b211-00
- 3.1.6 *flattened and slit wire*—flattened wire which has been slit to obtain square edges.
- 3.1.7 *flattened wire*—a solid section having two parallel flat surfaces and rounded edges produced by roll-flattening round wire.
 - 3.1.8 *producer*—the primary manufacturer of the material.
- 3.1.9 *rod*—a solid product 0.375 in. or greater in diameter that is long in relation to cross section.
- 3.1.10 *supplier*—includes only the category of jobbers and distributors as distinct from producers.
- 3.1.11 wire—a solid section long in relation to its cross-sectional dimensions, having a cross section that is round, hexagonal, or octagonal and whose diameter, width, or greatest distance between parallel faces is less than 0.375 in., or having a symmetrical cross section that is square or rectangular (excluding flattened wire) with sharp or rounded corners or edges.
 - 3.2 Definitions of Terms Specific to This Standard:
- 3.2.1 *capable of*—The term *capable of* as used in this specification means that the test need not be performed by the

 $^{^{\}it B}$ Analysis shall be made for the elements for which limits are shown in this table.

^C For purposes of determining conformance to these limits, an observed value or a calculated value obtained from analysis shall be rounded to the nearest unit in the last right-hand place of figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E 29E 29.

Dothers includes listed elements for which no specific limit is shown as well as unlisted metallic elements. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic Others elements. Should any analysis by the producer or the purchaser establish that an Others element exceeds the limit of Each or that the aggregate of several Others elements exceeds the limit of Total, the material shall be considered non-conforming

E Other elements—Total shall be the sum of unspecified metallic elements 0.010 % or more each, rounded to the second decimal before determining the sum.

F Vanadium 0.05 % max.

^G The aluminum content is the difference between 100.00 % and the sum of all other metallic elements and silicon present in amounts of 0.010 % or more each, rounded to the second decimal before determining the sum.

H Vanadium 0.05-0.15 % zirconium 0.10-0.25 %. The total for other elements does not include vanadium and zirconium.

Composition of cladding alloy as applied during the course of manufacture. Samples from finished wire shall not be required to conform to these limits.

^J 45 to 65 % of actual magnesium content.

¹¹ Available from Standardization Documents Order Desk, Bldg. 4 Section D, 700 Robbins Ave., Philadelphia, PA 19111-5094, Attn: NPODS.

¹² Available from SAE, 400 Commonwealth Drive, Warrendale, PA 15096.

producer of the material. However, should subsequent testing by the purchaser establish that the material does not meet these requirements, the material shall be subject to rejection.

4. Ordering Information

- 4.1 Orders for material to this specification shall include the following information:
- 4.1.1 This specification designation (which includes the number, the year, and the revision letter, if applicable),
 - 4.1.2 Quantity in pieces or pounds,
 - 4.1.3 Alloy (Section 7),
 - 4.1.4 Temper (Section 9),
- 4.1.5 *Product Form*—Rolled or cold finished bar, rolled or cold finished rod, or wire,
- 4.1.6 Geometry and Dimensions—Diameter for rounds; distance across flats for square-cornered squares, hexagons, or octagons; width and depth for square-cornered rectangles (orders for squares, hexagons, octagons, or rectangles with rounded corners usually require a drawing),
 - 4.1.7 Length,
- 4.1.8 Tensile property limits and dimensional tolerances for sized not covered in Table 2 and in ANSI H35.2, respectively.
- 4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:

- 4.2.1 Whether heat treatment in accordance with Practice B 597B 597 is required (8.2),
- 4.2.2 Whether 7075-O material is required to develop requirements for T73 temper (see 10.1.2),
- 4.2.3 Whether bend testing is required for 2017, 2024, or 3003 (Section 12),
- 4.2.4 When specified finish of bar and rod is not required (Section 16),
- 4.2.5 Whether marking for identification is required (Section 17),
- 4.2.6 Whether ultrasonic inspection is required (Section 18, Table 3).
- 4.2.7 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (Section 20).
 - 4.2.8 Whether certification is required (Section 22), and
- 4.2.9 Whether Practices B 660B 660 apply, and if so, the levels of preservation, packaging, and packing required (Section 23).

5. Manufacture

5.1 The products covered by this specification shall be produced either by hot extruding and cold finishing or by hot rolling with or without cold finishing, at the option of the producer.

TABLE 2 Mechanical Property Limits^A

TABLE 2 Mechanical Property Limits						
Temper	Specified Diameter or _	Tensile :	Strength, ksi	Yield Strength ^B (0.2 % offset),	Elonga- tion ^B in 2 in.	
тотпрог	Thickness, in.	CUI min en 1	Premax eW	min, ksi	or $4 \times$ Diameter, min, %	
		Alumi	num 1060			
0	0.124 and under	8.0	211 00 "		•••	
	0.125 and over	0.0	<u> 211-00</u>	2.5	25	
H14 H18 ttps://standards	0.374 and under 0.374 and under	lards/si ^{12.0} 62 fdb	61-e1a4-4 <u>9</u> ce-bebe-	8cffe9e <mark>10.0</mark> 5f1/astm	-b211-0 <u>0</u>	
		Alumi	num 1100			
0	0.124 and under	11.0	15.5			
	0.125 and over	11.0	15.5	3.0	25	
H12	0.374 and under	14.0				
H14	0.374 and under	16.0				
H16	0.374 and under	19.0				
H18	0.374 and under	22.0				
H112	all	11.0		3.0		
F	all	С		С		
		Allo	oy 2011			
T3	0.125-1.500	45.0		38.0	10	
	1.501-2.000	43.0		34.0	12	
	2.001-3.500	42.0		30.0	12	
T4 and T451 ^D	0.125-8.000	40.0		18.0	16	
T8	0.125-3.250	54.0		40.0	10	
		Allo	y 2014 ^E			
0	0.124 and under	•••	35.0			
	0.125-8.000		35.0		12	
T4, T42 ^F , and T451 ^D	0.124 and under	55.0				
	0.125-8.000 ^G	55.0		32.0	16	
T6, T62 ^F , and T651 ^D	0.124 and under	65.0				
	0.125-8.000 ^G	65.0	•••	55.0	8	
		Allo	y 2017 ^E			
0	0.124 and under		35.0			
	0.125-8.000		35.0		16	
T4, T42 ^F , and T451 ^D	0.124 and under	55.0			***	

TABLE 2 Continued

	Consisted Diameter or	Tensile St	trength, ksi	Yield Strength ^B	Elonga- tion ^B in 2 in.
Temper	Specified Diameter or Thickness, in.	min	max	- (0.2 % offset), min, ksi	or 4 × Diam eter, min, %
	0.125-8.000 ^H	55.0		32.0	12
			2024 ^E		
)	0.124 and under		35.0		
	0.125-8.000		35.0	•••	16
T36	0.124 and under	69.0		•••	
	0.125-0.375	69.0		52.0	10
Γ4′	0.124 and under	62.0		•••	
	0.125-0.499	62.0		45.0 ¹	10
	0.500-4.500 ^G	62.0		42.0 ¹	10
	4.501–6.500 ^J	62.0		40.0	10
	6.501-8.000 ^J	58.0		38.0	10
Γ42 ^{<i>F</i>}	0.124 and under	62.0	•••		
	0.125-1.000	62.0	•••	40.0	10
	1.001–6.500 ^{<i>G</i>}	62.0	•••	40.0	10
Γ351 ^{<i>D</i>}	0.500–6.500 ^G	62.0	•••	45.0	10
Γ6	0.124 and under	62.0			
	0.125–6.500 ^G	62.0		50.0	5
Г62 ^{<i>F</i>}	0.124 and under	60.0			
	0.125-6.500 ^G	60.0		46.0	5
Γ851 ^D	0.500–6.500 ^G	66.0		58.0	5
		Alloy	2219		
Γ851 ^D	0.500–2.000	58.0		40.0	4
•	2.001–4.000	57.0		39.0	4
			3003		
<u> </u>	all	14.0		5.0	25
)		17.0	19.0		
H12	0.374 and under 0.374 and under		1 01001010		
114		20.0	T		
H16	0.374 and under	24.0	ards"iteh	91)	
H18 H112	0.374 and under	14.0		5.0	
п112 F	all all	14.0 C		5.0 C	
-	all)		5052		
)	0.124 and under		32.0		
	0.125 and over	25.0	32.0	9.5	25
H32	0.124 and under	31.0	11 00		
			11-00	23.0	
	0.125-0.374	31.0			•••
-134 the //standa	1 10 074 444 4 454	1 / 34 0 (0 011 (200 1 /actm	
	1 10 074 444 4 454	1 / 34 0 (0 011 (e-8cffe9e <mark>26.0</mark> 5f1/astm	
	0.374 and under	ards/si34.0 62 fdb6	1-e1a4-49ce-bebe		-b211-0 0
H36	0.374 and under 0.124 and under	ards/s 34.0 62 fdb6	1-e1a4-49ce-bebe	29.0	-b211-0 <u>0</u>
136 138	0.374 and under 0.124 and under 0.125–0.374	ards/si ^{34.0} 62 fdb6 37.0 37.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38	0.374 and under 0.124 and under 0.125–0.374 0.374 and under	34.0 62 fdb6 37.0 37.0 39.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36	0.374 and under 0.124 and under 0.125–0.374 0.374 and under	34.0 62 db6 37.0 37.0 39.0 c	1-e1a4-49ce-bebe 	29.0	-b211-0 0
136 4957 Stantian 138 =	0.374 and under 0.124 and under 0.125–0.374 0.374 and under all	34.0 62 fdb6 37.0 37.0 39.0 c	1-e1a4-49ce-bebe	29.0 c	I-b211-0 0
H38 =	0.374 and under 0.125–0.374 0.374 and under all	34.0 62 db6 37.0 37.0 39.0 c	1-e1a4-49ce-bebe	29.0 c	-b211-0 0
H36 (PS7/Standard) H38 =	0.374 and under 0.124 and under 0.125-0.374 0.374 and under all 0.124 and under 0.125 and over	34.0 62 fdb6 37.0 37.0 39.0 C	1-e1a4-49ce-bebe	29.0 	-b211-0 0
H36 (PS7/Standard) H38 =	0.374 and under 0.124 and under 0.125-0.374 0.374 and under all 0.124 and under 0.125 and over 0.374 and under	34.0 62 fdb 6 37.0 37.0 39.0 <i>c</i> Alloy 44.0 46.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36	0.374 and under 0.124 and under 0.374 and under all 0.124 and under all 0.124 and under 0.125 and over 0.374 and under 0.374 and under	34.0 62 fdb6 37.0 37.0 39.0 <i>c</i> Alloy 44.0	1-e1a4-49ce-bebe	29.0 c	-b211-0 0
H38 = D	0.374 and under 0.125–0.374 0.374 and under all 0.124 and under 0.125 and over 0.374 and under	34.0 62 fdb 6 37.0 37.0 39.0 C Alloy 44.0 46.0 44.0 52.0	1-e1a4-49ce-bebe	29.0 c	-b211-0 0
H38 = D	0.374 and under 0.125–0.374 0.374 and under all 0.124 and under all 0.124 and under 0.125 and over 0.374 and under	34.0 62 fdb 6 37.0 37.0 39.0 C Alloy 44.0 46.0 44.0 52.0 50.0	1-e1a4-49ce-bebe	29.0 c	-b211-0 0
H38 = D	0.374 and under 0.125–0.374 0.374 and under all 0.124 and under 0.125 and over 0.374 and under	34.0 62 fdb 6 37.0 37.0 39.0 C Alloy 44.0 46.0 44.0 52.0	1-e1a4-49ce-bebe	29.0 c	-b211-0 0
H38 = D	0.374 and under 0.125-0.374 0.374 and under all 0.124 and under all 0.124 and under 0.125 and over 0.374 and under	34.0 62 fdb 6 37.0 37.0 39.0 C Alloy 44.0 46.0 44.0 52.0 50.0 58.0 55.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H38 = D	0.374 and under 0.124 and under 0.125-0.374 0.374 and under all 0.124 and under 0.125 and over 0.374 and under	34.0 62 fdb 6 37.0 37.0 39.0 <i>c</i> Alloy 44.0 46.0 44.0 52.0 50.0 58.0 55.0 60.0	1-e1a4-49ce-bebe	29.0 c	-b211-0 0
H38 = D	0.374 and under 0.125-0.374 0.374 and under all 0.124 and under all 0.124 and under 0.125 and over 0.374 and under	Alloy 44.0 46.0 44.0 52.0 50.0 58.0 58.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 = D H111 H12 H32 H14 H34 H18 H38 H192 H392	0.374 and under 0.125–0.374 0.374 and under all 0.124 and under 0.125 and over 0.125 and over 0.374 and under	Alloy 34.0 37.0 37.0 39.0 Alloy 44.0 46.0 44.0 52.0 50.0 58.0 55.0 60.0 58.0 Alclad A	1-e1a4-49ce-bebe	29.0	-b211-0 0
H38 =	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy Al	1-e1a4-49ce-bebe	29.0	-b211-0 0
H38 H38 = D H111 H12 H32 H14 H34 H18 H38 H192 H392	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy 44.0 46.0 44.0 52.0 50.0 58.0 58.0 Alclad A	1-e1a4-49ce-bebe	29.0	-b211-0 0
138 138 138 1111 112 132 114 134 118 138 1192 1392	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy Alloy Alloy Alloy 44.0 46.0 44.0 52.0 50.0 58.0 55.0 60.0 58.0 Alclad A	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 H38 H38 H31 H32 H32 H34 H34 H38 H38 H392 H392 H392	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy Alclad A 52.0 50.0 54.0 Alloy Alloy Alclad A 52.0 50.0 58.0 Alclad A	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 F O H111 H12 H32 H14 H34 H18 H38 H192 H392 H392 H393	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alloy 52.0 50.0 58.0 55.0 60.0 58.0 Alclad A 52.0 50.0 54.0 Alloy Alloy 30.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 = D H111 H12 H32 H14 H34 H18 H38 H192 H392 H392 D H32	0.374 and under 0.125–0.374 0.374 and under 0.125 – 0.374 0.374 and under 0.125 and over 0.125 and over 0.374 and under	Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alloy 52.0 50.0 58.0 55.0 60.0 58.0 Alclad A 52.0 50.0 Alloy Alloy Alloy Alloy 30.0 30.0 36.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 F O H1111 H12 H32 H14 H34 H18 H38 H192 H392 H392 O H32 H34	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alclad A 52.0 50.0 58.0 55.0 60.0 58.0 Alclad A 52.0 50.0 Alloy Alloy Alloy 30.0 30.0 36.0 39.0	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 F O H1111 H12 H32 H14 H34 H18 H38 H192 H392 H392 H393 O H32 H34 H36	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.125 and over 0.374 and under	Alloy Alcad A 52.0 50.0 58.0 Alcad A	1-e1a4-49ce-bebe	29.0	-b211-0 0
H36 H38 F O H1111 H12 H32 H14 H34 H18 H38 H192 H392 H392 O H32 H34	0.374 and under 0.125–0.374 0.374 and under 0.125–0.374 0.374 and under 0.125 and over 0.374 and under	Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alloy Alclad A 52.0 50.0 58.0 55.0 60.0 58.0 Alclad A 52.0 50.0 Alloy Alloy Alloy 30.0 30.0 36.0 39.0	1-e1a4-49ce-bebe	29.0	-b211-0 0

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TABLE 2 Continued

Temper	Specified Diameter or	Tensile S	trength, ksi	Yield Strength ^B (0.2 % offset),	Elonga- tion ^B in 2 in.	
	Thickness, in.	min	max	min, ksi	or $4 \times \text{Diam}$ - eter, min, %	
		Alloy	6061 ^E			
0	0.124 and under		22.0			
	0.125-8.000		22.0		18	
T4 and T451 ^D	0.124 and under	30.0				
	0.125-8.000 ^H	30.0		16.0	18	
T42 ^F	0.125-8.000 ^H	30.0		14.0	18	
T6, T62 ^F , and T651 ^D	0.124 and under	42.0				
	0.125–8.000 ^H	42.0		35.0	10	
T89 and T94	0.374 and under	54.0	•••	47.0		
		Alloy	<i>t</i> 6110			
Т9	0.374 and under	65.0	•••	63.0	2	
		Alloy	6262			
T6 and T651 ^D	0.125-8.000 ^G	42.0		35.0	10	
T9	0.125–2.000	52.0		48.0	5	
	2.001–3.000	50.0		46.0	5	
		Alloy	7075 ^E			
0	0.124 and under		40.0			
	0.125-8.000		40.0		10	
T6, T62	0.124 and under	77.0		66.0		
	0.125–4.000 ^K	77.0		66.0	7	
T651	0.124 and under	77.0		66.0		
	$0.125-4.000^{K}$	77.0		66.0	7	
	4.001-6.000	75.0		64.0	7	
	6.001-7.000	73.0	1	62.0	7	
T73 and T7351 ^D	0.124 and under	68.0	ndards			
	0.125-4.000	68.0		56.0	10	
	4.001–5.000	66.0		55.0	8	
Т	emper IIIUS	Specified Diameter	er or Thickness, in.	Bend Diamet	er Factor, N	
		Alloy	2017			
4, T42, and T451		0.125-	nd under		3^{L}	
ı		0.124 a	nd under	1		
351, T4, T42		0.124 a	nd under	3		
			-6.500 a4-49ce-bebe-	-8cffe9e575f1/astn	-b211-00	
			all	C		
12		0.374 a	nd under	2		
14			nd under	2		
16			nd under	8		

^A To determine conformance to this specification, each value for tensile strength and for yield strength shall be rounded to the nearest 0.1 ksi and each value for elongation to the nearest 0.5 %, both in accordance with the rounding-off method of Practice E 29E 29. The basis for establishment of tensile property limits is shown in Annex A1.

^B The measurement of yield strength and elongation is not required for wire less than 0.125 in. in thickness or diameter.

^C There are no tensile requirements for material in the F temper but it usually can be expected that material 1½ in. or less in thickness or diameter (except sections over 4 in. in width) will have a strength about equivalent to the H14 or H34 temper. As size increases the strength decreases to nearly that of the O temper.

^D For stress-relieved tempers, characteristics and properties other than those specified may differ somewhat from the corresponding characteristics and properties of material in the basic tempers.

^E Also available in the F temper for which no properties are specified and no tension tests are performed but for which tests are performed for confirmation of heat-treat response as required by Section 10.

F Material in the T42 or T62 tempers is not available from the materials producers. These properties can usually be obtained by the user when material is properly solution heat treated or solution and precipitation heat treated from the O or F temper. These properties also apply to samples of material in the O or F temper that are solution heat treated or solution and precipitation heat treated by the producer to determine that the material will respond to proper heat treatment. Properties attained by the user, however, may be lower than those listed if the material has been formed or otherwise cold or hot worked, particularly in the O temper, prior to solution heat treatment.

^G Properties listed for this full size increment are applicable to rod. Properties listed are also applicable to square, rectangular, hexagonal, or octagonal bar having a maximum thickness of 4 in. and a maximum cross-sectional area of 36 in.².

^H For bar, maximum cross-sectional area is 50 in.².

¹ Minimum yield strength for 2024-T4 wire and rod 0.125 in. and larger in thickness or diameter, produced in coil form for both straight length and coiled products, is 40.0 ksi.

^J Properties listed for this size increment are applicable to rod only.

^K For rounds, maximum diameter is 4 in.; for square, hexagonal, or octagonal bar, maximum thickness is 3½ in.; for rectangular bar, maximum thickness is 3 in. with corresponding maximum width of 6 in.; for rectangular bar less than 3 in. in thickness, maximum width is 10 in.

^L Bend diameter factor values stated for this full size increment apply to T4 product only. Values listed also apply to T451 product in the 0.500-8.000 in. size range.

6. Quality Assurance

- 6.1 Responsibility for Inspection and Tests—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use his own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order or at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set forth in this specification where such inspections are deemed necessary to ensure that material conforms to prescribed requirements.
- 6.2 Lot Definition—An inspection lot shall be defined as follows:
- 6.2.1 For heat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions traceable to a heat-treat lot or lots, and subjected to inspection at one time.
- 6.2.2 For nonheat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions subjected to inspection at one time.

7. Chemical Composition

7.1 Limits—The bars, rods, and wire shall conform to the chemical composition limits specified in Table 1. Conformance shall be determined by the producer by analyzing samples taken at the time the ingots are cast, or samples taken from the finished or semifinished product. If the producer has determined the chemical composition of the material during the course of manufacture, sampling and analysis of the finished product shall not be required.

Note 4—It is standard practice in the United States aluminum industry to determine conformance to the chemical composition limits prior to further processing of ingots into wrought products. Due to the continuous nature of the process, it is not practical to keep a specific ingot analysis identified with a specific quantity of finished material.

- 7.2 Number of Samples—The number of samples taken for determination of chemical composition shall be as follows:
- 7.2.1 When samples are taken at the time the ingots are cast, at least one sample shall be taken for each group of ingots cast simultaneously from the same source of molten metal.
- 7.2.2 When samples are taken from the finished or semifinished product, a sample shall be taken to represent each 4000 lb, or fraction thereof, in the lot, except that no more than one sample shall be required per piece.
- 7.3 *Methods of Sampling*—Samples for determination of chemical composition shall be taken in accordance with one of the following methods:
- 7.3.1 Samples for chemical analysis shall be taken from the material by drilling, sawing, milling, turning, clipping, etc., a representative piece or pieces to obtain a weight of prepared sample not less than 75 g. Sampling shall be in accordance with Practice E 55E 55.
- 7.3.2 Sampling for spectrochemical analysis shall be in accordance with Practices E 716E 716. Samples for other methods of analysis shall be suitable for the form of material being analyzed and the type of analytical method used.

- Note 5—It is difficult to obtain a reliable analysis of each of the components of clad materials using material in its finished state. A reasonably accurate determination of the core composition can be made if the cladding is substantially removed prior to analysis. The cladding composition is more difficult to determine because of the relatively thin layer and because of diffusion of core elements to the cladding. The correctness of cladding alloy used can usually be verified by a combination of metallographic examination and spectrochemical analysis of the surface at several widely separated points.
- 7.4 Method of Analysis—The determination of chemical composition shall be made in accordance with suitable chemical (Test Methods E 34E 34), or spectrochemical (Test Methods E 227, E 607, and E 1251E 227E 607E 1251), methods. Other methods may be used only when no published ASTM method is available. In case of dispute, the methods of analysis shall be agreed upon between the purchaser and the producer.

8. Heat Treatment

- 8.1 Unless otherwise specified in 8.2, producer or supplier heat treatment for the applicable tempers in Table 2 shall be in accordance with AMS-H-6088.
- 8.2 When specified, heat treatment of applicable tempers in Table 2 shall be in accordance with Practice B 597B 597.

9. Tensile Properties of Material As Supplied

- 9.1 *Limits*—The bar, rod, and wire shall conform to the tensile requirements in Table 2.
 - 9.2 Number of Specimens:
- 9.2.1 For material having a nominal weight of less than 1 lb/linear ft, one tension test specimen shall be taken for each 1000 lb or fraction thereof in the lot. Only one specimen shall be taken from any one piece when more than one piece is available.
- 9.2.2 For material having a nominal weight of 1 lb or more/linear ft, one tension test specimen shall be taken for each 1000 ft or fraction thereof in the lot. Only one specimen shall be taken from any one piece when more than one piece is available.
- 9.3 *Test Specimens*—Geometry of test specimens and the location in the product from which they are taken shall be as specified in Test Methods B 557B 557.
- 9.4 *Test Methods*—The tension tests shall be made in accordance with Test Methods B 557B 557.

10. Producer Confirmation of Heat-Treat Response

- 10.1 In addition to the requirements of 9.1, material in Alloys 2014, 2017, 2024, and 6061 produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for T42 temper material. The heat-treated samples may be tested prior to four days natural aging, but if they fail to conform to the T42 temper properties, the tests may be repeated after completion of four days natural aging without prejudice.
- 10.1.1 Alloy 7075 material produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and precipitation heat treatment, conform to the properties specified in Table 2 for T62 temper material.